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OVERALL AND BLADE-ELEMENT PERFORMANCE OF A 1.20-PRESSURE-RATIO FAN STAGE AT DESIGN BLADE SETTING ANGLE

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of 0.90 was obtained at 110 pe			_	flow of	
30.2 kg/sec. Maximum stage	pressure ratio w	as 1. 269 at 120 per	cent speed.		
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SUMMARY

A 51-centimeter-diameter model of a short-haul fan stage was tested in the Lewis single-stage compressor research facility. This stage was designed and built on contract by the Hamilton Standard Division of United Aircraft Corporation. Surveys of the air flow conditions ahead of the rotor, between the rotor and stator, and behind the stator were made over the stable operating range of the stage. Flow and performance parameters were calculated at the blade leading and trailing edges. Surveys were taken at equivalent rotative speeds of 80, 90, 100, 110, and 120 percent of design speed.

At the design speed of 213. 3 meters per second and weight flow of 31.2 kilograms per second (195.3 kg/(sec)($\rm m^2$) of annulus area), the stage pressure ratio of 1.16 was less than the design value of 1.2. Peak stage efficiency was 0.90 at 110 percent design speed, at a pressure ratio of 1.218 and at a flow rate of 30.2 kilograms per second. Maximum pressure ratio for this fan stage was 1.269 at 120 percent of design speed.

INTRODUCTION

NASA is currently engaged in investigating short-haul-type aircraft for commercial application. These aircraft must be dependable, economical, and have an efficient and reliable propulsion system which satisfies the low noise requirements for urban communities. The aircraft engines must be capable of a variety of operating conditions from takeoff, cruise, and approach to possible thrust reversal on landing.

In support of this program, the Lewis Research Center is investigating a variety of fan stages for short-haul engines. The low pressure ratio stages suitable for this application must operate at low tip speeds to attain the required low noise level. The Hamilton Standard Division of United Aircraft Corporation designed and fabricated under contract for NASA Lewis a full-scale (183-cm-diam.) fan stage and a 51-centimeter-

diameter model of the full-scale fan stage to be tested in the single-stage compressor test facility. The design and overall performance of the model stage was reported (ref. 1) at speeds of 80, 90, and 100 percent of design and for three rotor blade setting angles - design, design minus 5°, and design minus 7°. Noise characteristics for the full-scale fan are reported in reference 2. This report presents the design of the stage, and the experimental, overall, and blade-element performance for the stage at design rotor and stator blade setting angles.

Both overall and blade-element performance data are presented over the stable operating range for 80 to 120 percent of design speed. Surveys of the flow conditions were taken at nine radial positions. The tests were conducted in the single-stage compressor test facility at the Lewis Research Center.

AERODYNAMIC DESIGN

The fan stage was designed for a pressure ratio of 1.20, a rotor tip speed of 213.3 meters per second, efficiency of 0.908, and a weight flow per unit annulus area of 195.3 kilograms per second per square meter. The additional requirements for the fan stage were low noise and adjustable rotor blades. The overall design parameters for this stage which consists of rotor 55 and stator 55 (designated stage 55-55) are listed in table I. The selected flow path is presented in figure 1.

The rotor utilized double-circular-arc blade profiles. The rotor was designed with a tip solidity of 0.89 and a hub-tip radius ratio of 0.46. This resulted in 15 rotor blades with an aspect ratio of 1.43. The stator blades were designed utilizing NACA 400 Series airfoils. The constant chord stator blades had a tip solidity of 0.712 and a hub-tip radius ratio of 0.47. This resulted in 11 stator blades with an aspect ratio of 1.27.

The blade-element design parameters for this stage are presented in tables II and III, respectively. The blade geometry is given in table IV for the rotor and in table V for the stator. The blade-element design parameters shown are those supplied by the contractor. The symbols and equations are defined in appendixes A and B. The definitions and units used for the tabular data are presented in appendix C.

APPARATUS AND PROCEDURE

Compressor Test Facility

The compressor stage was tested in the Single-Stage Compressor Test Facility. A schematic diagram of the facility is shown in figure 2. Atmospheric air enters the test facility at an inlet located on the roof of the building, flows through the flow measuring

orifice and into the plenum chamber upstream of the test stage. The air then passes through the experimental compressor stage into the collector and is exhausted to the facility exhaust system.

Test Stage

A photograph of the test stage mounted in the research facility is shown in figure 3. Photographs of the rotor and stator are shown in figure 4. The 15 rotor blades were machined from a titanium alloy.

The rotor blade tips were contoured to provide adequate clearance for resetting the blades to a reverse flow position to provide reverse thrust capability. The nominal tip clearance at the rotor blade centerline was 0.06 centimeter. At the leading and trailing edges, the tip clearances were approximately 0.08 centimeter for the design setting angle. The stator blades were machined from an aluminum alloy. The stators are supported at both the hub and tip.

Instrumentation

The compressor weight flow was determined from measurements on a calibrated thin-plate orifice that was 38.9 centimeters in diameter. The orifice temperature was determined from an average of two Chromel-constantan thermocouples. Orifice pressures were measured by calibrated transducers.

Radial surveys of the flow were made upstream of the rotor, between the rotor and stator, and downstream of the stator (see fig. 1 for axial location). Photographs of the survey probes are shown in figure 5. Total pressure, total temperature, and flow angle were measured with the combination probe (fig. 5(a)) and the static pressure was measured with a 8° C-shaped wedge probe (fig. 5(b)). Each probe was positioned with a null-balancing, stream-directional sensitive control system that automatically alined the probe to the direction of flow. The thermocouple material was Chromel-constantan. Two combination probes and two wedge static probes were used at each of the three measuring stations.

Inner and outer wall static pressure taps were located at approximately the same axial stations as the survey probes. The circumferential locations of both types of survey probes along with inner and outer wall static pressure taps are shown in figure 6.

An electronic speed counter, in conjunction with a magnetic pickup, was used to measure rotative speed (rpm).

The estimated errors of the data based on inherent accuracies of the instrumentation and recording system are as follows:

Flow rate, kg/sec \ldots
Rotative speed, rpm
Flow angle, deg
Temperature, K
Rotor inlet total pressure, N/cm 2
Rotor outlet total pressure, N/cm^2
Stator outlet total pressure, N/cm^2
Rotor inlet static pressure, N/cm ²
Rotor outlet static pressure, N/cm^2
Stator outlet static pressure. N/cm 2

Test Procedure

The stage survey data were taken at 80, 90, 100, 110, and 120 percent of design speed. The data, at each speed line, were taken over a range of weight flow from maximum flow to the near-stall conditions. Data were recorded at nine radial positions for each speed and weight flow.

At each radial position the two combination probes behind the stator were circumferentially traversed to nine different locations across the stator gap. The wedge probes were set at midgap because previous studies showed that the static pressure across the stator gap was constant. Values of total pressure, temperature, and flow angle were recorded at each circumferential position. At the last circumferential position values of pressure, temperature, and flow angle were also recorded at stations 1 and 2. All probes were then traversed to the next radial position and the circumferential traverse procedure repeated.

At each speed the back pressure on the stage was increased by closing the sleeve valve in the collector until a stalled condition was detected by a sudden drop in stage outlet total pressure. This pressure was measured by a probe located at midpassage downstream of stators and was recorded on an X-Y plotter. Stall was corroborated by large increases in the measured blade stresses on the rotor with a sudden increase in audible noise level.

Calculation Procedure

Measured total temperatures and total pressures were corrected for Mach number and design streamline slope. These corrections were based on instrument probe calibrations given in reference 2. The stream static pressure was corrected for Mach number and streamline slope based on an average calibration for the type of probe used.

Due to the physical construction of the C-shaped static pressure wedges, it was not possible to obtain static pressure measurements at 5, 10, and 95 percent of span from the rotor tip. The static pressure at 95 percent span was obtained by assuming a linear variation in static pressure between the values at the inner wall and the probe measurement at 90 percent span. A similar variation was assumed between the static pressure measurements at the outer wall and the 15 percent span to obtain the static pressure at 5 and 10 percent span.

At each radial position, averaged values of the nine circumferential measurements of pressure, temperature rise, and flow angle downstream of the stator (station 3) were obtained. The nine values of total temperature were mass-averaged to obtain the stage total temperature rise. The nine values of total pressure were energy averaged. The measured values of pressure, temperature, and flow angle were used to calculate axial and tangential velocities at each circumferential position. The flow angles presented for each radial position are calculated based on these mass-averaged axial and tangential velocities. To obtain the overall performance, the radial values of total temperature were mass-averaged and the values of total pressure were energy-averaged. At each measuring station, the integrated weight flow was computed based on the radial survey data.

The data, measured at the three measuring stations, have been translated to planes approximating the blade leading and trailing edges by the method presented in reference 3.

The weight flow at stall was obtained in the following manner: during operation of the near stall condition, the collector valve was slowly closed in small increments. At each increment the weight flow was obtained. The weight flow obtained just before stall occurred is called the stall weight flow. The pressure ratio at stall was obtained by extrapolating the total pressure obtained from the survey data to the stall weight flow.

Orifice weight flow, total pressures, static pressures, and temperatures were all corrected to sea-level conditions based on the rotor inlet conditions.

RESULTS AND DISCUSSION

The results from this investigation are presented in three main sections. The overall performance for the rotor and the stage are presented first. Radial distributions of several performance parameters are then presented for both the rotor and the stator. Blade-element data are presented for both rotor and stator. The data are computer plotted, and occasionally a data point is omitted when it falls outside the range of the parameters shown in the figure.

All the plotted data together with some additional performance parameters for the fan stage are presented in tabular form. The overall performance data are presented in table VI. The blade-element data are presented first for the rotor in table VII and then

for the stator in table VIII. The definitions and units used for the tabular data are presented in appendix C.

Overall Performance

The overall performance for rotor 55 and stage 55-55 are presented in figures 7 and 8, respectively. For both of these computer-plotted figures, data are presented for five speeds from 80 to 120 percent of design speed. Design-point values are shown as solid symbols on both figures.

The stall points for each speed line were established by extrapolating the overall performance curves to the stall weight flow value recorded with the on-site computer. The stall lines (dashed lines) shown in figure 8 were then established by fairing a curve through the stall points associated with each speed line.

The design speed peak efficiency, of 0.87 for the stage, occurred at a measured equivalent weight flow of 26.7 kilograms per second ($167 \, \text{kg/sec/m}^2$ of annulus area) and pressure ratio of 1.177 as compared to the design weight flow of 31.2 kilograms per second ($213 \, \text{kg/sec/m}^2$ of annulus area) and pressure ratio of 1.196. At 110 percent speed, the peak efficiency of 0.90 occurred near design weight flow; the pressure ratio of 1.218 exceeded design.

The rotor efficiency (fig. 7) of 0.92 occurred at an equivalent weight flow of 30.0 kilograms per second at design speed and decreased from 0.94 at 80 percent speed to 0.88 at 120 percent speed while the stage efficiency peaked at 110 percent speed. The maximum stage pressure ratio of 1.269 was obtained at 120 percent of design speed.

At design speed, the stall margin for the stage was 18 percent. The stall margin, defined in appendix B, was based on the equivalent weight flow and pressure ratio at which peak efficiency occurred, as compared to the values just prior to stall.

Radial Distributions

The radial distributions of selected flow and performance parameters for both rotor and stator are shown in figures 9 and 10. The results are presented for three flow rates at design speed. The data shown represent the flow conditions at near stall, peak efficiency, and near choke. The performance at peak efficiency is compared to design even though peak efficiency occurred at a considerably lower weight flow (26.7 as compared to 31.2 kg/sec).

Rotor. - The measured and design energy input at peak efficiency agree at all span locations except near the hub (90 and 95 percent span) where a small dropoff in measured temperature ratio is noted. The measured total pressure ratio was less than design from

the rotor tip to 70 percent span and compares with design from 70 percent span to the hub. The deviation angles are higher than design from the tip to 70 percent span and slightly lower from 70 percent span to the hub. The loading as indicated by the D-factor was slightly higher or equal to design values across the blade span. Losses are somewhat greater than design values at all span locations except at the 70 and 95 percent stations where losses are equal to the design values.

Stator. - The total loss coefficient for the stator at peak efficiency was higher than design from the blade tip to about 80 percent span; from 80 percent span to the hub, the measured losses were lower than design. The stator blade loading (D-factor) had a similar distribution and was higher than design from the blade tip to 70 percent span and lower from 70 percent span to the hub.

Variations of Blade-Element Performance with Incidence Angle

The variations of selected blade-element performance parameters with incidence angle are shown in figure 11 for the rotor and in figure 12 for the stator. The data are presented for 80, 100, and 120 percent of speed at blade elements on streamlines located at 5, 10, 30, 50, 70, 90, and 95 percent of blade span as measured from the rotor-outlet blade tip. Design values are indicated by solid symbols.

Rotor. - At the three outer span locations (5, 10, and 30 percent span) the minimum loss with incidence angle is shown and at design speed corresponds to the design incidence angle. At the radial stations closer to the hub (50 to 95 percent span) losses continue to decrease with increasing incidence angle; however, at design incidence, the measured loss is somewhat greater than design. At design incidence angle both the work input (temperature ratio) and the pressure ratio are consistently less than design value. There is a marked increase in loss in the rotor tip region for 120 percent speed. The relative inlet Mach numbers are approaching unity at this speed (table VII) and apparently local shock patterns on the blade surfaces cause these higher losses.

Stator. - Minimum loss values were defined for this stator from the tip down to the 50 percent span location (fig. 12). At all elements except 90 and 95 percent spans, the measured losses were greater than the design values at design incidence. The loss curves indicate that the minimum loss value occurred at incidence angles about 8° higher than the design value at the 5 and 10 percent span locations. Minimum loss-incidence and design loss-incidence coincide at the 30 and 50 percent spans. At the 70, 90, and 95 percent span locations, the minimum loss-incidence was not established.

SUMMARY OF RESULTS

This report presents the aerodynamic design and both the overall and blade-element performance of a 51-centimeter-diameter fan stage compressor model of a 183-centimeter fan. This stage is one of a series of low-tip-speed - high-flow research fan stages presently being investigated for both aerodynamic design and noise generation. The stage has a design equivalent weight flow of 31.2 kilograms (195.3 kg/m² of annulus area) at a rotor blade tip speed of 213 meters per second. Radial surveys of the flow conditions at the rotor inlet, rotor outlet, and stator outlet were made over the stable operating flow range of the stage at equivalent rotative speeds from 80 to 120 percent design speed. Flow and performance parameters were calculated across a number of selected blade elements. The following principal results were obtained:

- 1. The fan stage peak efficiency of 0.87 occurred at an equivalent weight flow of 26.7 kilograms per second at a design speed of 213 meters per second and a pressure ratio of 1.177. Design values are 0.90, 31.2, and 1.196, respectively. Rotor peak efficiency occurred at an equivalent weight flow of 30.0 kilograms per second. Peak efficiency of 0.90 for the stage was obtained at 110 percent speed, a pressure ratio of 1.218, and weight flow of 30.2 kilograms per second.
- 2. Maximum stage pressure ratio of 1.290 was obtained at 120 percent speed and flow of 29.4 kilograms per second.
 - 3. The stall margin for this fan stage is 18 percent at design speed.
- 4. Radial distributions of rotor blade-element performance parameters at design speed and peak efficiency flow condition indicate somewhat higher than design losses and lower than design pressure ratios at design energy input levels. Stator blade losses and loading were also higher than design from the blade tip to 70 percent span.

Lewis Research Center,

National Aeronautics and Space Administration, Cleveland, Ohio, May 30, 1974, 501-24.

APPENDIX A

SYMBOLS

- $A_{\rm an}$ annulus area at rotor leading edge, 0.160 m²
- A_f frontal area at rotor leading edge, 0.203 m²
- $C_{\rm p}$ specific heat at constant pressure, 1004 J/(kg)(K)
- c aerodynamic chord, cm
- D diffusion factor
- i_{mc} mean incidence angle, angle between inlet air direction and line tangent to blade mean camber line at leading edge, deg
- N rotative speed, rpm
- P total pressure. N/cm²
- p static pressure, N/cm²
- r radius, cm
- SM stall margin
- T total temperature, K
- U wheel speed, m/sec
- V air velocity, m/sec
- W weight flow, kg/sec
- Z axial distance references from rotor blade hub leading edge, cm
- $\alpha_{\rm c}$ cone angle, deg
- $\alpha_{_{\mathbf{S}}}$ slope of streamline, deg
- β air angle, angle between air velocity and axial direction, deg
- $\beta_{\rm c}^{\rm '}$ relative meridional air angle based on cone angle, arctan (tan $\beta_{\rm m}^{\rm '}$ cos $\alpha_{\rm c}/{\rm cos}~\alpha_{\rm S}$), deg
- γ ratio of specific heats (1.40)
- δ ratio of rotor inlet total pressure to standard pressure of 10.13 N/m²
- δ^{O} deviation angle, angle between exit air direction and tangent to blade mean camber line at trailing edge, deg
- θ ratio of rotor inlet total temperature to standard temperature of 288.2 K

 η efficiency

 $\kappa_{
m mc}$ angle between the blade mean camber line and the meridional plane, deg

σ solidity, ratio of chord to spacing

 $\overline{\omega}$ total loss coefficient

 $\overline{\omega}_{\mathrm{p}}$ profile loss coefficient

 $\overline{\omega}_{_{\mathbf{S}}}$ shock loss coefficient

Subscripts:

ad adiabatic (temperature rise)

id ideal.

LE blade leading edge

m meridional direction

mom momentum rise

p polytropic

TE blade trailing edge

z axial direction

 θ tangential direction

1 instrumentation plane upstream of rotor

2 instrumentation plane between rotor and stator

3 instrumentation plane downstream of stator

Superscript:

relative to blade

APPENDIX B

EQUATIONS

Performance parameters are defined as follows:

Mean incidence angle

$$i_{mc} = (\beta'_c)_{LE} - (\kappa_{mc})_{LE}$$
 (B1)

Deviation angle

$$\delta^{O} = (\beta_{c}^{\dagger})_{TE} - (\kappa_{mc})_{TE}$$
 (B2)

Diffusion factor

$$D = 1 - \frac{V_{TE}^{\prime}}{V_{LE}^{\prime}} + \left| \frac{(rV_{\theta})_{TE} - (rV_{\theta})_{LE}}{(r_{TE} + r_{LE})\sigma(V_{LE}^{\prime})} \right|$$
(B3)

Total loss coefficient

$$\overline{\omega} = \frac{(P'_{id})_{TE} - (P')_{TE}}{(P')_{LE} - (p)_{LE}}$$
(B4)

Profile loss coefficient

$$\overline{\omega}_{p} = \overline{\omega} - \overline{\omega}_{s}$$
 (B5)

Total loss parameter

$$\frac{\overline{\omega}\cos\left(\beta_{\mathrm{m}}^{\prime}\right)_{\mathrm{TE}}}{2\sigma}\tag{B6}$$

Profile loss parameter

$$\frac{\overline{\omega}_{p} \cos (\beta'_{m})_{TE}}{2\sigma} \tag{B7}$$

Adiabatic (temperature-rise) efficiency

$$\eta_{\text{ad}} = \frac{\left(\frac{P_{\text{TE}}}{P_{\text{LE}}}\right)^{(\gamma-1)/\gamma} - 1}{\frac{T_{\text{TE}}}{T_{\text{LE}}} - 1}$$
(B8)

Momentum-rise efficiency

$$\eta_{\text{mom}} = \frac{\left(\frac{P_{\text{TE}}}{P_{\text{LE}}}\right)^{(\gamma-1)/\gamma} - 1}{\frac{(UV_{\theta})_{\text{TE}} - (UV_{\theta})_{\text{LE}}}{T_{\text{LE}}gJC_{n}}}$$
(B9)

Equivalent weight flow

$$\frac{W\sqrt{\theta}}{\delta}$$
 (B10)

Equivalent rotative speed

$$\frac{N}{\sqrt{\theta}}$$
 (B11)

Weight flow per unit annulus area

$$\frac{\left(\frac{\mathbf{W}\sqrt{\theta}}{\delta}\right)}{\mathbf{A}_{\mathbf{a}\mathbf{n}}} \tag{B12}$$

Weight flow per unit frontal area

$$\underbrace{\begin{pmatrix} \mathbf{W}\sqrt{\theta} \\ \mathbf{\delta} \end{pmatrix}}_{\mathbf{A_f}} \tag{B13}$$

Head-rise coefficient

$$\frac{C_{p}T_{LE}}{U_{tip}^{2}}\left[\left(\frac{P_{TE}}{P_{LE}}\right)^{(\gamma-1)/\gamma} - 1\right]$$
(B14)

Flow coefficient

$$\left(\frac{V_z}{U_{tip}}\right)_{LE}$$
 (B15)

Stall margin

$$SM = \left[\frac{\left(\frac{P_{TE}}{P_{LE}} \right)_{stall}}{\left(\frac{P_{TE}}{P_{LE}} \right)_{ref}} \times \frac{\left(\frac{W\sqrt{\theta}}{\delta} \right)_{ref}}{\left(\frac{W\sqrt{\theta}}{\delta} \right)_{stall}} - 1 \right] 100$$
(B16)

Polytropic efficiency

$$\eta_{p} = \frac{\ln\left(\frac{P_{TE}}{P_{LE}}\right)^{(\gamma-1)/\gamma}}{\ln\left(\frac{T_{TE}}{T_{LE}}\right)}$$
(B17)

APPENDIX C

DEFINITIONS AND UNITS USED IN TABLES

ABS absolute

AERO CHORD straight line between blade leading and trailing edges along design

streamline, cm

AREA RATIO ratio of actual flow area to critical area (where local Mach number

is one)

BETAM meridional air angle, deg

CONE ANGLE angle between axial direction and conical surface representing blade

element, deg

DEV deviation angle (defined by eq. (B2)), deg

D-FACT diffusion factor (defined by eq. (B3))

EFF adiabatic efficiency (defined by eq. (B8))

IN inlet (leading edge of blade)

INCIDENCE incidence angle (mean defined by eq. (B1)), deg

KIC angle between the blade mean camber line at the leading edge and the

meridional plane, deg

KOC angle between the blade mean camber line at the trailing edge and the

meridional plane, deg

KTC angle between the blade mean camber line at the transition point and

the meridional plane, deg

LOSS COEFF loss coefficient (total defined by eq. (B4) and profile defined by

eq. (B5))

LOSS PARAM loss parameter (total defined by eq. (B6) and profile defined by

eq. (B7))

MERID meridional

MERID VEL R meridional velocity ratio

OUT outlet (trailing edge of blade)

PERCENT SPAN percent of blade span from tip at rotor outlet

PHISS suction surface camber ahead of assumed shock location, deg

PRESS pressure, N/cm²

PROF profile

RADII radius, cm

REL relative to the blade

RI inlet radius (leading edge of blade), cm

RO outlet radius (trailing edge of blade), cm

RP radial position

RPM equivalent rotative speed, rpm

SETTING ANGLE angle between aerodynamic chord and meridional plane, deg

SOLIDITY ratio of aerodynamic chord to blade spacing

SPEED speed, m/sec

STREAMLINE SLOPE slope of streamline, deg

TANG tangential

TEMP temperature, K

TI thickness of blade at leading edge, cm

TM thickness of blade at maximum thickness, cm

TO thickness of blade at trailing edge, cm

TOT total

TOTAL CAMBER difference between inlet and outlet blade mean camber lines, deg

VEL velocity, m/sec

WT FLOW equivalent weight flow, kg/sec

X FACTOR ratio of suction surface camber ahead of assumed shock location

of a multiple circular arc blade section to that of a double

circular arc blade section

ZIC axial distance to blade leading edge from inlet, cm

ZMC axial distance to blade maximum thickness point from inlet, cm

ZOC axial distance to blade trailing edge from inlet, cm

ZTC axial distance to transition point from inlet, cm

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TABLE I. - DESIGN OVERALL PARAMETERS

FOR STAGE 55-55

ROTOR TOTAL PRESSURE RATIO	1,205
STAGE TOTAL PRESSURE RATIO	1.196
ROTOR TOTAL TEMPERATURE RATIO	1.058
STAGE TOTAL TEMPERATURE RATIO	1.058
ROTOR ADIABATIC EFFICIENCY	0.940
STAGE ADIABATIC EFFICIENCY	0.903
ROTOR POLYTROPIC EFFICIENCY	0.941
STAGE POLYTROPIC EFFICIENCY	0.906
ROTOR HEAD RISE COEFFICIENT	0.348
STAGE HEAD RISE COEFFICIENT FLOW COEFFICIENT	0.334
FLOW COEFFICIENT	0.861
	153.970
HT FLOW PER UNIT ANNULUS AREA	195.295
NT FLOW	31,.207
RPM 8	020.000
TIP SPEED	213.323

TABLE II. - DESIGN BLADE-ELEMENT PARAMETERS FOR ROTOR 55

RP TIP 1 2 3 4 5 6 7 8 9 HVB	RADI IN 25.400 2 24.730 2 24.026 2 23.323 2 21.172 2 18.320 1 15.539 1 13.541 1 12.288 1 11.684 1	0UT 25.400 24.714 24.028 23.343 21.285 8.542 25.799 3.741 3.056 2.370	ABS IN 0. -0. -0. -0. -0. -0. -0.	BETAM OUT 27.6 28.8 29.7 30.4 31.6 32.9 34.7 36.1 37.6	REL 1N 48.4 47.8 47.2 46.5 44.1 40.2 35.7 32.0 30.7 29.4 28.1	BETAM OUT 38.1 34.9 32.1 29.7 24.1 16.6 7.9 1.4 -0.7 -2.8 -4.8	TOTA 1N 288.2 288.2 288.2 288.2 288.2 288.2 288.2 288.2 288.2 288.2	L TEMP RATIO 1.063 1.065 1.067 1.064 1.057 1.057 1.044 1.042 1.040	TOTAL IN 10.14 10.14 10.14 10.14 10.14 10.14 10.14 10.14 10.14 10.14 10.14 10.14 10.14 10.14 10.14	PRESS RATIO 1,213 1,226 1,235 1,238 1,231 1,208 1,144 1,130 1,115 1,098
RP TIP 1 2 3 4 5 6 7 8 9 HUB	ABS IN 189.4 188.1 186.9 185.9 183.6 181.8 181.3 182.0 182.6 183.2 183.9	VEL 0UT 184.1 190.0 194.1 196.3 197.6 196.3 184.1 189.8 187.2 184.1 180.4	REL IN 285.3 280.2 275.0 270.1 255.6 238.2 223.4 214.7 212.3 210.3 208.5	VEL OUT 207.3 203.0 198.9 194.9 184.4 172.0 153.5 150.4 147.0 143.4	MER1 1N 189.4 188.1 186.9 185.9 181.6 181.3 182.0 182.6 183.2	0 YEL 0UT 163.1 166.5 168.6 169.3 168.3 164.8 159.9 153.5 150.4 146.9	TAN IN 0. 000000. 0. 0. 0.	G VEL OUT 85.3 91.5 96.2 99.4 103.6 106.6 111.7 111.5 111.0	WHEEL IN 213.3 207.7 201.8 195.9 177.8 153.9 130.5 113.7 108.4 103.2 98.1	SPEED 0UT 213.3 207.6 201.8 196.0 178.8 155.7 132.7 115.4 109.6 103.9 98.1
RP TIP 1 2 3 4 5 6 7 8 9 HUB	ABS MA IN 0.575 0.570 0.567 0.563 0.556 0.550 0.551 0.555 0.555	0.540 0.557 0.570 0.577 0.577 0.582 0.576 0.563 0.555 0.546 0.535	REL M. IN 0.865 0.850 0.834 0.818 0.774 0.721 0.676 0.650 0.643 0.631	ACH NO OUT 0.608 0.595 0.584 0.573 0.543 0.543 0.478 0.455 0.446 0.436	MERID M IN 0.575 0.570 0.567 0.563 0.556 0.554 0.554 0.553 0.553 0.555	ACH NO OUT 0.478 0.485 0.495 0.497 0.496 0.487 0.473 0.455 0.446 0.436	STREAML I IN 0.78 0.66 0.61 0.62 0.85 1.26 1.39 1.04 0.78 0.78	NE SLOPE OUT 0.46 0.55 0.66 0.79 1.14 1.43 1.40 0.98 0.71 0.37 -0.03	MERIP VEL R 0.861 0.885 0.902 0.911 0.917 0.907 0.882 0.843 0.824 0.802	
RP TIP 1 2 3 4 5 6 7 8 9 HUB	PERCENT SPAN 0. 5.00 10.00 15.00 30.00 50.00 70.00 85.00 90.00 95.00	MEAN -2.0 -2.4 -2.9 -3.2 -3.6 -3.7 -3.9	DENCE	DEV 6.1 7.2 8.0 8.5 10.5 12.2 12.6 12.4 12.3 12.2	D-FACT 0.441 0.458 0.470 0.479 0.493 0.503 0.512 0.517 0.520 0.524 0.529	0.903 0.917 0.928 0.936 0.958 0.970 0.949 0.884 0.792 0.724	LOSS C TOT 0.051 0.047 0.043 0.039 0.027 0.019 0.070 0.090 0.116 0.145	OEFF PROF 0.051 0.047 0.043 0.039 0.027 0.019 0.032 0.070 0.090 0.116 0.145	LOSS P TOT 0.023 0.022 0.020 0.019 0.013 0.009 0.015 0.031 0.039 0.049 0.059	PRAM PROF 0.023 0.022 0.020 0.019 0.015 0.009 0.015 0.031 0.039 0.049 0.059

TABLE III. - DESIGN BLADE-ELEMENT PARAMETERS FOR STATOR 55

RP TIP 1 2 3 4 5 6 7 8 9 HUB	RADI IN 25.938 2 25.231 2 24.547 2 23.877 2 21.847 2 19.166 1 16.502 1 14.518 1 13.859 1 13.202 1 12.548 1	0UT 5.938 5.299 4.672 4.048 2.222 9.826 7.464 5.682 5.069 4.447	ABS IN 27.9 28.9 29.7 30.3 31.2 32.3 34.0 35.4 35.4 35.9	BETAM OUT -0. 0. -0. -0. -0. -0. -0. -0.	REL IN 27.9 28.9 29.7 30.3 31.2 32.3 34.0 35.4 35.9 36.9	BETAM OUT -0. -0. -0. -0. -0. -0. -0.	TOTA IN 306.2 307.0 307.5 307.5 306.6 304.7 302.7 301.0 300.3 299.6	L TEMP RATIO 1.000 1.000 1.000 1.000 1.000 1.000 1.000 1.000 1.000	TOTAL IN 12.29 12.43 12.51 12.55 12.48 12.24 11.94 11.60 11.45 11.30	PRESS RATIO 0.992 0.993 0.994 0.997 0.996 0.991 0.982 0.979 0.976
RP TIP 1 2 3 4 5 6 7 8 9 HUB	190.0 192.7 194.8 193.0 189.3 182.6 179.3 175.3	VEL 0UT 169.2 175.1 178.9 180.8 179.9 172.7 160.6 143.7 135.4 125.7 114.6	REL IN 178.6 185.4 190.0 192.7 194.8 193.0 189.3 179.3 170.9	VEL 0UT 169.2 175.1 178.9 180.8 179.9 172.7 160.6 143.7 135.4 125.7 114.6	MER I IN 157.9 162.3 165.1 166.5 166.7 163.1 156.9 148.9 145.3 141.2	D VEL 0UT 169.2 175.1 178.9 180.8 179.9 172.7 160.6 143.7 135.4 125.7	IN .83.5 89.6 94.1 97.2 100.9 103.1 105.9 105.8 105.0	G VEL OUT -0. -0. -0. -0. -0. -0. -0. -0.	HHEEL IN 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.	SPEED OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP TIP 1 2 3 4 5 6 7 8 9 HUB	0.543 0.557 0.566 0.573 0.569 0.559 0.540 0.530 0.519	CH NO 0.494 0.511 0.523 0.528 0.526 0.526 0.420 0.420 0.396 0.367 0.334	REL M 1N 0.523 0.543 0.557 0.566 0.573 0.569 0.559 0.540 0.519	ACH NO 0.494 0.511 0.523 0.528 0.526 0.506 0.421 0.420 0.396 0.367 0.334	MERID M IN 0.462 0.475 0.484 0.488 0.490 0.481 0.464 0.440 0.430 0.418 0.404	0.494 0.511 0.523 0.528 0.526 0.526 0.420 0.420 0.334	STREAML! IN 0.63 0.86 1.10 1.34 2.08 3.13 4.25 5.10 5.35 5.58	NE SLOPE OUT -0.10 0.05 0.22 0.39 0.95 1.72 2.47 2.76 2.68 2.54	MERIU VEL R 1.079 1.084 1.086 1.079 1.058 1.024 0.965 0.965 0.839	
RP TIP 1 2 3 4 5 6 7 8 9 HUB	PERCENT SPAN 0. 5.00 10.00 15.00 30.00 50.00 70.00 85.00 90.00 95.00	INCI. MEAN -12.5 -11.6 -10.3 -9.8 -9.1 -7.8 -6.8 -6.4 -6.0 -5.6	DENCE	DEV 16.0 15.6 15.3 15.0 14.0 13.0 11.7 10.9 10.7 10.4	D-FACT 0.380 0.385 0.386 0.387 0.382 0.382 0.440 0.444 0.494	0. 0. 0. 0. 0.	LOSS C TOT 0.049 0.042 0.036 0.030 0.017 0.018 0.046 0.086 0.103 0.123 0.147	0EFF PROF 0.049 0.042 0.036 0.030 0.017 0.018 0.046 0.086 0.103 0.123	LOSS P. TOT 0.034 0.029 0.024 0.019 0.010 0.021 0.035 0.040 0.046 0.052	ARAM PROF 0.034 0.024 0.019 0.010 0.010 0.021 0.035 0.040 0.046 0.052

TABLE IV. - BLADE GEOMETRY FOR ROTOR 55

	MULL			3 020111		- 0-0 -	.0101	
RP TIP 1 2 3 4 5 6 7 8 9 HUB	5. 10. 15. 30. 70. 85. 90.	R1 25.400 24.730 24.026 23.323 21.172 18.320 15.539 13.541 12.907	R0 25.400 24.714 24.028 25.343 21.285 18.542 15.799 13.741 13.056 12.370	KIC 50.40 50.29 50.05 49.67 47.72 43.95 39.62 34.40 32.39 30.27	37.05 35.44 30.64 24.18 17.42 11.69 9.69 7.66	K0C 32.00 27.64 24.05 21.21 13.56 4.41 -4.79 -11.02 -13.01 -14.95	-0. -0.	CONE ANGLE 0.057 0.152 0.152 0.152 0.892 1.806 2.239 1.813 1.379 0.769 0.057
RP TIP 1 2 3 4 5 6 7 8 9 HUB		0.264 0.293 0.326 0.441 0.591 0.741 0.839 0.362 0.881	TO 0.019 0.025 0.031 0.036 0.050 0.063 0.083 0.091	Z! -0.636 -0.671 -0.685 -0.680 -0.659 -0.572 -0.371 -0.206 -0.142	ZMC 2.690 2.650 2.639 2.658 2.648 2.669 2.753 2.824 2.852	2.650 2.639 2.658 2.648 2.669 2.753 2.824 2.852 2.881	Z0 6.522 6.546 6.588 6.644 6.597 6.455 6.284 6.116 6.057	
RP T1P 1 2 3 4 5 6 7 8 9 HUB	AERO CHORD 9.499 9.274 9.105 8.980 8.428 7.703 6.978 6.458 6.290 6.126 5.966	41,14 38,96 37,05 35,44 30,66 24,22 17,48 11,74 9,73 7,69		0.896 0.905 0.919 0.948 0.998 1.063 1.130 1.157	1.000 1.000 1.000 1.000 1.000 1.000 1.000 1.000 1.000	0. 0. -0. -0. -0.	AREA RATIO 0. -0. -0. -0. -0. -0. -0. -0.	

TABLE V. - BLADE GEOMETRY FOR STATOR 55

RP TIP 1 2 3 4 5 6 7 8 9 HUB	5. 10. 15. 30. 70. 85. 90.	RAD R1 25.938 25.231 24.547 23.877 21.847 19.166 16.502 14.518 13.859 13.202 12.548	R0 25.938 25.299 24.672 24.048 22.222 19.826 17.464 15.682 15.069 14.447	K(C 40,40 40,47 40,54 40,61 41,42 41,78 42,13 42,23	18.05 18.23 18.40 19.02 19.69 20.44 20.97 21.15 21.32	K0C -16.01 -15.65 -15.31 -14.98 -14.04 -13.02 -11.73 -10.93	-0. -0. -0. -0. -0. -0.	CONE ANGLE 0.057 0.952 2.087 5.408 6.564 6.832 7.039 7.185
RP TIP 1 2 3 4 5 6 7 8 9 HUB	BLADE TI 0.188 0.188 0.188 0.188 0.188 0.188 0.188 0.188	THICKN TM 0.953 0.953 0.953 0.953 0.953 0.953 0.953 0.953	TO 0.087 0.087 0.087 0.087 0.087 0.087 0.087	Z1 21.634 21.628 21.631 21.642 21.650 21.662 21.673 21.684 21.684	XIAL D ZMC 25.502 25.489 25.490 25.473 25.453 25.404 25.398 25.392 25.387	ZTC 25.502 25.489 25.486 25.490 25.473 25.453 25.426 25.404 25.398 25.392	Z0 31.982 31.967 31.961 31.963 31.937 31.899 31.844 31.787 31.775	
RP TIP 1 2 3 4 5 6 7 8 9 HUB	AERO CHORD 10.584 10.584 10.584 10.585 10.586 10.588 10.588 10.589	SETTING ANGLE 11.92 12.15 12.57 13.28 14.07 15.67 15.88 16.09 16.30	CAMBER 56.40 56.12 55.85 55.59 55.04 54.44 53.51 53.06 52.88	SOLIDIT 0.714 0.733 0.753 0.773 0.841 0.951 1.091 1.228 1.282 1.341 1.406	X FACTOF 1.000 1.000 1.000 1.000 1.000 1.000 1.000 1.000 1.000	PHISS -0. 0. -0. -0. -0. -0. -0. -0.	AREA RATIO -0. 0. -0. -0. -0. -0. -0. -0.	

TABLE VI. - OVERALL PERFORMANCE FOR STAGE 55-55

(a) 80 Percent design speed

Parameter		Reading				
	1606	1600	1598	1597		
ROTOR TOTAL PRESSURE RATIO STAGE TOTAL PRESSURE RATIO ROTOR TOTAL TEMPERATURE RATIO STAGE TOTAL TEMPERATURE RATIO ROTOR TEMP. RISE EFFICIENCY ROTOR MOMENTUM RISE EFFICIENCY ROTOR HEAD RISE COEFFICIENT STAGE HEAD RISE COEFFICIENT FLOW COEFFICIENT HIT FLOW PER UNIT FRONTAL AREA HIT FLOW AT ROTOR INLET HIT FLOW AT ROTOR INLET HIT FLOW AT STATOR OUTLET ROTATIVE SPEED PERCENT OF DESIGN SPEED	1.093 1.071 1.030 1.029 0.869 0.683 0.864 0.255 0.195 1.071 150.66 191.10 30.54 31.88 6436.3	1.112 1.104 1.033 1.032 0.945 0.888 0.996 0.319 0.297 0.834 122.10 154.88 24.75 25.19 26.34 6287.0 78.4	1.116 1.110 1.035 1.035 0.905 0.889 0.381 0.329 0.312 0.650 98.63 125.11 19.99 20.34 20.78 21.86 6293.3	1.117 1.108 1.037 1.036 0.870 0.821 0.829 0.333 0.309 0.574 87.96 111.57 17.83 18.14 18.51 20.55 6283.1		

(b) 90 Percent design speed

Parameter			Reading		
۴. ۱	1604	1596	1595	1594	1588
ROTOR TOTAL PRESSURE RATIO STAGE TOTAL PRESSURE RATIO ROTOR TOTAL TEMPERATURE RATIO STAGE TOTAL TEMPERATURE RATIO ROTOR TEMP. RISE EFFICIENCY ROTOR MOMENTUM RISE EFFICIENCY ROTOR MEAD RISE COEFFICIENT STAGE HEAD RISE COEFFICIENT FLOW COEFFICIENT HT FLOW PER UNIT FRONTAL AREA HT FLOW PER UNIT ANNULUS AREA HT FLOW AT ROTOR INLET HT FLOW AT ROTOR OUTLET HT FLOW AT STATOR OUTLET ROTATIVE SPEED PERCENT OF DESIGN SPEED	1.122 1.097 1.040 1.039 0.845 0.693 0.851 0.261 0.209 1.021 157.81 200.17 33.45 33.46 90.2	1.142 1.132 1.041 0.932 0.888 0.921 0.311 0.290 0.832 135.03 171.27 27.37 27.86 28.42 29.03 7141.1	1.149 1.140 1.044 1.043 0.919 0.887 0.910 0.324 0.733 121.72 154.39 24.67 25.18 25.79 26.68 7146.6	1.151 1.144 1.044 1.044 0.903 0.881 0.328 0.313 142.25 142.25 22.75 23.19 23.69 25.01 7151.6	1.154 1.143 1.048 1.047 0.869 0.835 0.335 0.335 0.609 104.02 131.94 21.47 21.85 23.82 7136.9

(c) 100 Percent design speed

Parameter	Reading					
	1603	1611	1613	1614	1592	
ROTOR TOTAL PRESSURE RATIO STAGE TOTAL PRESSURE RATIO ROTOR TOTAL TEMPERATURE RATIO STAGE TOTAL TEMPERATURE RATIO STAGE TOTAL TEMPERATURE RATIO STAGE TEMP. RISE EFFICIENCY ROTOR MOMENTUM RISE EFFICIENCY ROTOR HEAD RISE COEFFICIENT STAGE HEAD RISE COEFFICIENT FLOM COEFFICIENT HT FLOM PER UNIT FRONTAL AREA HT FLOM PER UNIT ANNULUS AREA HT FLOM AT ROTOR INLET HT FLOM AT ROTOR OUTLET HT FLOM AT STATOR OUTLET ROTATIVE SPEED PERCENT OF DESIGN SPEED	1.162 1.143 1.050 1.049 0.874 0.265 0.265 0.255 0.255 0.959 160.67 203.79 32.56 34.52 34.52 99.0	1.185 1.164 1.054 1.052 0.918 0.852 0.903 0.315 0.282 0.838 147.88 187.58 29.97 31.135 8025.3	1.191 1.177 1.056 1.055 0.907 0.871 0.894 0.328 0.724 131.94 167.36 26.74 27.26 27.93 28.91 7990.1	1:195 1.184 1.059 1.057 0.865 0.865 0.334 0.316 0.357 122.36 155.21 24.23 25.23 25.82 27.50 8001.4	1.193 1.175 1.058 0.864 0.812 0.833 0.336 0.307 0.596 111.83 141.85 22.67 23.66 26.41 7927.8 98.9	

TABLE VI. - Concluded. OVERALL PERFORMANCE

FOR STAGE 55-55

(d) 110 Percent design speed

Parameter		Reading		
	1692	1693	1694	
ROTOR TOTAL PRESSURE RATIO STAGE TOTAL PRESSURE RATIO ROTOR TOTAL TEMPERATURE RATIO STAGE TOTAL TEMPERATURE RATIO ROTOR TEMP. RISE EFFICIENCY STAGE TEMP. RISE EFFICIENCY ROTOR MOMENTUM RISE EFFICIENT STAGE HEAD RISE COEFFICIENT STAGE HEAD RISE COEFFICIENT FLOW COEFFICIENT HIT FLOW PER UNIT FRONTAL AREA HIT FLOW PER UNIT ANNULUS AREA HIT FLOW AT ROTOR INLET HIT FLOW AT ROTOR OUTLET HIT FLOW AT STATOR OUTLET ROTATIVE SPEED PERCENT OF DESIGN SPEED	1.212 1.197 1.064 1.061 0.880 0.858 0.843 0.295 0.276 0.874 162.65 206.31 32.97 33.47 54.42 35.24 8845.4	1.235 1.218 1.068 1.065 0.912 0.899 0.891 0.325 0.304 0.765 148.97 188.96 30.19 30.65 31.44 32.28 8840.4	1.241 1.224 1.072 1.069 0.885 0.866 0.861 0.334 0.312 0.655 132.73 168.36 26.90 27.26 27.94 29.89 8825.6	

(e) 120 Percent design speed

Parameter	ŧ	Reading	
	1697	1696	1695
ROTOR TOTAL PRESSURE RATIO STAGE TOTAL PRESSURE RATIO ROTOR TOTAL TEMPERATURE RATIO STAGE TOTAL TEMPERATURE RATIO ROTOR TEMP. RISE EFFICIENCY STAGE TEMP. RISE EFFICIENCY ROTOR MOMENTUM RISE EFFICIENCY ROTOR HEAD RISE COEFFICIENT. STAGE HEAD RISE COEFFICIENT. FLOW COEFFICIENT WIT FLOW PER UNIT FRONTAL AREA HIT FLOW PER UNIT ANNULUS AREA HIT FLOW AT ROTOR INLET HIT FLOW AT ROTOR OUTLET HIT FLOW AT STATOR OUTLET ROTATIVE SPEED PERCENT OF DESIGN SPEED	1.251 1.236 1.079 1.075 0.839 0.835 0.807 0.295 0.278 0.816 163.80 207.77 33.20 33.71 34.85 35.94 9582.6 119.5	1.282 1.267 1.084 1.080 0.879 0.879 0.851 0.327 0.312 0.746 154.87 196.44 31.39 31.87 32.54 34.38 9587.2	1.290 1.269 1.086 1.082 0.883 0.857 0.855 0.313 0.677 144.85 183.73 29.36 29.79 30.37 32.92 9598.7 119.7

TABLE VII. - BLADE-ELEMENT DATA AT BLADE EDGES FOR ROTOR 55

(a) 80 Percent design speed; reading 1597

			• .	,	J	_		
RADII 1N OUT 24.729 24.714 24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 15.799 13.541 13.741 12.906 13.056 12.289 12.370	0. 0.0 0.0	52.9 47.7 45.1	59.7 58.2 57.9	00T 30.7 37.4 35.5 28.4 19.1 9.9 3.6 0.1	1N 288.7 288.5 288.3 288.1 288.0 288.0 287.9 287.9	RAT10 1.051 1.047 1.045 1.040 1.035 1.030 1.025 1.025	IN 10.08 10.13 10.14 10.14 10.14 10.14 10.14	PRESS RATIO 1.135 1.127 1.124 1.132 1.122 1.105 1.081 1.078
ABS VEL IN OUT 95.1 125.4 97.8 125.8 96.2 126.8 96.6 134.1 96.2 133.9 96.3 131.4 96.6 123.3	REL VE [N (188.7) 188.7) 185.7 16 181.1 16 169.6 16 154.1 16 140.7 16 131.4 6	EL 0UT 98.3 96.5 10.0 19.7 08.3 00.0	MERID 10 95.1 97.8 96.2 96.6 96.2 96.3 96.6 96.0	75.6 84.6 89.6 105.3 102.3 98.5 89.6 88.2	0 0.0 0.0 0.0 0.0	100.0 93.1 89.8 83.1 .86.4 87.0	103.0 157.8 153.4 139.3 120.4	SPEED OUT 162.9 157.8 153.6 140.1 121.8 104.3 90.4 85.7 81.4
ABS MACH NO IN OUT 0.281 0.364 0.290 0.366 0.285 0.369 0.286 0.392 0.285 0.393 0.286 0.386 0.286 0.363 0.284 0.362 0.279 0.366	IN 0.558 0.0.550 0.550 0.502 0.457 0.417 0.389 0.379 0.379	0UT .285 .310 .320 .350 .318 .294 .264	IN	OLT 0.219 0.246 0.261 0.308 0.300 0.290 0.263 0.260 0.259			1.022 0.927 0.920	
SPAN MEAN 5.00 9.4 10.00 8.2 15.00 7.5 50.00 7.4 70.00 7.2	' 1 1 1 1	2.1 3.4 4.3 4.9 5.0 4.7 4.6	0.775 0.703 0.663 0.553 0.580 0.583 0.605 0.601	0.718 0.741 0.749 0.902 0.961 0.980 0.899 0.880	TOT 0.248 0.215 0.212 0.082 0.034 0.018 0.087 0.107	PROF 0.248 0.215 0.212 0.082 0.034 0.018 0.087 0.107	TOT 0.107 0.094 0.094 0.038 0.016 0.008 0.038 0.046	ARAM PROF 0.107 0.094 0.038 0.016 0.008 0.038 0.046 0.031
	IN OUT 24.729 24.714 24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 13.056 12.289 12.370 ABS VEL IN OUT 95.1 125.4 97.8 125.8 96.6 134.1 96.2 133.9 96.3 131.4 96.2 133.9 96.3 131.4 96.2 133.9 96.3 131.4 96.2 133.9 96.3 131.4 96.2 133.9 96.3 131.4 96.2 133.9 96.3 132.3 06.0 122.9 94.0 124.3 ABS MACH NO IN OUT 0.281 0.366 0.285 0.369 0.286 0.392 0.285 0.366 0.286 0.366 0.285 0.369 0.286 0.366 0.286 0.366 0.286 0.366 0.286 0.366 0.286 0.363 0.284 0.362 0.279 0.366 PERCENT INCI SPAN MEAN 5.00 9.4 10.00 8.2 30.00 7.4 70.00 7.4 70.00 7.4 70.00 8.2 30.00 9.0	24.729 24.714	24.729 24.714 0. 52.9 24.026 24.028 0.0 47.7 23.322 23.343 0.0 45.1 21.173 21.285 0. 38.3 18.321 18.542 0.0 40.2 15.540 15.799 0.0 41.5 13.541 13.741 0.0 43.4 12.906 13.056 0.0 44.1 12.289 12.370 -0.0 44.9 ABS VEL REL VEL IN OUT IN OUT 95.! 125.4 188.7 98.3 97.8 125.8 185.7 106.5 96.2 126.8 181.1 110.0 96.6 134.1 169.6 119.7 96.2 133.9 154.1 108.3 96.3 131.4 140.7 100.0 96.6 123.3 131.4 89.7 96.0 122.9 128.0 88.2 ABS MACH NO REL MACH NO IN OUT IN OUT 0.281 0.364 0.558 0.285 0.290 0.366 0.550 0.310 0.285 0.369 0.556 0.320 0.286 0.369 0.556 0.320 0.286 0.369 0.556 0.320 0.286 0.369 0.556 0.320 0.286 0.369 0.550 0.310 0.285 0.369 0.556 0.320 0.286 0.366 0.550 0.310 0.285 0.369 0.550 0.310 0.285 0.369 0.550 0.310 0.286 0.366 0.550 0.310 0.286 0.366 0.550 0.310 0.285 0.366 0.550 0.310 0.286 0.366 0.550 0.310 0.286 0.366 0.550 0.310 0.286 0.366 0.550 0.310 0.285 0.369 0.556 0.320 0.286 0.368 0.260 PERCENT INCIDENCE DEV SPAN MEAN 5.00 9.4 12.1 10.00 8.2 13.4 15.00 8.2 14.3 30.00 7.5 14.9 50.00 7.4 15.0 70.00 7.2 14.7 85.00 8.3 14.6 90.00 9.0 13.1	IN OUT IN OUT IN 24.729 24.714 0. 52.9 59.7 24.026 24.028 0.0 47.7 58.2 23.322 23.345 0.0 45.1 57.9 21.175 21.285 0. 38.3 55.5 18.321 18.542 0.0 40.2 51.4 15.540 15.799 0.0 41.5 46.8 13.541 13.741 0.0 43.4 42.7 12.906 13.056 0.0 44.1 41.4 12.289 12.370 -0.0 44.9 40.7 ABS VEL REL VEL MERIO IN OUT IN OUT IN OUT IN OUT IN OUT IN OUT IN 95.1 125.4 188.7 98.3 95.1 97.8 125.8 185.7 106.5 97.8 96.2 126.8 181.1 110.0 96.2 96.6 134.1 169.6 119.7 96.6 96.2 133.9 154.1 108.3 96.2 96.6 122.9 128.0 88.2 96.0 94.0 124.3 124.0 88.2 94.0 ABS MACH NO REL MACH NO MERIO IN OUT OLSE 0.366 0.356 0.350 0.286 0.285 0.281 0.286 0.386 0.457 0.318 0.285 0.286 0.285 0.389 0.536 0.320 0.286 0.286 0.386 0.417 0.294 0.286 0.286 0.386 0.417 0.294 0.286 0.286 0.386 0.417 0.294 0.286 0.286 0.386 0.417 0.294 0.286 0.284 0.362 0.379 0.260 0.284 0.286 0.365 0.389 0.264 0.284 0.286 0.284 0.362 0.379 0.260 0.284 0.279 0.366 0.368 0.260 0.279 PERCENT INCIDENCE DEV D-FACT SPAN MEAN 5.00 9.4 12.1 0.775 14.9 0.553 15.00 8.2 14.3 0.663 30.00 7.5 14.9 0.553 50.00 7.4 15.0 0.580 7.00 7.2 14.7 0.583 85.00 8.3 14.6 0.605 90.00 9.0 13.1 0.601	24.729 24.714	24.729 24.714	24.729 24.714	24, 729 24, 714

TABLE VII. - Continued. BLADE-ELEMENT DATA AT BLADE EDGES

FOR ROTOR 55

(b) 80 Percent design speed; reading 1598

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 24.729 24.714 24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 15.799 13.541 13.741 12.906 13.056 12.289 12.370	ABS 8 1N 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.	38.6 35.1 34.3 34.8 37.5 39.2 42.0 42.9 43.5	REL IN 56.2 54.7 54.4 51.7 47.9 43.5 39.5 38.6 37.8	BETAM 0UT 40.2 37.2 34.8 28.4 19.1 9.4 1.8 -0.5 -4.9	TOTA IN 288.7 288.6 288.3 288.1 288.0 287.9 287.9 287.8 287.8	L TEMP RATIO 1.043 1.041 1.037 1.034 1.030 1.026 1.025 1.025	TOTAL IN 10.07 10.13 10.14 10.14 10.14 10.14 10.14	PRESS RATIO 1.126 1.126 1.129 1.128 1.119 1.106 1.083 1.076 1.080
RP 1 2 3 4 5 6 7 8 9	ABS VEL IN OUT 109.0 126.8 111.7 132.1 110.0 135.2 110.1 138.2 109.1 138.4 108.5 137.7 108.1 130.6 106.7 127.7 104.4 129.9	193.4 188.9 177.8 162.8	0UT . 129.7 135.5 136.0 129.1 116.3 108.3 97.1 93.6	110.0 110.1 109.1 108.5	99.1 108.0 111.7 113.5 109.9 106.8 97.1	TAN !N 0.0 0.0 -0.0 0.0 -0.0 0.0	VEL 0UT 79.0 76.0 76.2 78.9 84.2 86.9 87.3 86.9 89.5	162.8 157.9 153.6 139.6 120.8 103.0 89.0	SPEED 0UT 162.7 158.0 153.7 140.4 122.3 104.7 90.3 86.1 81.5
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.323 0.369 0.332 0.386 0.327 0.495 0.324 0.407 0.324 0.405 0.321 0.385 0.317 0.376 0.310 0.383	0.574 0.561 0.528 0.483 0.444 0.416 0.405	CH NO OUT 0.378 0.396 0.398 0.379 0.342 0.319 0.286 0.276	0.321	0.289 0.315 0.327 0.333 0.323 0.314 0.286 0.276 0.278			MERID VEL R 0.909 0.967 1.016 1.031 1.007 0.984 0.898 0.877 0.903	
RP 1 2 3 4 5 6 7 8 9	PERCENT INC SPAN MEAN 5.00 5.9 10.00 4.7 15.00 4.7 30.00 4.0 70.00 3.9 85.00 5.1 90.00 6.2 95.00 7.5		DEV 12.5 13.1 13.5 14.9 15.0 14.2 12.8 12.5 10.1	0.516 0.499 0.509 0.546 0.552 0.584 0.591	EFF 0.808 0.848 0.860 0.936 0.960 0.976 0.839 0.839	LOSS C TOT 0.134 0.103 0.099 0.047 0.031 0.020 0.086 0:129 0.108	PROF 0.134 0.103 0.099	LOSS P TOT 0.057 0.045 0.044 0.022 0.015 0.009 0.038 0.056 0.045	PROF 0.057 0.045 0.044

FOR ROTOR 55

(c) 80 Percent design speed; reading 1600

RP 1 23 4 5 6 7 8 9	RADII IN OUT 24.729 24.714 24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 15.799 13.541 13.741 12.906 13.056 12.289 12.370	0.0 2: 0.0 2: 0.0 2: 0.0 2: 0.0 2: 0.0 3: 0.0 3:	AM REL UT IN 4.3 49.6 3.5 47.7 4.1 47.5 6.6 44.7 6.6 40.6 9.6 36.5 1.6 33.1 4.3 31.9 6.2 30.8	BETAM OUT 37.5 34.3 32.4 26.3 20.0 10.8 3.3 -0.3 -2.5	IN 288.9 288.8	TEMP RAT I 0 1.034 1.035 1.035 1.034 1.032 1.029 1.027 1.025	TOTAL IN 10.02 10.13 10.14 10.14 10.15 10.14 10.14	PRESS RATIO 1.124 1.121 1.118 1.117 1.116 1.107 1.098 1.086 1.068
RP 1 23 4 5 6 7 8 9	ABS VEL 1N 0UT 138.3 146.2 143.5 154.1 140.6 157.5 141.2 157.5 140.8 157.9 138.5 157.8 137.1 158.2 136.9 154.4 135.9 146.8	REL VEL 1N 0U 213.4 167 213.3 171 207.9 167 198.5 157 185.3 150 172.2 139 163.7 134 161.3 127 158.2 118	.9 138.3 .2 143.5 .7 140.6 .1 141.2 .2 140.8 .7 138.5 .9 137.1 .7 136.9	D VEL OUT 133.2 141.4 141.6 140.8 141.2 137.2 134.7 127.7 118.5	TANC 1N 0.0 0.0 0.0 0.0 0.0 0.0 0.0	VEL 0UT 60.2 61.3 63.3 70.6 70.7 77.9 83.0 86.9 86.6	WHEEL IN 162.5 157.9 153.2 139.5 120.6 102.3 89.4 85.2 81.0	SPEED 0UT 162.4 157.9 153.4 140.3 122.0 104.0 90.7 86.2 81.5
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.413 0.430 0.429 0.454 0.420 0.457 0.423 0.465 0.421 0.466 0.414 0.467 0.410 0.469 0.409 0.458 0.406 0.434	REL MACH IN OU 0.637 0.4 0.638 0.5 0.621 0.4 0.594 0.4 0.554 0.4 0.489 0.4 0.482 0.3 0.473 0.3	T {N 93 0.413 0.429 94 0.420 664 0.423 4.44 0.411 4.3 0.414 0.0 0.410 78 0.409	OUT 0.391 0.416 0.417 0.416 0.417 0.406 0.399 0.378 0.351			MERID VEL R 0.963 0.985 1.007 0.997 1.003 0.990 0.982 0.932 0.872	
RP 1 2 3 4 5 6 7 8 9	PERCENT SPAN MEAN 5.00 -0.7 10.00 -2.3 15.00 -3.1 50.00 -3.4 70.00 -3.2 85.00 -0.5 95.00 0.5	DENCE DI 9 10 11 12 15 15 14 12	.3 0.356 .2 0.359 .8 0.397 .8 0.382 .6 0.404 .3 0.402 .7 0.443	0.988 0.968 0.931 0.956 0.948 0.940 0.890 0.759	LOSS CO TOT 0.006 0.015 0.036 0.024 0.031 0.046 0.039 0.069 0.145	PROF 0.006 0.015 0.036 0.024 0.031 0.046 0.039 0.069 0.145	LOSS P. TOT 0.003 0.007 0.016 0.011 0.015 0.021 0.017 0.030 0.061	PR0F

FOR ROTOR 55

(d) 80 Percent design speed; reading 1606

		• •			-	•	_			
RP 1 2 3 4 5 6 7 8 9	RAD IN 24.729 ; 24.026 ; 23.322 ; 21.173 ; 18.321 ; 15.540 ; 13.541 ; 12.906 ; 12.289	24.714 24.028 23.343 21.285	0.0	16.1 15.6 16.0 18.2	42.6 41.0 40.6 37.7		TOTA IN 289.1 289.0 288.6 287.8 287.8 287.8 287.8	L TEMP RATIO 1.029 1.030 1.031 1.032 1.032 1.030 1.026 1.023 1.022	TOTAL IN 9.95 10.13 10.14 10.15 10.15 10.15 10.14 10.09	PRESS RATIO 1.096 1.095 1.097 1.102 1.103 1.098 1.073 1.047
RP 1 2 3 4 5 6 7 8 9	!N 181.6 186.2 182.6 184.2 184.1 184.8 185.9 184.0	177.6 188.7 191.7 196.8 200.8 203.1 196.5 184.7	240.5 232.9 221.7 212.5 207.2	207.2 213.1 211.4 204.0 191.7 185.7 177.8 165.6	186.2 182.6 184.2 184.1 184.8 185.9	D VEL OUT 170.6 181.8 184.3 186.9 185.5 184.4 177.6 165.5 156.4	IN 0.0 0.0 0.0 0.0 0.0 0.0	83.9 82.0 81.5	IN 167.0 161.7 156.5 142.5 123.5 104.9 91.4 87.2 82.8	SPEED 0UT 166.8 161.7 156.6 143.3 125.0 106.7 92.8 88.3 83.4
RP 1 2 3 4 5 6 7 8 9	ABS MA IN 0.549 0.564 0.552 0.558 0.560 0.564 0.558	0.528 0.563 0.572 0.589 0.602 0.610 0.590 0.553 0.527	REL M IN 0.745 0.746 0.727 0.706 0.672 0.644 0.628 0.617 0.600	OUT 0.616 0.635 0.631 0.610 0.575 0.558 0.534	MERID M IN 0.549 0.552 0.558 0.558 0.560 0.564 0.558	OUT 0.507 0.542 0.550 0.559 0.556 0.554	•		MERID VEL R 0.939 0.976 1.009 1.014 1.008 0.998 0.955 0.968	
RP 1 2 3 4 5 6 7 8 9	PERCENT SPAN 5.00 10.00 15.00 30.00 50.00 70.00 85.00 90.00	INC II MEAN -7.7 -9.1 -10.0 -10.1 -10.1 -8.2 -7.0 -5.6		0EV 6.9 7.4 8.2 10.0 10.4 11.5 13.9 15.2	D-FACT 0.271 0.249 0.240 0.264 0.310 0.316 0.322 0.362 0.385	0.915 0.878 0.874 0.895 0.902 0.905 0.798	LOSS C TOT 0.027 0.040 0.044 0.040 0.040 0.040 0.075 0.151 0.194	PROF 0.027 0.040 0.044 0.040 0.040 0.040 0.075 0.151	LOSS PATOT 0.012 0.019 0.020 0.019 0.033 0.065 0.082	ARAM PROF 0.012 0.019 0.021 0.019 0.033 0.065 0.082

FOR ROTOR 55

(e) 90 Percent design speed; reading 1588

		(0)	00 - 0.		-2-6 2	, -				
RP 1 2 3 4 5 6 7 8 9	RAD I IN 24.729 2 24.026 2 23.322 2 21.173 2 18.321 1 15.540 1 13.541 1 12.906 1	0UT 4.714 4.028 3.343 1.285 8.542 5.799 3.741 3.056 2.370	ABS IN 0.0 0.0 0.0 0.0 0.0 0.0	BETAM OUT 47.9 42.3 40.3 37.2 40.0 41.1 43.7 44.2 45.2	REL 1N 58.2 56.5 56.2 53.5 49.7 45.3 41.2 40.0 39.3	37.7 35.0	288.4 288.0 288.0 287.9 287.9 287.8	1.058	10.13 10.14 10.14 10.14 10.14 10.14	1.165 1.167 1:175
RP 1 2 3 4 5 6 7 8 9	114.5 118.8 116.8 116.9	140.3 144.3 147.9 155.4 152.1 152.1 143.1 142.0	217.1 215.2 209.8 196.5 179.4 163.8 153.7 149.9	123.6 134.9 137.6 139.7 123.4 116.1	114.5 118.8 116.8	VEL 0UT 94.0 106.8 112.8 123.7 116.5 114.6 103.4 101.8 101.2	0.	104.1 97.0 95.6 94.0 97.8 99.9 98.9	184.5 179.5 174.3 158.0 136.9 116.4 101.3 96.4	184.4 179.5 174.5 158.8 138.5 118.4 102.8
0. 123 4 15 6 7 8 9	0.353 0.347 0.348 0.345 0.343 0.344	CH NO 0.406 0.419 0.454 0.454 0.446 0.447 0.421 0.423	IN 0.645 0.640 0.624 0.585 0.534 0.487 0.457		IN 0.340 0.353 0.347 0.348 0.345	0UT 0.272 0.310			MERID VEL R 0.821 0.899 0.966 1.059 1.003 0.995 0.894 0.888 0.900	
RP 1 2 3 4 5 6 7 8 9	PERCENT SPAN 5.00 10.00 15.00 30.00 50.00 70.00 85.00 90.00 95.00	1NC1E MEAN 7.9 6.5 6.5 5.8 5.7 6.8 7.0	·.	12.9 13.6 13.8	D-FACT 0.698 0.622 0.592 0.542 0.587 0.580 0.614 0.607 0.597	0.724 0.765 0.782 0.911 0.946 0.966 0.877 0.847 0.900	0.074 0.046 0.030 0.105 0.134	PROF 0.232 0.185 0.177 0.074 0.046 0.030 0.105 0.134	0.022 0.014 0.046	PROF 0.098 0.081

FOR ROTOR 55

(f) 90 Percent design speed; reading 1594

RP ! 2 3 4 5 6 7 8 9	RADII IN OUT 24.729 24.714 24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 15.799 13.541 13.741 12.906 13.056 12.289 12.370	1N 0.0 -0.0	BETAM OUT 38.2 34.7 33.7 34.4 37.7 39.0 42.1 43.1	IN 55.6 54.1 53.8 51.1 47.3	BETAM 0UT 40.1 37.2 34.6 28.3 19.3 8.9 1.4 -0.9 -5.4	TOTA 1N 288.8 288.7 288.3 288.0 287.9 287.9 287.8	RAT 10 1.055 1.053 1.053 1.049 1.043 1.038 1.034 1.033	TOTAL IN 10.04 10.14 10.14 10.14 10.14 10.14 10.13	PRESS RATIO 1.166 1.164 1.169 1.153 1.153 1.109 1.100
RP 1 23 4 5 6 7 8 9	ABS VEL IN OUT 126.9 144.5 130.0 150.7 127.9 154.7 128.4 158.1 126.6 156.3 125.4 157.2 125.2 149.4 123.6 145.9 121.1 148.9	216.3 204.3 186.6 171.0 161.0 157.0	VEL 0UT 148.4 155.5 156.4 148.2 131.0 123.6 110.9 106.5 108.0	MER II IN 126.9 130.0 127.9 128.4 126.6 125.4 125.2 123.6 121.1	D VEL 0UT 113.6 123.9 128.7 130.4 123.7 122.1 110.9 106.5	IN 0.0 -0.0 0.0 0.0	G VEL OUT 89.3 85.8 85.8 99.0 100.1 99.8	WHEEL IN 185.0 179.8 174.4 158.9 137.1 116.3 101.4 96.9 92.1	SPEED OUT 184.9 179.8 174.6 159.7 138.8 118.2 102.9 98.0 92.7
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO 1N OUT 0.378 0.420 0.387 0.439 0.381 0.452 0.383 0.463 0.377 0.459 0.374 0.463 0.368 0.430 0.368 0.439	0.661 0.645 0.609 0.556 0.510 0.480 0.468	OUT 0.432 0.453 0.457 0.434 0.385 0.364 0.327 0.314 0.318	MERID M IN 0.378 0.387 0.381 0.383 0.377 0.374 0.373 0.368 0.361	ACH NO OUT 0.330 0.361 0.376 0.382 0.363 0.360 0.327 0.314 0.317			MERID VEL R 0.895 0.953 1.007 1.015 0.977 0.974 0.886 0.862 0.888	-
RP 1 2 3 4 5 6 7 8 9	PERCENT INC SPAN MEAN 5.00 5.3 10.00 4.1 15.00 3.3 50.00 3.3 70.00 3.2 85.00 4.6 90.00 5.7 95.00 7.0	I DENCE	DEV 12.5 13.1 13.4 14.8 15.1 13.7 12.4 12.1 9.5	D-FACT 0.560 0.513 0.493 0.506 0.556 0.552 0.588 0.598	0.808 0.835 0.865 0.929 0.959 0.982 0.838 0.838	LOSS C TOT 0.135 0.114 0.096 0.052 0.032 0.015 0.087 0.129 0.098	OEFF PROF 0.135 0.114 0.096 0.052 0.032 0.035 0.087 0.129 0.098	LOSS P. TOT 0.057 0.050 0.043 0.024 0.015 0.007 0.039 0.056 0.041	ARAM PROF 0.057 0.050 0.043 0.024 0.015 0.007 0.039 0.056 0.041

FOR ROTOR 55

(g) 90 Percent design speed; reading 1595

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 24.729 24.714 24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 15.799 13.541 13.741 12.906 13.056 12.289 12.370	ABS BETAM IN OUT 0.0 31. 0.0 29. 0.0 31. 0.0 34. 0.0 36. 0.0 39. 0.0 41.	IN OUT 3 53.0 39.6 51.4 36.3 1 51.0 33.9 27.6 3 44.4 18.3 44.4 18.3 46.1 8.5 36.4 0.8 0 35.4 -1.3	TOTAL TEMP IN RATIO 288.8 1.051 288.8 1.049 288.5 1.049 288.0 1.047 287.9 1.038 287.9 1.034 287.8 1.033 287.7 1.032	TOTAL PRESS IN RATIO 10.02 1.162 10.13 1.163 10.14 1.152 10.15 1.136 10.14 1.152 10.15 1.136 10.14 1.191 10.14 1.098
RP 1 23 4 5 67 8 9	ABS VEL 1N 0UT 139.8 151.0 144.0 159.6 141.6 163.1 141.3 165.0 139.9 166.0 138.1 165.0 137.2 159.2 135.6 153.0 133.1 154.3	REL VEL IN OUT 232.0 167.4 230.6 173.2 245.0 171.8 212.4 159.2 196.0 144.3 180.5 133.9 170.5 123.2 166.4 115.5	144.0 139.5 141.6 142.6 141.3 141.1 139.9 137.1 138.1 132.5 137.2 123.2 135.6 115.5	TANG VEL IN OUT 0.0 78.4 0.0 77.5 0.0 79.2 0.0 85.6 0.0 93.6 0.0 98.3 0.0 100.9 0.0 100.4 0.0 102.0	WHEEL SPEED IN OUT 185.2 185.0 180.2 180.2 174.9 175.1 158.6 159.4 137.2 138.9 116.2 118.2 101.1 102.6 96.5 97.7 91.7 92.3
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.417 0.441 0.430 0.467 0.423 0.478 0.423 0.485 0.419 0.489 0.413 0.487 0.410 0.470 0.405 0.452 0.397 0.456	REL MACH NO IN OUT 0.693 0.489 0.507 0.673 0.504 0.586 0.425 0.540 0.396 0.510 0.364 0.497 0.341 0.483 0.343	IN OUT 0.417 0.377 0.430 0.408 0.423 0.418 0.423 0.415 0.419 0.404 0.413 0.391 0.410 0.364 0.405 0.341		MERID VEL R 0.923 0.969 1.007 0.998 0.979 0.859 0.852 0.870
RP 123456789	PERCENT INCI SPAN MEAN 5.00 2.7 10.00 1.3 15.00 1.3 30.00 0.6 50.00 0.5 70.00 0.5 85.00 2.0 90.00 3.1 95.00 4.3	DENCE DEV 11.9 12.3 12.7 14.0 14.1 13.3 11.8 11.7	0.467 0.857 0.435 0.893 0.428 0.909 0.464 0.944 0.504 0.953 0.516 0.967 0.541 0.890 0.568 0.810	LOSS COEFF TOT PROF 0.087 0.087 0.064 0.064 0.057 0.057 0.037 0.037 0.033 0.033 0.024 0.024 0.078 0.078 0.134 0.134 0.123 0.123	LOSS PARAM TOT PROF 0.038 0.038 0.029 0.029 0.026 0.026 0.017 0.017 0.016 0.016 0.011 0.011 0.035 0.035 0.058 0.058 0.052 0.052

(h) 90 Percent design speed; reading 1596

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 24.729 24.714 24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 15.799 13.541 13.741 12.906 13.056 12.289 12.370	ABS 8 IN 0.0 0.0 0.0 -0.0 0.0 -0.0 0.0 -0.0	BETAM OUT 25.5 24.2 24.9 27.4 30.6 32.7 35.2 37.1 37.5	REL IN 49.4 47.8 47.5 44.7 40.8 36.5 32.8 31.8 31.1	BETAM 0UT 37.4 34.3 32.4 26.3 17.7 8.6 0.8 -1.0 -3.4	IN	TEMP RATIO 1.046 1.045 1.046 1.045 1.041 1.037 1.034 1.032	TOTAL IN 10.00 10.12 10.14 10.15 10.15 10.14 10.14 10.15	PRESS RAT[0 1.163 1.161 1.154 1.154 1.131 1.112 1.090 1.090
RP 1 234 5 67:89	ABS VEL 1N 0UT 158.4 164.9 162.6 173.3 159.8 175.2 159.5 176.7 158.7 177.0 157.5 177.5 157.6 175.1 155.9 165.9 152.6 165.2	241.8 1 236.4 1 224.5 1 209.6 1 195.9 1 187.4 1 183.4 1	/EL OUT 87.2 91.4 88.1 75.0 59.8 51.1 43.1 32.4 31.3	MER II 158.4 162.6 159.5 159.5 157.5 157.6 155.9 152.6	0UT 148.8 158.1 158.9 156.9 152.3	0.0 0.0 0.0 -0.0	VEL OUT 71.0 73.7 81.3 90.1 95.8 100.9 100.0	WHEEL IN 184.7 179.0 174.3 158.0 137.0 116.4 101.5 96.6 92.1	SPEED OUT 184.6 179.0 174.4 158.9 138.7 118.4 103.0 97.7 92.7
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.475 0.484 0.488 0.510 0.480 0.516 0.480 0.522 0.477 0.524 0.473 0.526 0.474 0.520 0.469 0.492 0.458 0.490	0.726 0 0.710 0 0.675 0 0.630 0 0.589 0 0.563 0	CH NO OUT 0.550 0.564 0.554 0.517 0.473 0.448 0.425 0.392 0.389	MERID MA (N 0.475 0.488 0.480 0.477 0.473 0.474 0.469 0.458	OUT 0.437 0.465 0.468 0.463 0.451 0.443 0.425 0.392		:	MERID VEL R 0.940 0.972 0.984 0.960 0.948 0.908 0.859	
RP 1 2 3 4 5 6 7.	PERCENT INCI SPAN MEAN 5.00 -0.9 10.00 -2.3 15.00 -2.2 30.00 -3.0 50.00 -3.1 70.00 -3.2 85.00 -1.6 90.00 -0.6 95.00 0.9	DENCE	9.7 10.3 11.2 12.7 13.5 13.4 11.9 12.0	D-FACT 0.393 0.370 0.374 0.412 0.454 0.461 0.476 0.515 0.502	EFF 0.948 0.959 0.935 0.933 0.946 0.963 0.891 0.774 0.779	LOSS COTOT 0.027 0.021 0.035 0.038 0.031 0.022 0.065 0.130 0.135	DEFF PROF 0.027 0.021 0.035 0.038 0.031 0.022 0.065 0.130 0.135	LOSS P. TOT 0.012 0.010 0.016 0.018 0.015 0.010 0.029 0.056 0.057	ARAM PROF 0.012 0.010 0.016 0.018 0.015 0.015 0.056 0.057

(i) 90 Percent design speed; reading 1604

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RP 1 2 3 4 5 6 7 8 9	RAD IN 24.729 24.026 23.322 21.173 18.321 15.540 13.541 12.906 12.289	0UT 24.714 24.028 23.343 21.285 18.542 15.799 13.741 13.056	ABS IN 0.0 0.0 0.0 0.0 0.0 0.0 0.0	26.2	IN 43.9 42.4 42.0 39.1 35.1 30.8 27.2	BETAM OUT 34.8 31.4 29.5 23.8 15.0 7.4 3.9 2.6 0.9	288.6 288.0 287.8 287.8 287.8 287.8	RATIO 1.040 1.041 1.043 1.043 1.041 1.038 1.032	9.93 10.12 10.15 10.15 10.15 10.15	1.137 1.136 1.137 1.131 1.120 1.085
RP 1 2334 5 6 7 8 9	ABS 194.6 199.2 196.0 197.7 197.3 198.1 199.5 197.1	216.8 206.9 195.5	1N 269.8 269.7 264.0 254.7	228.6 226.5 216.8 203.4	MERI IN 194.6 199.2 196.0 197.7 197.3 198.1 199.5 197.1	0UT 181.3 195.2 197.1 198.4	0.0 0.0 0.0 0.0	G VEL OUT 60.8 62.9 65.5 74.0 87.6 94.6 91.4 90.9 91.2	IN 187.0 181.9 176.8 160.5 138.7 117.9	181.9 177.0 161.4 140.4 119.9
RP 1 2 3 4 5 6 7 8 9	ABS M. IN 0.590 0.606 0.596 0.602 0.601 0.603 0.603 0.600 0.586	ACH NO 0UT 0.568 0.611 0.620 0.633 0.645 0.652 0.652 0.586 0.558	REL M IN 0.819 0.820 0.775 0.734 0.702 0.684 0.670 0.651	ACH NO OUT 0.655 0.682 0.676 0.648 0.610 0.591 0.559 0.519 0.487	MERID M IN 0.590 0.606 0.602 0.601 0.603 0.608 0.600 0.586	0UT 0.538 0.582 0.588 0.593 0.589 0.586 0.558			MERID VEL R 0.932 0.980 1.004 0.996 0.984 0.931 0.878 0.845	
RP 1 2 3 4 5 6 7 8 9	PERCENT SPAN 5.00 10.00 15.00 30.00 50.00 70.00 85.00 90.00	INCI MEAN -6.4 -7.7 -7.6 -8.6 -8.9 -8.9 -7.2 -6.0		7.2 7.3 8.3 10.2 10.9 12.2 15.0 15.6	D-FACT 0.307 0.281 0.277 0.302 0.340 0.342 0.352 0.392 0.419	0.895 0.909 0.872 0.862 0.865 0.863 0.738	LOSS C TOT 0.039 0.035 0.061 0.062 0.063 0.105 0.165 0.204	PROF 0.039 0.035 0.053 0.061 0.062 0.063 0.105 0.165	LOSS P TOT 0.018 0.017 0.025 0.029 0.030 0.029 0.046 0.071 0.086	PROF 0.018 0.017 0.025 0.029 0.030 0.029 0.046 0.071

(j) 100 Percent design speed, reading 1592

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 24.729 24.714 24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 15.799 13.541 13.741 12.906 13.056 12.289 12.370	ABS BI IN 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	OUT 53.0 46.8 43.9 37.8 40.1 41.5 43.9 44.5 45.2	REL IN 58.9 57.2 56.8 54.0 50.3 45.7 41.9 40.8 39.9	BETAM 0UT 39.6 37.4 34.9 28.0 18.9 8.8 2.1 -1.2 -5.0	TOTA IN 288.6 288.6 288.5 288.1 288.0 287.9 287.9 287.9	TEMP RAT10 1.083 1.076 1.073 1.064 1.056 1.048 1.041 1.040	TOTAL IN 10.04 10.13 10.14 10.14 10.14 10.14 10.14 10.14	PRESS RATIO 1.221 1.206 1.206 1.216 1.200 1.174 1.135 1.131
RP 1 2 3 4 5 6 7 8 9	ABS VEL 1N OUT 123.9 158.4 128.7 159.4 127.1 162.3 127.3 170.6 126.3 169.9 125.6 168.2 125.3 158.5 124.1 158.2 122.0 158.6	239.8 12 237.4 13 231.8 14 216.7 15 197.8 13 180.0 12 168.4 15 164.0 15	EL 00T 23.6 37.3 42.5 52.6 37.4 27.5 14.3	MERII IN 123.9 128.7 127.3 126.3 125.6 125.3 124.1 122.0	95.2 109.1 116.9 134.7 130.0 126.0 114.2 112.7	TAN IN 0.0 0.0 0.0 0.0 0.0 0.0 0.0	G VEL 0UT 126.5 116.2 112.6 104.6 109.5 111.5 110.0 110.9	WHEEL IN 205.4 199.5 193.9 175.3 152.2 128.9 112.5 107.3 101.9	SPEED OUT 205.3 199.5 194.1 176.3 154.0 131.1 114.1 108.5 102.6
RP 1 2 3 4 5 6 7 8	ABS MACH NO IN OUT 0.369 0.456 0.383 0.461 0.379 0.470 0.379 0.498 0.377 0.498 0.374 0.467 0.370 0.466 0.363 0.467	0.714 0 0.707 0 0.691 0 0.646 0 0.590 0 0.537 0 0.502 0 0.489 0	H NO 0UT .356 .397 .413 .445 .375 .336 .332	MERID M. IN 0.369 0.383 0.379 0.377 0.374 0.374 0.370 0.363	ACH NO 0UT 0.274 0.315 0.339 0.393 0.381 0.370 0.336 0.332 0.330			MERID VEL R 0.769 0.848 0.920 1.058 1.029 1.003 0.911 0.909 0.917	
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 8.6 10.00 7.1 15.00 7.1 30.00 6.3 50.00 6.3 70.00 6.1 85.00 7.5 90.00 8.4 95.00 9.6		DEV 11.9 13.3 13.7 14.4 14.8 13.6 13.1	D-FACT 0.779 0.692 0.650 0.551 0.584 0.586 0.612 0.606 0.593	0.706 0.729 0.749 0.902 0.954 0.968 0.901 0.886 0.914	LOSS C TOT 0.267 0.230 0.216 0.083 0.041 0.029 0.086 0.102 0.079	0EFF PROF 0.267 0.230 0.216 0.083 0.041 0.029 0.086 0.102 0.079	LOSS P TOT 0.115 0.101 0.096 0.039 0.019 0.013 0.038 0.044 0.033	ARAM PROF 0.115 0.101 0.096 0.039 0.019 0.013 0.038 0.044 0.033

(k) 100 Percent design speed; reading 1614

RP 17 2 3.4 4 5 6 7 8 9	RADII IN 24.729 24 24.026 24 23.322 23 21.173 21 18.321 18 15.540 15 13.541 13 12.289 12	0UT	0 41.1 0 38.5 0 37.3 0 36.5 C 38.4 0 39.9 0 42.6 0 43.3	1N 56.0 54.4 54.1 51.3 47.5 43.3 39.5 38.3	BETAM OUT 40.4 36.9 34.2 27.0 18.8 8.6 1.6 -1.2 -5.0	288.9 288.7 288.4 286.1 288.0 287.9 287.9 287.9	TEMF RATIO 1.073 1.069 1.068 1.065 1.056 1.049 1.042 1.042	TOTAL IN 10.02 10.13 10.14 10.14 10.14 10.14 10.15 10.11	PRESS RATIO 1.210 1.212 1.216 1.223 1.200 1.177 1.137 1.129 1.133
RP 1 2 3 4 5 6 7 8 9	138.6 19 144.3 19 141.9 1 142.3 1 140.3 1 138.2 1 138.1 10 136.7 10	EL FOUT IN 1898.4 248.8 248.7 1.0 241.7 174.9 2077.4 189.5 174.6 174.7 168.6 174.7 168.	1 156.7 2 163.2 9 164.5 9 160.5 9 144.9 9 135.6 9 121.5 3 118.5	MER II 138.6 144.3 141.9 142.3 140.3 138.2 138.1 136.7 133.5	0 VEL 0UT 119.4 130.5 136.1 143.0 137.1 134.0 121.5 118.5	-0.0 1 -0.0 1 -0.1 1 -0.1 1 -0.0 1	VEL OUT 104.1 103.9 103.6 106.0 108.5 111.9 111.9	WHEEL IN 205.7 201.9 195.9 177.9 153.4 130.1 113.6 108.1 102.9	SPEED OUT 205.6 201.9 196.1 178.9 155.2 132.3 115.3 109.3 103.6
RP 1 2 3 4 5 6 7 8 9	0.414 0 0.432 0 0.424 0 0.426 0 0.420 0 0.413 0 0.413 0 0.409 0	H NO RELL DUT IN .458 0.74 .495 0.74 .498 0.72 .521 0.66 .513 0.62 .514 0.53 .480 0.52 .486 0.50	2 0.474 3 0.479 32 0.470 2 0.425 8 0.399 5 0.358	MERID M (N 0.414 0.432 0.424 0.426 0.420 0.413 0.409 0.399	0.345 0.379 0.379 0.396 0.418 0.402 0.395 0.358 0.349			MERID VEL R 0.862 0.904 0.959 1.005 0.977 0.970 0.879 0.867 0.890	
RP 1 2 3 4 5 6 7 8 9	PERCENT SPAN 5.00 10.00 15.00 30.00 50.00 70.00 85.00 90.00 95.00	INCIDENCE MEAN 5.7 4.4 4.4 3.6 3.7 5.0 5.9 7.4	DEV 12.7 12.8 13.0 13.4 14.7 13.4 12.6 11.8 9.9	D-FACT 0.602 0.574 0.553 0.542 0.566 0.566 0.600 0.599 0.579	EFF 0.761 0.817 0.843 0.915 0.953 0.966 0.884 0.843 0.890	0.184 0 0.133 0 0.118 0 0.067 0 0.038 0 0.028 0 0.093 0	PROF 0.184 0.133 0.118 0.067 0.038 0.028 0.093 0.130 0.095	LOSS PATOT 0.078 0.059 0.053 0.032 0.018 0.013 0.041 0.056 0.040	ARAM PROF 0.078 0.059 0.053 0.032 0.013 0.041 0.056 0.040

FOR ROTOR 55

(1) 100 Percent design speed; reading 1613

	(6)	100 1 01 00	,	0	
RP 1 2 3 4 5 6 7 8 9	24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 15.799 13.541 13.741 12.906 13.056	-0.0 31.9 -0.0 31.6 -0.0 32.5 -0.0 35.8	IN 0UT 53.3 39.7 51.7 36.0 51.4 35.4 48.6 27.7 44.8 18.2 40.5 8.3 36.8 0.8 35.7 -1.8	TOTAL TEMP IN RATIO 288.9 1.065 288.9 1.064 288.7 1.064 287.8 1.061 287.8 1.055 287.9 1.044 287.8 1.041 287.7 1.041	IN RATIO 10.00 1.207 10.12 1.209 10.14 1.214 10.15 1.210 10.15 1.194 10.15 1.175 10.15 1.138 10.14 1.123
RP 1 23 4 5 6 1 8 9		REL VEL 1N 0UT 258.3 180.2 256.2 184.4 249.9 183.5 236.3 173.3 217.6 155.5 200.7 146.5 189.1 132.4 184.5 125.6 179.0 127.1	MERID VEL IN 00T 154.4 138.7 158.8 149.2 156.1 155.1 156.2 153.5 154.4 147.7 152.5 145.0 151.4 132.4 149.8 125.6 146.5 126.5	-0.1 112.9 -0.0 112.9	WHEEL SPEED IN OUT 207.1 206.9 201.1 201.1 195.2 195.3 177.3 178.2 153.4 155.2 130.3 132.5 113.1 114.8 107.7 109.0 102.7 103.4
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.463 0.484 0.477 0.513 0.468 0.526 0.469 0.534 0.464 0.536 0.458 0.540 0.454 0.514 0.449 0.498 0.439 0.505	REL MACH NO IN 0UT 0.774 0.524 0.769 0.538 0.750 0.536 0.710 0.508 0.654 0.458 0.602 0.433 0.567 0.391 0.553 0.375	IN OUT 0.463 0.404 0.477 0.436 0.468 0.448 0.469 0.451 0.464 0.435 0.458 0.428 0.454 0.391		MERID VEL R 0.898 0.940 0.981 0.983 0.957 0.950 0.874 0.838 0.863
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 3.0 10.00 1.7 15.00 0.9 50.00 0.9 70.00 0.9 85.00 2.4 90.00 3.3 95.00 4.8	DENCE DEV 12.0 11.9 12.2 14.1 14.1 13.1 11.2 9.6	D-FACT EFF 3.501 0.853 0.481 0.875 0.471 0.894 0.486 0.925 0.532 0.942 0.533 0.960 0.566 0.660 0.585 0.812 0.562 0.846	LOSS COEFF TOT PROF 0.094 0.094 0.080 0.080 0.071 0.071 0.053 0.053 0.043 0.043 0.030 0.030 0.104 0.104 0.138 0.138 0.119 0.119	LOSS PARAM TOT PROF 0.040 0.040 0.036 0.032 0.032 0.032 0.025 0.025 0.020 0.020 0.014 0.014 0.046 0.046 0.059 0.059

FOR ROTOR 55

(m) 100 Percent design speed; reading 1611

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 24.729 24.714 24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 15.799 13.541 13.056 12.289 12.370	ABS BETAM IN OUT -0.0 26. -0.0 26. -0.0 28. -0.0 31. -0.0 33. -0.0 36. -0.0 37. -0.0 38.	IN OUT 3 49.0 36.8 3 47.7 33.9 5 47.3 31.9 5 44.5 25.9 4 40.6 17.4 36.4 8.0 4 32.8 0.6 7 31.7 -0.8	TOTAL TEMP IN RATIO 289.2 1.060 289.1 1.059 288.6 1.060 287.8 1.059 287.9 1.053 287.8 1.048 287.8 1.044 287.7 1.044	TOTAL PRESS IN RATIO 9.97 1.208 10.13 1.208 10.14 1.209 10.15 1.184 10.14 1.168 10.15 1.133 10.08 1.115
RP 1 234 567 89	ABS VEL 1N OUT 180.4 184.1 183.3 192.8 181.1 195.9 181.4 198.4 179.2 197.3 177.4 198.0 177.5 192.0 175.5 183.1 171.1 182.2	REL VEL IN OUT 275.1 203.3 272.4 208.1 266.8 206.9 254.1 194.3 236.2 176.5 220.2 166.7 211.0 154.5 206.4 144.9 200.0 143.9	177.5 154.5 175.5 144.9	TANG VEL IN OUT -0.1 85.9 -0.0 85.5 -0.0 94.0 -0.0 102.8 -0.1 109.4 -0.1 114.0 -0.0 111.9 -0.0 112.1	WHEEL SPEED IN OUT 207.6 207.5 201.4 201.4 195.9 196.1 177.9 178.9 153.8 155.6 130.5 132.7 114.1 115.8 108.5 109.8 103.6 104.2
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.545 0.540 0.554 0.567 0.547 0.577 0.549 0.586 0.542 0.584 0.536 0.588 0.537 0.570 0.531 0.543	REL MACH NO 1N OUT 0.831 0.596 0.823 0.612 0.806 0.609 0.769 0.574 0.715 0.523 0.666 0.495 0.634 0.430 0.604 0.427			MERID VEL R 0.903 0.943 0.970 0.964 0.940 0.931 0.870 0.825 0.839
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 -1.3 10.00 -2.4 15.00 -2.4 30.00 -3.3 50.00 -3.3 70.00 -3.3 85.00 -1.7 90.00 -0.7 95.00 0.9	DENCE DEV 9.1 9.8 10.7 12.3 13.3 12.8 11.7 12.2 11.8	D-FACT EFF 0.435 0.922 0.409 0.941 0.401 0.933 0.431 0.924 0.472 0.930 0.479 0.942 0.509 0.861 0.534 0.768 0.518 0.798		LOSS PARAM TOT PROF 0.019 0.019 0.015 0.015 0.018 0.018 0.022 0.022 0.021 0.021 0.017 0.017 0.037 0.037 0.059 0.059 0.052 0.052

FOR ROTOR 55

(n) 100 Percent design speed; reading 1603

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 24.729 24.714 24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 15.799 13.541 13.741 12.906 13.056 12.289 12.370	ABS 1 IN 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.	BETAM OUT 21.9 20.7 21.8 24.1 27.3 28.8 30.4 32.2 32.8	REL IN 45.6 44.1 43.7 40.7 36.8 32.4 28.9 28.1 27.5	BETAM 0UT 35.1 32.2 29.9 23.7 14.8 7.1 2.9 1.0 0.6	TOTA (N 289.1 289.0 288.7 287.8 287.8 287.8 287.8	L TEMP RAT10 1.056 1.055 1.055 1.054 1.052 1.046 1.037 1.035	TOTAL IN 9.93 10.13 10.15 10.15 10.16 10.16 10.14 10.09	PRESS RATIO 1.186 1.183 1.179 1.172 1.153 1.107 1.079 1.058
RP 1 2 3 4 5 6 7 8 9	ABS VEL IN OUT 201.6 201.0 206.1 211.7 202.7 214.3 204.2 218.3 203.8 222.2 203.5 222.3 204.1 207.9 201.5 197.9 196.7 187.5	286.9 280.4 269.3 254.4 241.1 233.1 228.3	VEL 0UT 227.9 234.2 229.6 217.8 204.1 196.2 179.5 167.4 157.7	MER II 1N 201.6 206.1 202.7 204.2 203.8 203.5 204.1 201.5 196.7	D VEL OUT 186.5 198.1 199.0 199.3 197.4 194.7 179.3 167.4 157.7	TAN IN 0.0 0.0 0.0 0.0 0.0 0.0 0.0	G VEL OUT 74.8 74.7 79.5 89.0 102.1 107.2 105.3 105.6 101.5	WHEEL IN 206.0 199.6 193.8 175.7 152.3 129.3 1129.7 107.4	SPEED OUT 205.8 199.6 194.0 176.6 154.1 131.5 114.4 108.7 103.0
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.613 0.594 0.628 0.628 0.651 0.625 0.665 0.621 0.667 0.623 0.615 0.599 0.560	0.875 0.854 0.822 0.776 0.736 0.712 0.696	CH NO OUT 0.673 0.695 0.682 0.649 0.611 0.589 0.538 0.501	MERID M IN 0.613 0.628 0.617 0.623 0.622 0.621 0.623 0.615 0.599	ACH NO OUT 0.551 0.588 0.591 0.594 0.590 0.584 0.537 0.501 0.471			MERID VEL R 0.925 0.961 0.982 0.976 0.969 0.957 0.831 0.802	
RP 1 2 3 4 5 6 7 8 9	PERCENT INC SPAN MEAN 5.00 -4.7 10.00 -6.0 15.00 -7.0 50.00 -7.2 70.00 -7.2 85.00 -5.5 90.00 -4.3 95.00 -2.8	DENCE	7.4 8.2 8.7 10.2 10.6 11.9 13.9 14.1	D-FACT 0.354 0.328 0.335 0.366 0.400 0.397 0.432 0.468 0.482	0.894 0.896 0.888 0.888 0.894 0.907 0.785 0.634 0.538	LOSS C TOT 0.050 0.048 0.054 0.056 0.055 0.047 0.094 0.153 0.178	OEFF PROF 0.050 0.048 0.054 0.056 0.055 0.047 0.094 0.153 0.178	LOSS P TOT 0.023 0.022 0.025 0.027 0.027 0.022 0.041 0.066 0.075	ARAM PROF 0.023 0.025 0.025 0.027 0.027 0.027 0.041 0.066 0.075

FOR ROTOR 55

(o) 110 Percent design speed; reading 1694

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 24.729 24.714 24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 15.799 13.541 13.741 12.906 13.056 12.289 12.370	ABS BETAM IN OUT 0.0 44. 0.0 39. 0.0 36. 0.0 36. 0.0 40. 0.0 40. 0.0 44. 0.0 44.	IN OUT 2 55.6 40.5 3 55.0 38.2 8 54.3 35.4 5 1.6 28.3 3 47.7 18.3 4 39.0 0.8 1 38.1 -1.6	TOTAL TEMP IN RATIO 288.9 1.092 288.7 1.087 288.2 1.085 288.0 1.076 288.0 1.069 287.9 1.060 288.0 1.051 287.9 1.050 287.8 1.049	TOTAL PRESS IN RATIO 10.08 i.260 10.13 1.265 10.14 1.271 10.14 1.220 10.14 1.167 10.14 1.156 10.12 i.161
RP 1 2 3 4 5 6 7 8 9	ABS VEL IN OUT 157.0 174.9 157.2 180.4 154.5 184.5 155.0 191.0 153.6 191.8 153.0 192.4 153.7 181.4 151.9 178.6 148.5 180.7	REL VEL IN OUT 277.8 164.9 273.7 177.6 264.9 181.1 249.5 174.2 228.3 156.5 209.3 148.2 197.8 131.8 193.0 128.3 187.1 130.2	MERID VEL IN OUT 157.0 125.3 157.2 139.6 154.5 147.4 153.6 148.5 153.0 146.7 153.7 131.8 151.9 128.3 148.5 129.6	0.0 121.5 0.0 124.4 0.0 124.6 0.0 124.2	WHEEL SPEED (N OUT 229.3 229.2 224.1 224.2 215.2 215.4 195.5 196.6 169.0 171.0 142.9 145.3 124.6 126.4 119.2 120.6 113.8 114.5
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.471 0.503 0.472 0.522 0.464 0.535 0.465 0.558 0.461 0.563 0.459 0.567 0.461 0.535 0.456 0.526 0.445 0.533	REL MACH NO IN OUT 0.833 0.475 0.821 0.513 0.795 0.525 0.749 0.509 0.685 0.459 0.628 0.437 0.594 0.389 0.579 0.378 0.561 0.384	MERID MACH NO 1N OUT 0.471 0.361 0.472 0.403 0.464 0.428 0.465 0.435 0.459 0.432 0.461 0.389 0.456 0.378 0.445 0.383		MERID VEL R 0.798 0.888 0.956 0.990 0.967 0.959 0.858 0.845
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 5.3 10.00 4.9 15 t0 4.7 30.00 3.9 50.00 3.8 70.00 3.4 85.00 4.6 90.00 5.7 95.00 7.2	DENCE DEV 12.9 14.2 14.8 14.3 12.9 11.8 11.4	D-FACT EFF 0.651	LOSS COEFF TOT PROF 0.201 0.201 0.161 0.161 0.144 0.144 0.058 0.058 0.038 0.038 0.016 0.016 0.091 0.091 0.122 0.122 0.097 0.097	LOSS PARAM TOT PROF 0.085 0.085 0.070 0.070 0.064 0.064 0.027 0.027 0.018 0.018 0.008 0.008 0.040 0.040 0.053 0.053 0.041 0.041

(p) 110 Percent design speed; reading 1693

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 24.729 24.714 24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 15.791 13.541 13.741 12.906 13.056 12.289 12.370	ABS 8 IN 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.	BETAM OUT 33.1 30.3 30.0 32.2 35.4 36.7 39.5 40.8 40.6	REL IN 51.4 50.5 50.1 47.2 43.3 38.7 34.8 33.9 33.2	BETAM OUT 39.3 36.1 33.8 26.6 17.8 7.7 0.1 -1.6 -4.0	TOTA IN 289.0 288.8 288.5 287.9 287.9 287.9 287.9 287.8	L TEMP RATIO 1.080 1.077 1.074 1.066 1.059 1.052 1.049	IN 10.06 10.14 10.14 10.14 10.14 10.14 10.14	PRESS RATIO 1.244 1.260 1.266 1.238 1.217 1.170 1.145 1.147
RP 1 2 3 4 5 6 7 8 9	ABS VEL IN OUT 183.0 186.1 183.3 196.3 180.6 200.5 181.3 205.8 179.7 203.8 179.7 203.8 179.7 180.0 199.2 177.8 191.2 173.8 191.8	288.4 281.6 266.8 247.0 230.0 219.2 214.2	VEL 0UT 201.2 209.9 208.9 194.9 174.4 167.7 153.7 144.9 146.1	IN 183.0 183.3 180.6 181.3 179.7 179.4 180.0	VEL 0UT 155.8 169.6 173.6 174.2 166.1 166.2 153.7 144.9	IN 0.0 0.1 0.1 0.1 0.1 0.0	G VEL OUT 101.7 99.0 100.2 109.6 118.2 123.9 126.8 124.8 124.7	HEELL IN 229.2 222.6 216.1 195.9 169.5 143.9 125.2 119.4 113.7	SPEED OUT 229.1 222.7 216.3 196.9 171.5 146.3 127.1 120.8 114.4
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.553 0.540 0.554 0.573 0.546 0.605 0.544 0.601 0.543 0.614 0.545 0.566 0.525 0.569	0.872 0.851 0.808 0.747 0.696 0.663 0.648	CH NO OUT 0.585 0.613 0.611 0.573 0.514 0.497 0.496 0.429	MERID M IN 0.553 0.554 0.546 0.549 0.544 0.543 0.545 0.538	0.453 0.495 0.508			MERID VEL R 0.851 0.925 0.962 0.961 0.924 0.927 0.854 0.815 0.839	• •
RP 1 2 3 4 5 6 7 8	PERCENT INC SPAN MEAN 5.00 1.1 10.00 0.5 15.00 0.4 30.00 -0.5 50.00 -0.7 70.00 -0.9 95.00 1.5 95.00 2.9	•	DEV 11.6 12.1 12.6 13.1 13.7 12.5 11.1 11.4	D-FACT 0.507 0.462 0.452 0.487 0.535 0.526 0.557 0.577	0.804 0.890 0.918 0.939 0.954 0.973	LOSS COTOT 0.125 0.069 0.053 0.042 0.032 0.019 0.074 0.130 0.118	DEFF PROF 0.125 0.069 0.053 0.042 0.032 0.019 0.074 0.130 0.118	0.054 0.031 0.024 0.020 0.015	ARAM PROF 0.054 0.031 0.024 0.020 0.015 0.009 0.033 0.056 0.049

FOR ROTOR 55

(q) 110 Percent design speed; reading 1692

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 24.729 24.714 24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 15.799 13.541 13.741 12.906 13.056 12.289 12.370	ABS BETAM IN OUT 0.0 27.1 0.0 26.5 0.0 28.3 0.0 30.8 0.0 34.5 0.0 35.6	IN OUT 47.8 36.3 34.4 46.9 31.1 43.4 24.8 39.4 15.9 8.0 2 31.2 27 5 30.3 1.1	TOTAL TEMP IN RATIO 289.1 1.075 288.9 1.072 288.4 1.072 288.0 1.071 287.8 1.064 287.9 1.056 288.0 1.045 287.9 1.039	TOTAL PRESS IN RATIO 10.04 1.227 10.13 1.239 10.14 1.243 10.15 1.240 10.15 1.195 10.15 1.195 10.14 1.107 10.10 1.088
RP 1 2 3 4 5 6 7 8 9	ABS VEL IN OUT 208.1 206.6 208.7 215.6 205.6 218.5 207.2 223.9 206.1 226.6 205.8 227.9 207.2 211.7 204.6 202.9 199.7 195.3	REL VEL 1N 0UT 309.5 228.2 305.2 231.5 297.9 228.5 285.4 217.2 266.8 202.4 251.0 196.3 242.2 175.2 236.9 165.3 229.9 159.5	MERID VEL 1N OUT 208.1 184.0 208.7 193.4 205.6 195.5 207.2 197.1 206.1 194.6 205.8 194.4 207.2 175.1 204.6 165.2 199.7 159.5	TANG VEL IN OUT 0.1 94.0 0.1 95.5 0.1 97.6 0.0 106.2 0.1 116.0 0.1 119.1 0.1 117.7 0.1 112.7	WHEEL SPEED IN OUT 229.2 229.0 222.8 222.8 215.6 215.8 196.3 197.4 169.5 171.5 143.7 146.1 125.5 127.3 119.5 120.9 114.1 114.8
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO 1N 0UT 0.635 0.606 0.637 0.636 0.627 0.646 0.633 0.663 0.675 0.629 0.682 0.633 0.625 0.605 0.608 0.582	REL MACH NO IN OUT 0.944 0.669 0.931 0.675 0.872 0.644 0.815 0.603 0.767 0.524 0.723 0.475	MERID MACH NO IN OUT 0.635 0.539 0.637 0.570 0.627 0.578 0.633 0.584 0.630 0.579 0.629 0.581 0.635 0.475		MER (0 VEL R 0.884 0.927 0.951 0.951 0.944 0.945 0.845 0.808 0.799
RP 1 2 3 4 5 6 7 8 9	PERCENT INC SPAN MEAN 5.00 -2.5 10.00 -3.2 15.00 -3.3 30.00 -4.5 70.00 -4.7 85.00 -3.2 90.00 -2.1 95.00 -0.5	8.6 9.3 9.9 11.3 11.8 12.7 13.7 14.1	0.436 0.899 0.460 0.936 0.443 0.945 0.496 0.804 0.518 0.685	LOSS COEFF TOT PROF 0.109 0.109 0.066 0.066 0.059 0.059 0.058 0.038 0.032 0.032 0.098 0.098 0.152 0.152 0.170 0.170	LOSS PARAM TOT PROF 0.049 0.049 0.030 0.030 0.027 0.027 0.028 0.018 0.018 0.018 0.015 0.015 0.043 0.043 0.066 0.066 0.072 0.072

FOR ROTOR 55

(r) 120 Percent design speed; reading 1695

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 24.729 24.714 24.026 24.028 23.322 23.343 21.173 21.285 18.321 18.542 15.540 15.799 13.541 13.741 12.906 13.056 12.289 12.370	ABS BI IN 0.0 0.0 0.0 0.0 0.0 0.0 0.0	TTAM R OUT IN 45.6 54.3 39.3 53.3 57.8 53.3 56.7 50.3 39.5 46.4 40.3 42.4 43.8 38.3 44.1 37.4 44.1 36.4	9 40.0 7 37.7 4 35.2 9 28.3 9 17.8 7.7 5 0.5 0 -1.8	TOTAL IN 289.0 288.8 288.5 298.0 287.9 287.9 287.9 287.9	TEMP RATIO 1.110 1.101 1.099 1.091 1.083 1.072 1.061 1.058	TOTAL IN 10.05 10.14 10.14 10.14 10.14 10.14 10.14	PRESS RATIO 1.314 1.310 1.317 1.327 1.304 1.267 1.200 1.182
RP 1 2 3 4 5 6 7 8 9	ABS VEL IN OUT 174.0 190.4 177.2 196.0 174.2 200.9 173.1 208.3 172.7 211.2 173.6 212.3 173.2 198.6 171.4 194.1 167.5 195.9	302.7 17 299.5 19 292.4 19 274.5 18 252.6 17 233.6 16 220.6 14 214.6 13	OUT IN 73.8 174.0	0 133.3 2 151.7 2 158.8 1 167.1 7 162.9 6 162.0 2 143.4 4 139.4	IN 0.0 0.0	VEL 0UT 135.9 124.1 123.0 124.4 134.4 137.5 137.5 135.1	WHEEL IN 247.7 241.4 234.9 213.0 184.3 156.4 136.7 129.2	SPEED 0UT 247.6 241.5 235.1 214.1 186.5 159.0 138.7 130.7 124.2
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.525 0.546 0.535 0.566 0.523 0.607 0.521 0.619 0.524 0.626 0.523 0.586 0.517 0.573 0.505 0.579	0.913 0. 0.904 0. 0.882 0. 0.829 0. 0.763 0. 0.706 0. 0.666 0.	I NO MERID 2498 0.525 553 0.535 563 0.525 553 0.525 551 0.525 482 0.524 482 0.524 411 0.515 417 0.505	0.438 0.460 0.487 0.477 4 0.478 5 0.423 7 0.411			MERID VEL R 0.766 0.956 0.912 0.965 0.943 0.933 0.828 0.813 0.839	
RP 1 2 3 4 5 6 7 8 9	PERCENT INC SPAN MEAN 5.00 4.6 10.00 3.7 15.00 3.8 30.00 3.2 50.00 2.9 70.00 2.4 85.00 3.9 90.00 4.6 95.00 6.1	1 1 1 1 1	DEV D-FAG 2.3 0.676 3.7 0.586 4.0 0.564 3.6 0.596 2.4 0.576 1.5 0.626 1.2 0.624 0.0 0.596	0.739 0.791 4 0.827 0.928 1 0.945 0.972 8 0.876 1 0.839	0.023 0.097 0.125	PEFF PROF 0.210 0.160 0.135 0.058 0.046 0.023 0.097 0.125 0.085	LOSS PATOT 0.090 0.070 0.060 0.027 0.022 0.011 0.043 0.054 0.036	ARAM PROF 0.090 0.070 0.060 0.027 0.022 0.011 0.043 0.054 0.036

(s) 120 Percent design speed; reading 1696

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RP 1 2 5 4 5 6 7 8 9	RADII IN OUT 24.729 24.71. 24.026 24.02! 23.322 23.34! 21.173 21.28! 18.321 18.54! 15.540 15.79! 13.541 13.74 12.906 13.050 12.289 12.37	3 0.0 5 0.0 5 0.0 2 0.0	BETAM OUT 41.9 34.4 33.0 34.1 36.3 38.0 41.2 41.6 41.2	52.0 51.4 50.9 47.8 43.9 39.5	39.2 35.5 32.6 26.0 17.5 7.5 0.1	289.0 288.9 288.5 288.0 287.9 287.9 287.8 287.8	1.106 1.099 1.097 1.089 1.080 1.070	TOTAL IN 10.07 10.14 10.14 10.14 10.14 10.14 10.14	PRESS RATIO 1.280 1.311 1.327 1.321 1.292 1.259 1.196 1.174 1.173
RP 1 2 3 4 5 6 7 8 9	ABS VEL IN OUT 193.8 194. 192.6 209. 190.6 216. 192.9 221. 191.5 220. 189.1 220. 189.1 220. 189.2 201. 187.5 202.	IN 315.2 308.4 7 301.8 7 287.2	187.2 211.9 215.8 204.4	190.6 192.9	145.1 172.6 181.8 183.6	1N 0.0 0.0 0.0 0.0 0.0 0.0 0.0	118.1 118.1 124.2 130.4 135.7 137.8 134.4 132.8	WHEEL IN 248.5 241.0 234.1 212.8 184.1 155.9 136.0 129.7 123.0	SPEED OUT 248.4 241.0 234.3 213.9 186.4 158.5 138.0 131.2 123.8
RP \ 23456789	ABS MACH NI OUT 0.588 0.56 0.584 0.60 0.578 0.65 0.65 0.582 0.65 0.574 0.65 0.569 0.569 0.554 0.554	IN 0.956 7 0.936 8 0.916 9 0.873 9 0.807 2 0.744 9 0.710 0 0.692	0.466	MERID M IN 0.588 0.584 0.578 0.586 0.582 0.574 0.569 0.554	OUT 0.417 0.501 0.530 0.539 0.524 0.514 0.466 0.449			MER (D VEL R 0.749 0.897 0.954 0.952 0.928 0.918 0.828 0.829	
RP 1 2 3 4 5 6 7 8 9	SPAN ME 5.00 1 10.00 1 15.00 1 30.00 0 50.00 -0 70.00 -0	8 3 2 1 1	11.5 11.4 11.4 12.5 13.3 12.3	D-FACT 0.636 0.524 0.498 0.517 0.546 0.548 0.590 0.591 0.566	0.690 0.810 0.863 0.928 0.948 0.970 0.857	LOSS C TOT 0.226 0.137 0.100 0.053 0.039 0.023 0.100 0.133 0.108	PR0F 0.226	LOSS P. TOT 0.098 0.0961 0.046 0.025 0.019 0.011 0.044 0.057	ARAM PROF 0.098 0.061 0.046 0.025 0.019 0.011 0.044 0.057

TABLE VII. - Concluded. BLADE-ELEMENT DATA AT BLADE EDGES

FOR ROTOR 55

(t) 120 Percent design speed; reading 1697

		(-, -				,				
RP 1 2 3 4 5 6 7 8 9	RAD IN 25.230 24.547 23.876 21.847 19.164 16.502 14.519 13.858 13.200	0UT 25.298 24.671 24.049 22.222 19.827 17.465 15.682 15.070	ABS IN 33.8 29.3 28.4 30.6 32.8 32.8 36.2 37.1 36.1	BETAM OUT -3.3 -2.6 -2.5 -1.9 -0.8 0.6 -3.1 -4.3 -8.0	IN 33.8 29.3	BETAM OUT -3.3 -2.5 -1.9 -0.8 0.6 -3.1 -4.3 -8.0	TOTA IN 316.4 315.1 314.3 312.5 310.6 307.4 304.1 303.3 301.7	RATIO 0.993 0.995 0.995 0.997 0.996 0.998 1.001 1.003	TOTAL IN 12.44 12.88 13.02 13.03 12.95 12.63 11.83 11.63	PRESS RATIO 0.985 0.966 0.968 0.993 0.984 1.001 1.010
RF 1 2 3 4 5 6 7 8 9	ABS IN 200.2 217.1 223.5 228.6 233.6 232.8 211.7 205.6 196.3	VEL 0UT 181.5 188.8 194.8 206.9 210.6 207.6 197.8 195.7 188.6	REL (N 200.2 217.1 223.5 228.6 233.6 232.8 211.7 205.6 196.3	VEL 0UT 181.5 188.8 194.8 206.9 210.6 207.6 197.8 195.7 188.6	IN 166.4 189.2 196.6 196.8 196.4 195.6	D VEL 0UT 181.2 188.6 194.6 206.8 210.6 207.6 197.5 195.1	TAN IN 111.3 106.4 106.3 116.2 126.6 126.2 125.0 124.0 115.7	-14.8	WHEEL (N 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.	SPEED OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9	ABS M IN 0.580 0.634 0.655 0.674 0.692 0.694 0.629 0.611 0.583	ACH NO OUT 0.525 0.548 0.567 0.666 0.620 0.613 0.558 0.556	REL M IN 0.580 0.634 0.655 0.674 0.692 0.694 0.629 0.611	ACH NO OUT 0.525 0.548 0.567 0.606 0.620 0.613 0.585 0.578	MERID M IN 0.482 0.553 0.576 0.580 0.582 0.583 0.508 0.487 0.471	ACH NO OUT 0.524 0.547 0.566 0.606 0.620 0.613 0.584 0.576			MERID VEL R 1.089 0.997 0.990 1.051 1.073 1.061 1.156 1.190 1.178	
RP 1 2 3 4 5 6 7 8 9	PERCENT SPAN 5.00 10.00 15.00 50.00 70.00 85.00 90.00 95.00	INC I MEAN -6.7 -11.2 -10.4 -8.6 -9.0 -6.0 -5.2 -6.3	DENCE	DEV 12.3 12.7 12.5 12.1 12.2 12.3 7.8 6.3 2.4	D-FACT 0.508 0.481 0.459 0.412 0.385 0.346 0.318 0.303 0.302	EFF 0. 0. 0. 0. 0. 0.	LOSS C TOT 0.076 0.145 0.126 0.032 0.026 0.058 -0.005 -0.044 -0.093	PROF 0.076 0.145 0.126 0.032 0.026 0.058 -0.005 -0.044	LOSS PA TOT 0.052 0.096 0.082 0.019 0.014 0.027 -0.002 -	PROF 0.052 0.096 0.082 0.019 0.014 0.027 0.002

(a) 80 Percent design speed; reading 1597

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 25.230 25.298 24.547 24.671 23.876 24.049 21.847 22.222 19.164 19.827 16.502 17.465 14.519 15.682 13.858 15.070 13.200 14.448	ABS BE IN 52.9 47.6 44.8 37.8 39.4 40.6 42.5 43.2 44.0	TAM REL OUT IN 4.4 52.9 4.5 47.6 2.9 44.8 1.1 37.8 1.1 39.4 1.0 40.6 -2.0 42.5 -5.7 43.2 -8.5 44.0	BETAM OUT 4,4 4.5 2.9 1.1 1.1 1.0 -2.0 -5.7 -8.5	303.5 0. 302.0 0. 301.4 0. 299.6 0. 298.0 0. 296.5 1. 295.1 1. 295.0 1.	EMP TOTAL T10 IN 995 11.44 997 11.42 997 11.49 999 11.37 001 11.20 002 10.96 002 10.93 002 10.93	0.988 0.992 0.993 1.001
RP 1 2 3 4 5 6 7 8 9	ABS VEL IN OUT 122.8 107.6 123.5 109.5 124.6 111.2 132.2 116.0 131.6 114.7 128.0 108.1 118.7 99.8 117.8 97.3 118.5 94.9	122.8 10 123.5 10 124.6 11 132.2 11 131.6 11 128.0 10 118.7 9	MERII 17.6 74.0 19.5 83.3 1.2 88.4 6.0 104.5 4.7 101.6 18.1 97.1 19.8 87.5 17.3 85.9 14.9 85.3	0 VEL 0UT 107.3 109.1 111.0 116.0 114.7 108.1 99.8 96.9 93.9	98.0 91.2 87.8 80.9 83.6 83.3 80.2	EL WHEEL UT IN 8.3 0. 8.5 0. 5.7 0. 2.2 0. 2.1 0. 2.0 0. 3.6 0. 9.6 0. 4.1	SPEED OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.356 0.312 0.359 0.318 0.363 0.323 0.387 0.338 0.386 0.335 0.376 0.316 0.349 0.292 0.346 0.285 0.348 0.278	0.356 0. 0.359 0. 0.363 0. 0.387 0. 0.386 0. 0.376 0. 0.349 0. 0.346 0.	H NO MERID M 312 0:215 318 0.242 323 0.257 338 0.306 335 0.298 316 0.285 292 0.257 285 0.252 278 0.251	ACH NO OUT 0.311 0.317 0.323 0.338 0.335 0.316 0.292 0.283 0.275		MERID VEL R 1.449 1.311 1.256 1.109 1.129 1.113 1.140 1.128	
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 12.5 10.00 7.1 15.00 4.2 30.00 -3.2 50.00 -2.0 70.00 -1.2 85.00 0.3 90.00 0.9 95.00 1.6	2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	DEV D-FACT 0.1 0.621 9.8 0.557 7.9 0.532 5.1 0.474 4.1 0.448 2.8 0.438 8.9 0.436 5.0 0.462 1.9 0.492	0. 0. 0. 0.	0.184 0. 0.142 0. 0.105 0. 0.118 0. 0.078 0. 0.071 0. -0.017 -0.	OF TOT 184 0.125 142 0.094 105 0.068 118 0.070 078 0.041 071 0.032 017 -0.007	PROF 0.125 0.094 0.068 0.070 0.041 0.032 -0.007

(b) 80 Percent design speed; reading 1598

RP 1 2 3 4 5 6 7 8 9	RAD! IN 25.230 2 24.547 2 23.876 2 21.847 2 19.164 1 16.502 1 14.519 1 13.858 1	0UT	ABS 1N 38.6 35.0 34.3 36.8 38.3 41.1 42.0 42.6	BETAM OUT 0.2 0.0 -0.7 -1.7 -1.4 0.6 -2.1 -3.7 -7.3	REL IN 38.6 35.0 34.3 36.8 38.3 41.1 42.0 42.6	BETAM OUT 0.2 0.0 -0.7 -1.7 -1.4 0.6 -2.1 -3.7 -7.3	TOTA IN 301.1 300.3 300.1 298.9 297.8 296.6 295.4 295.1	TEMP RATIO 1.000 1.000 0.999 1.000 0.999 1.001 1.001	TOTAL IN 11.34 11.44 11.44 11.34 11.21 10.98 10.91	PRESS RATIO 0.991 0.991 0.995 0.995 0.995 0.998 1.001 0.997
RP 1 2 3 4 5 6 7 8 9	133.1 136.4 136.1	VEL 0UT 113.3 117.0 119.9 124.3 121.6 116.0 106.1 103.9 102.1	REL IN 124.2 129.7 133.1 136.4 136.1 134.2 125.8 122.4 123.9	VEL 0UT 113.3 117.0 119.9 124.3 121.6 116.0 106.1 103.9 102.1	MERI IN 97.1 106.2 110.3 112.7 109.1 105.3 94.8 91.0 91.3	D VEL OUT 113.3 117.0 119.9 124.3 121.6 116.0 106.0 103.7	TAN IN 77.4 74.4 74.5 76.9 81.5 83.2 82.7 81.9	G VEL OUT 0.3 0.0 -1.6 -3.6 -3.1 1.3 -3.9 -6.8	WHEEL IN 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.	SPEED OUT 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9	0.389 0.400 0.400 0.395 0.370 0.360	CH NO OUT 0.329 0.341 0.350 0.363 0.356 0.340 0.311 0.304 0.299	REL M. 1N 0.362 0.379 0.400 0.400 0.395 0.370 0.360 0.365	ACH NO OUT 0.329 0.341 0.350 0.363 0.356 0.340 0.311 0.304 0.299	MERID M IN 0.283 0.310 0.322 0.330 0.320 0.310 0.279 0.268 0.269	OUT 0.329 0.341 0.350 0.363 0.356 0.340 0.310 0.304 0.297	,		MERID VEL R 1.167 1.102 1.087 1.102 1.115 1.102 1.118 1.139 1.109	
RP 1 2 3 4 5 6 7 8 9	PERCENT SPAN 5.00 10.00 15.00 30.00 50.00 70.00 85.00 90.00 95.00	INCII MEAN -1.9 -5.5 -6.6 -6.7 -4.7 -3.5 -1.1 -0.3	DENCE	DEV 15.8 15.3 14.2 12.4 11.6 12.3 8.8 6.9 3.1	D-FACT 0.511 0.478 0.467 0.437 0.428 0.408 0.427 0.424 0.458	EFF 0. 0. 0. 0. 0. 0.	LOSS C TOT 0.101 0.100 0.084 0.045 0.045 0.069 0.025 -0.011	PROF 0.101 0.100 0.084 0.045 0.045 0.069 0.025	LOSS PA TOT 0.069 0.067 0.054 0.026 0.024 0.032 0.010 70.004	PROF 0.069 0.067 0.054 0.026 0.024 0.032 0.010

(c) 80 Percent design speed; reading 1600

RP 1 2 3 4 5 6 7 8 9	RADII IN 0UT 25.230 25.298 24.547 24.671 23.876 24.049 21.847 22.222 19.164 19.827 16.502 17.465 14.519 15.682 13.858 15.070 13.200 14.448	ABS Bi IN 24.4 23.4 25.9 26.2 26.0 28.9 30.9 33.5 35.3	TTAM RE OUT IN -4.6 24.4 -3.7 23.4 25.9 -4.4 26.0 -4.1 28.9 -2.5 30.9 -3.4 35.5 3	-3.7 -3.4 -4.4 -4.4 -4.1 -2.5 -3.4	TOTAL TEMP IN RATIO 298.9 0.999 298.7 0.999 298.5 0.999 297.7 1.000 297.6 1.000 297.0 1.000 296.1 1.001 295.5 1.001 294.9 1.001	TOTAL PRESS IN RATIO 11.27 0.977 11.35 0.984 11.33 0.995 11.32 0.996 11.23 1.001 11.14 0.998 11.02 0.995 10.82 1.001
RP 1 23 4 5 6 7 8 9	ABS VEL 1N OUT 143.0 134.2 151.3 141.9 152.7 144.6 155.8 148.9 156.0 149.4 154.4 151.3 153.0 150.1 148.6 144.0 140.5 139.6	143.0 13 151.3 14 152.7 14 155.8 14 156.0 14 154.4 15 153.0 15 148.6 14	CL MER DUT IN 54.2 130.3 11.9 138.9 14.6 139.6 18.9 139.8 19.4 140.2 51.3 135.2 50.1 131.3 50.1 124.0	141.6 144.3 148.4 149.0 150.9 150.0 143.7	TANG VEL IN OUT 59.0 -10.7 60.0 -9.1 61.9 -8.7 68.8 -11.4 68.4 -11.4 74.6 -10.8 78.6 -6.5 81.9 -8.5 81.2 -10.0	HHEEL SPEED IN OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.420 0.393 0.445 0.417 0.450 0.425 0.460 0.439 0.460 0.440 0.456 0.446 0.453 0.444 0.439 0.425 0.415 0.412	0.420 0. 0.445 0. 0.450 0. 0.460 0. 0.460 0. 0.456 0. 0.453 0. 0.439 0.	OUT IN .393 0.383 .417 0.409 .425 0.411 .439 0.413	0.416 0.424 0.437 0.439 0.445 0.443		MERID VEL R 1.027 1.020 1.034 1.062 1.063 1.116 1.142 1.159
RP 1 2 3 4 5 6 7 8 9	SPAN MEAN 5.00 -16.1 10.00 -17.2 15.00 -16.7 30.00 -14.8	1	DEV D-FAC 1.1 0.394 11.6 0.365 11.5 0.352 9.7 0.348 8.7 0.269 8.5 0.238 7.3 0.260 6.3 0.239	0. 0. 0. 0. 0. 0.	LOSS COEFF TOT PROF 0.199 0.199 0.128 0.128 0.078 0.078 0.038 0.038 0.029 0.029 -0.005 -0.005 0.018 0.018 0.043 0.043 -0.013 -0.013	LOSS PARAM TOT PROF 0.135 0.135 0.085 0.085 0.050 0.050 0.022 0.022 0.015 0.015 -0.002 -0.002 0.007 0.007 0.017 0.017 -0.005 -0.005

(d) 80 Percent design speed; reading 1606

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RP 1 2 3 4 5 6 7 8 9	RADII 1N 0UT 25.230 25.298 24.547 24.671 23.876 24.049 21.847 22.222 19.164 19.827 16.502 17.465 14.519 15.682 13.858 15.070 13.200 14.448	ABS BETAM IN OUT 16.2 -6.9 15.5 -5.6 15.9 -5.5 17.9 -5.4 22.0 -4.1 24.2 -3.6 24.7 -4.1 25.7 -5.9 26.9 -6.7	15.5 -5.6 15.9 -5.5 17.9 -5.4 22.0 -4.1 24.2 -3.6 24.7 -4.1 25.7 -5.9	297.4 0.998 297.6 0.998	TOTAL PRESS IN RATIO 10.90 0.958 11.09 0.967 11.12 0.972 11.19 0.977 11.20 0.985 11.15 0.992 10.90 0.990 10.61 0.998
RP 1 2 3 4 5 6 7 8 9	ABS VEL IN OUT 173.3 172.5 184.9 184.7 188.6 188.5 194.8 195.8 196.3 204.5 198.5 212.3 189.8 210.9 177.8 201.4 169.0 194.0	REL VEL IN OUT 173.3 172.5 184.9 184.7 188.6 188.5 194.8 195.8 198.3 204.5 198.5 212.3 189.8 210.9 177.8 201.4 169.0 194.0	MERID VEL IN OUT 166.4 171.2 178.2 183.8 181.3 187.6 185.3 194.9 183.9 203.9 181.1 211.9 172.4 210.4 160.2 200.3 150.8 192.7	TANG VEL IN OUT 48.2 -20.6 49.6 -18.1 51.7 -18.1 60.0 -18.5 74.3 -14.7 81.4 -13.2 79.5 -15.2 77.3 -20.8 76.4 -22,5	WHEEL SPEED IN OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.514 0.512 0.551 0.550 0.562 0.563 0.583 0.586 0.594 0.614 0.595 0.640 0.569 0.636 0.531 0.606 0.504 0.583	REL MACH NO IN OUT 0.514 0.512 0.551 0.550 0.562 0.563 0.583 0.586 0.594 0.614 0.595 0.640 0.596 0.636 0.531 0.606 0.504 0.583	MERID MACH NO		MERID VEL R 1.029 1.031 1.034 1.052 1.109 1.170 1.220 1.251 1.278
RP 1 2 3 4 5 6 7 8 9	PERCENT INC SPAN MEAN 5.00 -24.3 10.00 -25.0 15.00 -24.7 30.00 -23.1 50.00 -19.4 70.00 -17.4 90.00 -16.5 95.00 -15.5	8.8 9.7 9.5 8.6 8.9 8.2 6.8	D-FACT EFF 0.276 0. 0.244 0. 0.240 0. 0.233 0. 0.202 0. 0.145 0. 0.087 0. 0.077 0. 0.065 0.	LOSS COEFF TOT PROF 0.257 0.257 0.178 0.178 0.143 0.143 0.111 0.111 0.069 0.069 0.039 0.039 0.050 0.050 0.055 0.055 0.073 0.073	LOSS PARAM TOT PROF 0.174 0.174 0.118 0.118 0.092 0.092 0.066 0.066 0.036 0.036 0.018 0.018 0.020 0.020 0.021 0.021 0.027 0.027

FOR STATOR 55

(e) 90 Percent design speed; reading 1588

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 25.230 25.298 24.547 24.671 23.876 24.049 21.847 22.222 19.164 19.827 16.502 17.465 14.519 15.682 13.858 15.070 13.200 14.448	ABS BETAM IN OUT 48.0 2.6 42.2 2.4 40.0 1.3 36.7 -0.1 39.3 -0.2 40.3 0.5 42.9 -3.0 43.3 -6.5 44.3 -9.5	48.0 2.6 42.2 2.4 40.0 1.3 36.7 -0.1 39.3 -0.2 40.3 0.5 42.9 -3.0 43.3 -6.5	TOTAL TEMP IN RATIO 307.1 0.995 305.4 0.998 305.0 0.997 303.0 0.998 300.9 0.999 299.2 1.000 297.6 1.003 297.4 1.003 297.3 1.003	TOTAL PRESS IN RATIO 11.76 0.986 11.80 0.987 11.83 0.989 11.91 0.986 11.73 0.995 11.55 0.989 11.23 1.001 11.17 1.002 11.18 0.999
RP 1 23456789	ABS VEL 1N OUT 137.3 122.1 141.6 125.6 145.3 129.1 153.1 134.0 149.3 132.7 148.0 124.1 137.6 116.2 135.9 112.8 136.8 110.4	REL VEL 1N 0UT 137.3 122.1 141.6 125.6 145.3 129.1 153.1 134.0 149.3 132.7 148.0 124.1 137.6 116.2 135.9 112.8 136.8 110.4	MERID VEL IN OUT 92.0 122.0 104.9 125.5 111.2 129.1 122.8 134.0 115.5 132.7 112.9 124.1 100.9 116.0 98.9 112.1 97.9 108.9	TANG VEL [N OUT 102.0 5.5 95.0 5.3 93.5 3.0 91.6 -0.2 94.7 -0.5 95.7 1.1 93.6 -6.0 93.2 -12.8 95.6 -18.2	WHEEL SPEED IN OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9	ÂBS MACH NO IN OUT 0.397 0.353 0.411 0.364 0.422 0.374 0.448 0.390 0.458 0.387 0.435 0.362 0.404 0.339 0.399 0.330 0.402 0.322	REL MACH NO IN OUT 0.397 0.353 0.411 0.364 0.422 0.374 0.448 0.390 0.435 0.362 0.404 0.339 0.399 0.330 0.402 0.322	MERID MACH NO IN OUT 0.266 0.352 0.305 0.363 0.323 0.374 0.359 0.397 0.332 0.362 0.296 0.339 0.291 0.327 0.288 0.318		MERID VEL R 1.326 1.196 1.160 1.092 1.149 1.099 1.151 1.133
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 7.5 10.00 1.6 15.00 -0.6 30.00 -4.3 50.00 -2.1 70.00 -1.6 85.00 0.7 90.00 1.0 95.00 1.9	DENCE DEV 18.2 17.7 16.3 13.9 12.8 12.2 8.0 4.2	D-FACT EFF 0.590 0. 0.533 0. 0.513 0. 0.478 0. 0.441 0. 0.446 0 0.446 0. 0.465 0. 0.493 0.	LOSS COEFF TOT PROF 0.136 0.136 0.123 0.123 0.099 0.099 0.109 0.109 0.042 0.042 0.088 0.088 -0.012 -0.012 -0.018 -0.018 0.009 0.009	LOSS PARAM TOT PROF 0.093 0.093 0.081 0.081 0.064 0.064 0.065 0.065 0.022 0.022 0.040 0.040 -0.005 -0.005 -0.007 -0.007 0.003 0.003

FOR STATOR 55

(f) 90 Percent design speed; reading 1594

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RP 1 2 3 4 5 6 7 8 9	RAD(I IN OUT 25.230 25.298 24.547 24.671 23.876 24.049 21.847 22.222 19.164 19.827 16.502 17.465 14.519 15.682 13.858 15.070 13.200 14.448	ABS BETAM IN OUT 38.2 -0.8 34.6 -0.6 33.5 -1.7 37.0 -1.5 38.3 0.4 41.2 -2.4 42.3 -4.9 42.9 -8.0	34.6 -0.6 33.5 -1.2 33.9 -1.7 37.0 -1.5 38.3 0.4 41.2 -2.4 42.3 -4.9	TOTAL TEMP IN RATIO 304.8 0.998 304.1 0.998 302.1 0.999 300.5 0.999 298.9 1.000 297.6 1.002 297.3 1.002 297.2 1.001	TOTAL PRESS IN RATIO 11.71 0.990 11.80 0.989 11.84 0.994 11.69 0.996 11.54 0.991 11.24 0.997 11.15 1.001 11.18 0.995
RP 1 2 3 4 5 6 7 8 9	ABS VEL 1N OUT 141.4 128.5 147.9 133.8 152.1 137.1 155.9 141.5 153.6 138.2 153.1 131.4 143.8 123.0 139.7 120.1 141.8 117.9	REL VEL IN 0UT 141.4 128.5 147.9 133.8 152.1 137.1 155.9 141.5 153.6 138.2 153.1 131.4 143.8 123.0 139.7 120.1 141.8 117.9	MERID VEL IN OUT 111.1 128.5 121.8 133.8 126.9 137.1 129.4 141.4 122.7 138.2 120.2 131.4 108.1 122.9 103.4 119.7 103.8 116.8	TANG VEL IN OUT 87.5 -1.7 84.0 -1.5 83.9 -2.8 87.1 -4.2 92.4 -3.7 94.8 0.9 94.8 -5.1 94.0 -10.4 96.5 -16.4	WHEEL SPEED IN OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0,411 0.373 0.451 0.389 0.444 0.399 0.457 0.413 0.451 0.404 0.451 0.385 0.423 0.360 0.411 0.351 0.417 0.345	REL MACH NO IN OUT 0.411 0.373 0.451 0.389 0.445 0.457 0.413 0.451 0.404 0.451 0.385 0.423 0.360 0.411 0.351 0.417 0.345	MERID MACH NO IN OUT 0.323 0.373 0.355 0.389 0.371 0.399 0.379 0.413 0.360 0.404 0.354 0.385 0.318 0.360 0.304 0.350 0.306 0.342		MERID VEL R 1.157 1.099 1.080 1.094 1.126 1.093 1.137 1.157
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 -2.2 10.00 -5.9 15.00 -7.1 30.00 -7.1 50.00 -4.4 70.00 -3.6 85.00 -1.0 90.00 -0.0	DENCE DEV 14.9 14.7 13.8 12.3 11.5 12.1 8.6 5.7 2.4	D-FACT EFF 0.521 0. 0.478 0. 0.466 0. 0.438 0. 0.424 0. 0.414 0. 0.417 0. 0.422 0. 0.456 0.	LOSS COEFF TOT PROF 0.096 0.096 0.094 0.094 0.079 0.079 0.043 0.043 0.034 0.034 0.072 0.072 0.026 0.026 -0.008 -0.008 0.040 0.040	LOSS PARAM TOT PROF 0.065 0.065 0.062 0.062 0.051 0.051 0.026 0.056 0.018 0.018 0.033 0.033 0.011 0.011 -0.003 -0.003 0.015 0.015

(g) 90 Percent design speed; reading 1595

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 25.230 25.298 24.547 24.671 23.876 24.049 21.847 22.222 19.164 19.827 16.502 17.465 14.519 15.682 13.858 15.070 13.200 14.448	31.3 -2.9 29.0 -2.6 28.9 -2.7 30.8 -3.0	31.3 -2.9 29.0 -2.6 28.9 -2.7 30.8 -3.0 33.7 -2.4 35.8 -0.8 38.5 -1.9 40.2 -4.0	TOTAL TEMP IN RATIO 303.6 0.999 303.0 0.998 302.7 0.998 301.5 0.999 300.4 0.999 298.9 1.000 297.7 1.001 297.2 1.002 297.1 1.001	TOTAL PRESS IN RATIO 11.64 0.987 11.78 0.984 11.82 0.988 11.80 0.994 11.68 0.995 11.52 0.992 11.27 0.998 11.11 1.004 11.10 0.997
RP 1 2 3 4 5 6 7 8 9	156.6 142.3 160.5 146.6 162.9 150.9 163.3 149.5 160.8 144.6 153.4 139.0 146.6 134.4	160.8 144.6 153.4 139.0 146.6 134.4	MERID VEL. 1N OUT 126.2 137.2 137.0 142.2 140.6 146.4 139.9 150.7 135.9 149.4 130.4 144.6 120.0 139.0 112.1 134.1 111.7 128.4	TANG VEL 1N OUT 76.8 -6.9 75.9 -6.4 77.5 -6.8 83.4 -7.8 90.6 -6.3 94.1 -4.5 94.5 -9.4 95.6 -16.6	0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8	ABS MACH NO IN OUT 0.431 0.400 0.458 0.415 0.470 0.428 0.479 0.442 0.481 0.439 0.474 0.425 0.452 0.408 0.432 0.395 0.434 0.380	REL MACH NO 1N OUT 0.431 0.400 0.458 0.415 0.470 0.428 0.479 0.442 0.481 0.439 0.474 0.425 0.452 0.408 0.432 0.395 0.434 0.380	MERID MACH NO IN OUT 0.368 0.399 0.401 0.415 0.412 0.428 0.411 0.442 0.400 0.438 0.385 0.425 0.354 0.408 0.330 0.394 0.329 0.377		MERID VEL R 1.088 1.038 1.042 1.077 1.099 1.109 1.158 1.196
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 -9.1 10.00 -11.6 15.00 -11.7 30.00 -10.2 50.00 -7.8 70.00 -6.0 85.00 -3.7 90.00 -2.2 95.00 -1.9	12.7 12.3 11.1 10.6 11.0 9.1 6.7	0.350 0.	LOSS COEFF TOT PROF 0.108 0.108 0.122 0.122 0.088 0.088 0.042 0.035 0.035 0.035 0.053 0.053 0.019 0.019 -0.029 -0.029 0.028 0.028	LOSS PARAM TOT PROF 0.073 0.075 0.081 0.081 0.057 0.057 0.025 0.025 0.018 0.018 0.025 0.025 0.008 0.008 -0.011 -0.011 0.010 0.010

(h) 90 Percent design speed; reading 1596

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RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 25.230 25.298 24.547 24.671 23.876 24.049 21.847 22.222 19.164 19.827 16.502 17.465 14.519 15.682 13.858 15.070 13.200 14.448	ABS BETAM IN OUT 25.6 -4.6 24.2 -3.6 24.7 -3.5 27.0 -4.1 30.0 -4.0 32.0 -2.3 34.4 -3.6 36.3 -4.2 36.7 -4.8	IN OUT 25.6 -4.6 24.2 -3.6 24.7 -3.5 27.0 -4.1 30.0 -4.0 32.0 -2.3 34.4 -3.6 36.3 -4.2	TOTAL TEMP IN RATIO 302.4 0.998 302.0 0.998 301.9 0.998 300.9 1.000 299.6 1.000 298.5 1.000 297.7 1.000 296.9 1.001	TOTAL PRESS IN RATIO 11.63 0.974 11.75 0.979 11.76 0.995 11.58 1.000 11.47 0.996 11.28 0.994 11.05 1.002 11.02 0.996
R123456789	ABS VEL IN 0UT 161.1 150.5 170.0 158.2 172.3 161.5 174.6 165.7 174.4 167.1 173.2 165.6 168.8 159.9 159.2 154.4 157.7 150.0	REL VEL IN 0UT 161.1 150.5 170.0 158.2 172.3 161.5 174.6 165.7 174.4 167.1 173.2 165.6 168.8 159.9 159.2 154.4 157.7 150.0	MERID VEL IN OUT 145.4 150.0 155.1 157.9 156.5 161.2 155.6 465.3 151.0 166.7 146.9 165.4 139.2 159.5 128.3 154.0 126.5 149.5	TANG VEL IN OUT 69.6 -12.2 69.5 -9.8 72.1 -9.7 79.2 -11.9 87.2 -11.6 91.7 -6.8 95.5 -10.1 94.2 -11.4 94.2 -12.7	0. 0.
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.472 0.441 0.500 0.464 0.501 0.488 0.516 0.493 0.513 0.489 0.500 0.472 0.471 0.456 0.466 0.443	REL MACH NO IN OUT 0.472 0.441 0.500 0.464 0.507 0.474 0.515 0.488 0.516 0.493 0.513 0.489 0.500 0.472 0.471 0.456 0.466 0.443	MERID MACH NO IN OUT 0.426 0.439 0.456 0.463 0.461 0.474 0.459 0.487 0.447 0.492 0.435 0.489 0.412 0.471 0.380 0.455 0.374 0.441		MERID VEL R 1.032 1.018 1.030 1.062 1.104 1.126 1.146 1.200 1.182
RP 1 2 3 4 5 6 7 8 9	PERCENT INC SPAN MEAN 5.00 -14.9 10.00 -16.4 15.00 -15.9 30.00 -14.0 50.00 -91.4 70.00 -9.8 90.00 -6.0 95.00 -5.7	11.0 11.8 11.5 9.9 9.0 9.4 7.3 6.5	0.412 0. 0.378 0. 0.369 0. 0.359 0. 0.336 0. 0.298 0. 0.300 0.	0.035 0.035 -0.017 -0.017	TOT PROF 0.123 0.123 0.090 0.090 0.059 2.059 0.019 0.019 -0.001 -0.001 0.012 0.012 0.014 0.014

(i) 90 Percent design speed; reading 1604

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 25.230 25.298 24.547 24.671 23.876 24.049 21.847 22.222 19.164 19.827 16.502 17.465 14.519 15.682 13.858 15.070 13.200 14.448	ABS BETAM IN OUT 18.6 -6.1 17.9 -5.2 18.3 -5.2 20.2 -5.1 23.6 -4.2 25.4 -3.6 25.7 -4.5 27.1 -6.4 28.6 -7.2	IN OUT 18.6 -6.1 17.9 -5.2 18.3 -5.2 20.2 -5.1 23.6 -4.2 25.4 -3.6 25.7 -4.5 27.1 -6.4	TOTAL TEMP IN RATIO 300.7 0.998 300.9 0.998 300.5 0.998 299.7 0.999 298.8 1.001 297.0 1.002 296.4 1.002	TOTAL PRESS IN RATIO 11.23 0.956 11.51 0.957 11.53 0.968 11.54 0.976 11.49 0.986 11.37 0.992 11.01 0.990 10.71 0.989 10.50 0.990
RP 1 2 3 4 5 6 7 8 9	187.8 209.6	REL VEL N OUT 186.4 185.5 200.6 196.4 204.0 201.5 209.3 209.2 212.1 216.4 211.5 222.7 199.6 219.6 178.5 201.9	MERID VEL 1N OUT 176.6 184.5 190.9 195.6 193.7 200.7 196.5 208.3 194.4 215.8 191.1 222.3 179.9 218.9 167.2 208.2 156.7 200.3	TANG VEL IN OUT 59.6 -19.8 61.6 -17.9 64.0 -18.4 72.1 -18.6 84.7 -16.0 90.5 -14.1 86.5 -17.2 85.6 -23.5 85.4 -25.4	0. 0.
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.552 0.550 0.597 0.585 0.601 0.625 0.626 0.635 0.650 0.634 0.671 0.598 0.662 0.562 0.631 0.532 0.606	REL MACH NO IN OUT 0.552 0.550 0.597 0.585 0.608 0.601 0.625 0.626 0.635 0.650 0.634 0.671 0.598 0.662 0.552 0.631	MERID MACH NO IN OUT 0.523 0.547 0.568 0.582 0.577 0.582 0.582 0.582 0.582 0.582 0.582 0.648 0.573 0.669 0.539 0.660 0.500 0.627 0.468 0.602		MERID VEL R 1.045 1.024 1.036 1.060 1.110 1.163 1.217 1.246 1.278
	PERCENT INCI SPAN MEAN 5.00 -21.8 10.00 -22.7 15.00 -22.3 30.00 -20.8 50.00 -17.9 70.00 -16.5 85.00 -16.5 90.00 -15.2 95.00 -13.8	DENCE DEV 9.5 10.1 9.7 8.9 8.8 8.1 6.5 4.3 3.2	D-FACT EFF 0.295 0. 0.284 0. 0.273 0. 0.257 0. 0.226 0. 0.169 0. 0.106 0. 0.105 0. 0.095 0.	LOSS COEFF TOT PROF 0.236 0.236 0.200 0.200 0.145 0.145 0.102 0.102 0.059 0.059 0.032 0.032 0.048 0.048 0.055 0.055 0.058 0.058	LOSS PARAM TOT PROF 0.160 0.160 0.132 0.132 0.094 0.094 0.060 0.060 0.031 0.031 0.015 0.015 0.020 0.020 0.022 0.022 0.021 0.021

(j) 100 Percent design speed; reading 1592

		(1)	100 1 01	cont u	Pren pl	occu, r	cauring	1002		
RP 1 2 3 4 5 6 7 8 9	RAD IN 25.230 24.547 23.876 21.847 19.164 16.502 14.519 13.858 13.200	0UT 25.298 24.671 24.049 22.222 19.827	IN 53.1 46.7 43.7 37.4 39.4 40.8	BETAM 0ÚT 4.3 4.6 3.2 2.1 1.4 1.5 -3.2 -6.7 -8.9	REL IN 53.1 46.7 43.7 37.4 49.8 43.1 43.7 44.3	BETAM OUT 4.3 4.6 3.2 2.1 1.4 1.5 -3.2 -6.7 -8.9	TOTA 10.4 312.7 310.4 306.4 306.4 301.9 299.6 299.5 298.9	0.995	TOTAL IN 12.26 12.22 12.22 12.33 12.17 11.91 11.51 11.46	PRESS RATIO 0.974 0.979 0.981 0.978 0.990 0.989 1.000 1.000 0.998
RP 1 2 3 4 5 6 7 8 9	ABS 1N 155.0 156.3 159.3 168.0 166.7 163.5 152.3 151.2	VEL 0UT. 137.2 139.3 141.5 146.2 147.7 138.8 127.8 127.8 125.0	REL IN 155.0 156.3 159.3 168.0 163.5 152.3 151.2	127.8	MERI IN 93.1 107.1 115.2 133.5 128.8 123.9 111.2 109.3	D VEL OUT 136.8 138.8 141.3 146.1 147.7 138.8 127.6 124.2 120.2	TAN 1N 123.9 113.7 110.1 102.0 105.9 106.7 104.5 105.4		HHEEL IN 0. 0. 0. 0. 0. 0.	SPEED OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9	IN 0.446	0.409 0.409 0.409 0.425 0.430 0.430 0.355 0.355	IN 0.446 0.451 0.461	ACH NO OUT 0.395 0.402 0.409 0.425 0.430 0.405 0.373 0.365 0.355	MERID M 1N 0.268 0.309 0.333 0.390 0.377 0.364 0.327 0.321 0.317	ACH NO OUT 0.394 0.400 0.408 0.424 0.430 0.405 0.372 0.362 0.351			MERID VEL R 1.470 1.296 1.227 1.095 1.147 1.120 1.148 1.136 1.115	
RP - 1 2 3 4 5 6 7 8 9	PERCENT SPAN 5.00 10.00 15.00 30.00 70.00 85.00 90.00 95.00	INCI MEAN 12.6 6.2 3.1 -3.6 -2.0 -1.1 0.9 1.4	DENCE	DEV 19.9 19.9 18.1 16.1 14.5 13.2 7.7 4.0	D-FACT 0.615 0.544 0.525 0.468 0.431 0.448 0.471 0.491	0. 0. 0. 0.	LOSS C TOT 0.201 0.160 0.138 0.148 0.067 0.074 0.001 0.002	PROF 0.201 0.160 0.138 0.148 0.067 0.074 0.001	LOSS PA TOT 0.136 0.106 0.089 0.088 0.035 0.034 0.000 0.001	PROF 0.136 0.106 0.089 0.088 0.035 0.034 0.000 0.001

(k) 100 Percent design speed; reading 1614

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 25.230 25.298 24.547 24.671 23.876 24.049 21.847 22.222 19.164 19.827 16.502 17.465 14.519 15.682 13.858 15.070 13.200 14.448	ABS IN 41.1 38.4 37.1 36.1 37.7 39.1 41.8 42.5 43.0	BETAM OUT -0.5 -0.1 -1.4 -1.7 0.3 -3.1 -5.3 -8.9	REL IN 41.1 38.4 37.1 36.1 37.7 39.1 41.8 42.5 43.0	BETAM OUT -0.5 -0.1 -1.4 -1.7 0.3 -3.1 -5.3 -8.9	IN RA 310.1 0. 308.6 0. 308.0 0. 306.7 0. 304.1 0. 302.1 1. 300.1 1. 299.8 1.	EMP TOT T10 IN 996 12.1 998 12.3 997 12.4 999 12.1 000 11.9 002 11.5 002 11.4	7 0.985 3 0.985 0 0.987 7 0.992 4 0.987 4 0.997 4 1.000
RP 1 2 3 4 5 61 8 9	ABS VEL IN OUT 155.0 139.1 163.6 145.5 168.0 148.6 175.4 155.5 171.7 151.6 169.8 143.7 159.8 143.7 156.7 127.4	REL IN 155.0 163.6 168.0 175.4 171.7 169.8 158.7 155.8 156.7	VEL 0UT 139.1 145.5 148.6 155.5 151.6 143.7 133.3 130.3	MERI IN 116.7 128.1 134.0 141.7 135.8 131.7 118.2 114.8 114.6	0UT 139.1 145.5 148.5	101.9 - 101.7 - 101.3 - 103.3 - 105.0 - 107.1 105.9 - 105.4 -1	TEL WHE INT IN 1.2 0.3 0.3 0.3 3.7 0.3 8 0.4.6 0.6 0.7 7.3 0.2.1 0.9.6	0.
RP: 23456789	ABS MACH NO IN OUT 0.448 0.401 0.475 0.421 0.489 0.431 0.513 0.453 0.503 0.442 0.499 0.420 0.467 0.389 0.458 0.381 0.461 0.372	REL M 1N 0.448 0.475 0.489 0.513 0.503 0.499 0.467 0.458 0.461	0UT 0.401	MERID M 1N 0.337 0.372 0.390 0.414 0.398 0.387 0.348 0.338	0.401 0.421		MERI VEL. 1.19 1.13 1.10 1.09 1.11 1.09 1.12	R 15 8 6 6 1 6 0
RP 1 2 3 4 5 6 7 8 9	SPAN MEAN 5.00 0.7 10.00 -2.1 15.00 -3.5 30.00 -4.9 50.00 -3.7	DENCE	DEV - 15.1 15.2 13.6 12.7 11.3 12.0 7.8 5.4 1.6	D-FACT 0.556 0.524 0.518 0.474 0.447 0.433 0.441 0.448 0.479	0. 0. 0. 0. 0. 0. 0.	0.098 0. 0.101 0. 0.098 0. 0.076 0. 0.050 0. 0.081 0. 0.023 0. -0.003 -0.	0F TOT 098 0.06 101 0.06 098 0.06 076 0.04 050 0.02 081 0.03 023 0.00	3 0.063 5 0.045 6 0.026 7 0.037 9 0.009 1 -0.001

FOR STATOR 55

(1) 100 Percent design speed; reading 1613

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 25.230 25.298 24.547 24.671 23.876 24.049 21.847 22.222 19.164 19.827 16.502 17.465 14.519 15.682 13.858 15.070 13.200 14.448	IN 33.6 31.8 31.4 32.1 35.2 36.8	BETAM OUT -2.6 -2.3 -2.2 -2.7 -2.3 -0.5 -3.1 -5.6 -8.3	REL IN 33.6 31.8 31.4 32.1 35.2 36.8 39.7 41.2	BETAM OUT -2.6 -2.3 -2.7 -2.3 -0.5 -3.1 -5.6 -8.3	TOTA IN 307.6 307.3 307.2 305.4 303.7 302.0 300.5 299.6	RATIO 0.999 0.998 0.997 0.999 0.998 1.000 1.001	TOTAL IN 12.06 12.24 12.31 12.28 12.12 11.93 11.54 11.38	PRESS RATIO 0.986 0.981 0.983 0.993 0.985 0.989 0.998 1.004 0.997
RP 1 2 3 4 5 6 7 8 9	ABS VEL IN OUT 162.7 148.2 172.3 154.0 176.7 158.2 179.5 164.4 179.0 157.7 177.8 156.2 167.3 148.2 161.5 143.4 162.8 140.1	172.3 176.7 179.5 179.0 177.8 167.3	0UT 148.2 154.0 158.2 164.4 157.7 156.2	MER I IN 135.5 146.4 150.8 152.1 146.3 142.4 128.7 121.6 121.9	D VEL OUT 148.1 153.9 158.1 164.3 157.6 156.2 147.9 142.7 138.7	TAN IN 90.1 90.8 92.1 95.3 103.1 106.5 106.8 106.3	G VEL OUT -6.8 -6.2 -6.0 -7.9 -6.3 -1.3 -7.9 -14.1 -20.1	WHEEL IN 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.	SPEED OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.473 0.430 0.502 0.447 0.516 0.460 0.526 0.480 0.526 0.461 0.524 0.458 0.493 0.434 0.476 0.420 0.480 0.410	0.502 (0.516 (0.526 (0.526 (0.524 (0.493 (0.476 (0.	CH NO OUT 0.430 0.447 0.460 0.480 0.46! 0.458 0.434 0.420	MERID M 1N 0.394 0.427 0.440 0.446 0.430 0.420 0.379 0.358 0.359	OUT 0.429 0.447 0.460 0.480 0.461 0.461 0.458 0.433 0.418			MERID VEL R 1.093 1.051 1.049 1.080 1.077 1.097 1.149 1.174 1.138	
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 -6.9 10.00 -8.7 15.00 -9.2 30.00 -8.9 50.00 -6.3 70.00 -5.0 85.00 -2.5 90.00 -1.1 95.00 -0.9	DENCE	DEV 13.0 13.0 12.8 11.3 10.7 11.2 7.9 5.0 2.2	D-FACT 0.495 0.479 0.463 0.423 0.436 0.392 0.384 0.394 0.423	EFF 0. 0. 0. 0. 0. 0.	LOSS C TOT 0.102 0.123 0.103 0.042 0.086 0.063 0.011 -0.025 0.019	PROF 0.102 0.103 0.042 0.086 0.063 0.011	LOSS PA TOT 0.069 0.082 0.066 0.025 0.045 0.029 0.004 -0.010	PROF 0.069 0.082 0.066 0.025 0.045 0.029 0.004

FOR STATOR 55

(m) 100 Percent design speed; reading 1611

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 25.230 25.298 24.547 24.671 23.876 24.049 21.847 22.222 19.164 19.827 16.502 17.465 14.519 15.682 13.858 15.070 13.200 14.448	ABS BETAN IN OUT 27.9 -4 26.3 -4 26.1 -3 27.9 -3 30.8 -5 32.9 -3 35.8 -4 37.0 -3	0 26.3 -4.0 7 26.1 -3.7 9 27.9 -3.9 5 30.8 -3.5 5 32.9 -2.3 0 35.8 -4.0	TOTAL TEMP IN RATIO 306.6 0.998 306.1 0.998 305.9 0.997 304.7 0.998 301.6 0.999 301.6 0.999 300.4 1.000 299.4 1.001	TOTAL PRESS IN RATIO 12.04 0.971 12.23 0.971 12.26 0.978 12.21 0.988 12.02 0.993 11.84 0.981 11.55 0.985 11.29 0.986 11.25 0.969
R 1 23456789	ABS VEL IN OUT 179.8 165.9 188.9 173.5 192.4 177.6 195.8 182.9 194.2 182.1 192.9 177.3 184.7 169.1 175.3 160.9 173.6 148.7	REL VEL IN OUT 179.8 165.9 188.9 173.1 192.4 177.6 195.8 182.9 194.2 182.1 192.9 177.1 184.7 169. 175.3 160.9 173.6 148.	5 169.3 173.1 6 172.7 177.3 9 173.1 182.4 166.7 181.8 5 162.0 177.2 1 149.9 *168.7 9 140.1 160.5	83.7 -12.0 84.8 -11.4 91.6 -12.4 99.5 -11.0 104.7 -7.1 107.9 -11.8 105.4 -10.8	0. 0. 0 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.526 0.484 0.555 0.508 0.521 0.578 0.538 0.574 0.537 0.572 0.523 0.547 0.499 0.519 0.474 0.514 0.437	REL MACH NIN 0UT 0.526 0.498 0.555 0.500 0.500 0.578 0.537 0.572 0.552 0.572 0.5514 0.437	IN OUT 4 0.465 0.483 8 0.497 0.507 1 0.508 0.520 7 0.493 0.537 7 0.493 0.523 5 0.480 0.523 9 0.444 0.498 4 0.415 0.473		MERID VEL R 1,041 1,022 1,026 1,054 1,090 1,094 1,126 1,146 1,071
RP 1 2 3 4 5 6 7 8 9	PERCENT INC SPAN MEAN 5.00 -12.6 10.00 -14.2 15.00 -14.5 30.00 -10.6 70.00 -8.9 85.00 -6.4 90.00 -5.4 95.00 -5.2	DENCE DE 10.6 11. 11. 11. 10. 9. 6.9 6.6 6.6 4.	3 0.449 0. 5 0.417 0. 5 0.399 0. 6 0.380 0. 6 0.357 0. 6 0.340 0. 9 0.340 0. 9 0.332 0.	LOSS COEFF TOT PROF 0.170 0.170 0.152 0.152 0.114 0.114 0.060 0.060 0.034 0.034 0.077 0.077 0.106 0.106 0.085 0.085 0.187 0.187	0.043 0.043 0.033 0.033

FOR STATOR 55

(n) 100 Percent design speed; reading 1603

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 25.230 25.298 24.547 24.671, 23.876 24.049 21.847 22.222 19.164 19.827 16.502 17.465 14.519 15.682 13.858 15.070 13.200 14.448	ABS BETAM 1N OUT 22.0 -5. 20.7 -4. 21.7 -4. 23.7 -4. 26.8 -4. 28.3 -3. 29.9 -4. 31.6 -6. 32.1 -7.	IN OUT 4 22.0 -5.4 7 20.7 -4.7 7 21.7 -4.7 2 23.7 -4.5 2 26.8 -4.2 2 26.8 -4.2 6 28.3 -3.6 9 29.9 -4.9 2 31.6 -6.2	TOTAL TEMP IN RATIO 305.2 0.996 304.8 0.997 304.7 0.998 303.6 0.999 302.7 0.999 301.0 1.001 298.5 1.003 297.8 1.003 296.5 1.006	TOTAL PRESS IN RATIO 11.77 0.955 11.98 0.960 12.01 0.969 11.97 0.986 11.89 0.995 11.71 0.994 11.24 1.000 10.95 1.002 10.67 1.012
RP - 254561-89	218.7 216.6 216.5 217.3 200.1 210.5 189.7 201.5	REL VEL IN OUT 195.8 185.6 207.0 196.1 210.3 200.4 215.4 210.0 218.7 216.6 216.5 217.3 200.1 210.5 189.7 201.5 179.0 194.7	193.6 195.4 195.4 199.7 197.2 209.3 195.1 216.0 190.6 216.8 173.5 209.7 161.5 200.3	TANG VEL IN OUT 73.3 -17.5 73.1 -16.2 77.7 -16.6 86.7 -16.6 98.7 -15.8 102.7 -13.5 99.7 -18.0 99.5 -21.6 95.1 -23.8	0. 0.
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.577 0.547 0.613 0.580 0.624 0.593 0.642 0.624 0.653 0.647 0.648 0.650 0.598 0.630 0.566 0.630 0.581	REL MACH NO IN OUT 0.577 0.547 0.547 0.598 0.624 0.653 0.647 0.648 0.650 0.598 0.656 0.656 0.566 0.533 0.581	IN 0UT 0.535 0.544 0.574 0.578 0.580 0.591 0.587 0.622 0.583 0.645		MERID VEL R 1.018 1.009 1.002 1.061 1.107 1.138 1.208 1.208 1.274
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 -18.5 10.00 -19.8 15.00 -18.9 30.00 -17.3 50.00 -13.5 85.00 -12.3 90.00 -10.7 95.00 -10.3	DENCE DEV 10.2 10.6 10.2 9.5 8.9 8.2 6.0 4.5	0.181 0.	LOSS COEFF TOT PROF 0.222 0.222 0.177 0.177 0.133 0.133 0.056 0.019 0.019 0 024 0.024 0.001 0.001 -0.012 -0.012 -0.069 -0.069	LOSS PARAM TOT PRCF 0.151, 0.151 0.117 0.117 0.086 0.086 0.033 0.010 0.011 0.011 0.000 0.000 -0.005 -0.005 -0.025 -0.025

FOR STATOR 55

(o) 110 Percent design speed; reading 1694

		•				•	_			
RP 1 2 3 4 5 6 7 8 9	RAD I IN 25.230 2 24.547 2 23.876 2 21.847 2 19.164 1 16.502 1 14.519 1 13.858 1	24.671 24.049 22.222 9.827 7.465	44.3 39.3 36.7 36.2 38.7 39.6	2.5 2.3 1.2 0.5 0.6	44.3 39.3 36.7 36.2 38.7 39.6	0UT 2.5 2.3 1.2 0.5 0.6 1.6 -2.4	IN 315.4 314.0 312.7 310.0 307.8 305.1 302.6 302.3	0.994 0.997 0.996 0.998 0.996 0.998 1.001	TOTAL IN 12.70 12.81 12.82 12.88 12.67 12.37 11.75	
RP 1 23 4 5 6 7 8 9	IN 171.0 176.8 181.2 188.1 188.1 186.8 174.0 170.5	VEL 0UT 151.8 156.6 159.4 167.1 163.3 152.8 142.7 141.1 139.9	IN 171.0 176.8 181.2 188.1 188.1 186.8 174.0	0UT 151.8 156.6 159.4 167.1 163.3 152.8 142.7 141.1	MERII 1N 122.3 136.9 145.3 151.9 146.9 144.0 128.0 124.0	151.6 156.5 159.3 167.1 163.3 152.8	IN 119.5 111.9 108.2 111.0 117.5 119.1 117.9	3.5	IN 0. 0.	SPEED OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9	0.511 0.525 0.549 0.551 0.550 0.512	OUT 0.436 0.451 0.460 0.485 0.476 0.446 0.416	IN 0.492 0.511 0.525 0.549 0.551 0.550	0.436 0.451 0.460 0.485 0.476 0.446 0.416	MERID M. 1N 0.352 0.395 0.421 0.443 0.430 0.423 0.377 0.365 0.367	OUT 0.435 0.450 0.460 0.485 0.476 0.445 0.416 0.410 0.403			MERID VEL R 1.240 1.143 1.096 1.100 1.112 1.061 1.114 1.134 1.108	
RP 1 2 3 4 5 6 7 8 9	PERCENT SPAN 5.00 10.00 15.00 30.00 50.00 70.00 85.00 90.00	INCII MEAN 3.9 -1.3 -3.9 -4.8 -2.8 -2.2 0.4 1.0	DENCE	DEV 18.1 17.6 16.2 14.6 13.6 13.3 8.6 5.7	D-FACT 0.562 0.510 0.493 0.455 0.450 0.455 0.460 0.458 0.481	EFF 0. 0. 0. 0. 0. 0.	LOSS CO TOT 0.140 0.128 0.110 0.072 0.063 0.108 0.022 -0.017	PROF 0.140 0.128 0.110 0.072 0.063 0.108 0.022 -0.017	LOSS PATOT 0.096 0.085 0.071 0.043 0.033 0.049 0.009 -0.007 -0.002	PROF 0.096 0.085 0.071 0.043 0.033 0.049 0.009

(p) 110 Percent design speed; reading 1693

RP 1 2 3 4 5 6 7 8 9	RAD IN 25.230 24.547 23.876 21.847 19.164 16.502 14.519 13.858 13.200	0UT 25.298 24.671 24.049 22.222 19.827 17.465 15.682	ABS 1N 33.2 30.3 29.8 31.8 34.8 36.1 38.9 40.1	BETAM OUT -3.1 -2.6 -2.5 -2.5 -2.0 0.0 -2.7 -4.2 -7.6	IN 33.2 30.3 29.8 31.8 34.8 36.1 38.9 40.1	BETAM 0UT -3.1 -2.6 -2.5 -2.5 -2.0 0.0 -2.7 -4.2 -7.6	IN 312.2 310.9 310.4 309.3 306.9 304.9 302.7 302.0	0.999	TOTAL IN 12.52 12.77 12.84 12.56 12.33 11.86 11.61	PRESS RATIO 0.983 0.978 0.982 0.988 0.993 0.978 0.986 1.000 0.993
RP 1 2 3 4 5 6 7 8 9	181.7 192.3 196.8 202.8 200.1 201.5	VEL 0UT 165.3 172.3 176.6 182.1 178.9 172.4 164.5 161.8 157.1	REL IN 181.7 192.3 196.8 202.8 200.1 201.5 191.2 182.7 182.3	VEL 0UT 165.3 172.3 176.6 182.1 178.9 172.4 164.5 161.8	MERI. IN 152.0 166.1 170.7 172.4 164.2 162.8 148.8 139.8 139.9	D VEL 0UT 165.1 172.2 176.4 181.9 178.8 172.4 164.3 161.4 155.7	99.6 96.9 97.9 106.7 114.3 118.6 120.0	-6.1 0.0 -7.8	IN 0. 0. 0.	
RP 1 2 3 4 5 6 7 8 9	ABS M. IN 0.527 0.561 0.575 0.595 0.589 0.596 0.565 0.539	ACH NO OUT 0.478 0.500 0.514 0.532 0.524 0.524 0.483 0.475 0.460	REL M IN 0.527 0.561 0.595 0.595 0.596 0.565 0.539 0.539	ACH NO OUT 0.478 0.500 0.514 0.532 0.524 0.524 0.483 0.475 0.460	MERID M IN 0.441 0.499 0.506 0.484 0.481 0.440 0.413	OUT 0.478 0.500 0.513 0.531 0.524 0.506		·	MERID VEL R 1.086 1.037 1.033 1.055 1.089 1.058 1.104 1.1154	
RP 1 2 3 4 5 6 7 8	PERCENT SPAN 5.00 10.00 15.00 30.00 50.00 70.00 85.00 90.00 95.00	INC! MEAN -7.2 -10.3 -10.8 -9.2 -6.6 -5.8 -3.3 -2.2		DEV 12.6 12.7 12.4 11.6 11.1 11.7 8.2 6.5 2.8	D-FACT 0.497 0.464 0.449 0.436 0.417 0.407 0.403 0.381 0.411	0. 0. 0. 0.	LOSS C TOT 0.098 0.115 0.088 0.056 0.035 0.101 0.074 -0.002 0.041	PROF 0.098 0.115 0.088 0.056 0.035 0.101 0.074 -0.002	LOSS P TOT 0.067 0.076 0.057 0.033 0.018 0.046 0.030 -0.001	PROF 0.067 0.076 0.057 0.033 0.018 0.046 0.030

FOR STATOR 55

(q) 110 Percent design speed; reading 1692

RP 1 2 3 4 5 6 7 8	RADII IN OUT 25.230 25.298 24.547 24.671 23.876 24.049 21.847 22.222 19.164 19.827 16.502 17.465 14.519 15.682 13.858 15.070 13.200 14.448	27.2 - 26.3 - 26.4 - 28.0 - 30.3 - 30.9 - 33.7 - 34.9 -	AM RELL UT IN 3.6 27.2 3.1 26.3 3.1 26.4 33.0 28.0 2.0 30.3 0.4 30.9 3.5 33.7 4.6 34.9 7.0 34.6	-3.1 -3.1 -3.0 -2.0 -0.4 -3.5 -4.6	310.8 309.6 309.0 308.3 306.3 306.3 301.2 301.2	RATIO 0.995 0.998 0.998 0.997 0.998 0.999	TOTAL IN 12.31 12.55 12.60 12.59 12.44 12.17 11.52 11.00	RATIO 0.968 0.967
RP 1 2 3 4 5 6 7 8 9	ABS VEL IN OUT 201.3 184.9 210.8 191.9 214.3 196.4 220.6 205.4 222.6 209.3 221.5 208.0 203.4 199.6 194.0 195.6 185.9 188.7	REL VEL IN 00/ 201.3 184 210.8 191 214.3 196 220.6, 205 222.6 209 221.5 208 203.4 199 194.0 195 185.9 188	T IN 179.0 189.0 191.9 194.4 192.2 190.0 169.3 159.2	209.2 208.0 199.3 195.0	93.4 95.4 103.5 112.2 113.9 112.8	OUT -11.5 -10.4 -10.5 -10.6 -7.3 -1.3	WHEEL IN 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.	SPEED OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO 1N OUT 0.589 0.539 0.620 J.562 0.632 0.576 0.653 0.605 0.662 0.620 0.661 0.618 0.606 0.593 0.577 0.581 0.552 0.559	REL MACH 1 IN 0U 0.589 0.55 0.620 0.56 0.632 0.5 0.663 0.66 0.662 0.62 0.661 0.65 0.5577 0.55 0.552 0.55	IN 1N 0.524 0.556 0.566 0.577 0.571 18 0.567 0.504 31 0.473	0.538 0.561 0.576 0.605 0.619 0.618 0.592 0.579			MERID VEL R 1.031 1.014 1.022 1.053 1.088 1.095 1.177 1.225 1.224	
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 -13.2 10.00 -14.2 15.00 -14.2 30.00 -11.2 70.00 -11.2 70.00 -8.5 90.00 -7.5 95.00 -7.8	DENCE DE 12. 12. 11. 11. 11. 7. 6. 3.	1 0.432 2 0.416 9 0.402 1 0.374 0 0.338 4 0.293 5 0.261 1 0.238	0. 0. 0. 0. 0. 0. 0. 0.	0.153 0 0.146 0 0.105 0 0.038 0 0.011 0	ROF 1.153 1.146 1.105 1.038 1.011 1.035 1.001	0.104 0.097 0.068 0.023 0.006 0.016	PROF- 0.104 0.097 0.068 0.023 0.006 0.016 0.000 0.026

FOR STATOR 55

(r) 120 Percent design speed; reading 1695

RP 1 2 3 4 5 6 7 8 9	RAD IN 25.230 24.547 23.876 21.847 19.164 16.502 14.519 13.858 13.200	0UT 25.298 24.671 24.049 22.222 19.827 17.465 15.682	ABS IN 45.7 39.3 37.6 36.3 39.0 39.7 43.1 43.4 43.5	BETAM OUT 3.5 2.1 0.9 1.0 1.2 1.5 -3.3 -6.2	REL IN 45.7 39.3 37.6 36.3 39.0 39.7 43.1 43.4 43.5	BETAM OUT 3.5 2.1 0.9 1.0 1.5 -3.3 -6.2	TOTA [N 320.8 318.0 317.0 314.1 311.9 308.6 305.4 304.6	L TEMP RATIO 0.994 0.997 0.996 0.998 0.995 0.998 1.002 1.003	TOTAL IN 13.21 13.36 13.46 13.23 12.85 12.16 11.99	PRESS RATIO 0.977 0.978 0.977 0.982 0.987 0.980 0.999 1.011
RP 1 2 3 4 5 6 7 8 9	ABS [N 186.1 191.9 197.1 204.9 205.9 190.3 185.1 185.7	VEL 0UT 167.0 171.2 173.8 181.7 179.3 169.2 157.3 156.3 151.9	REL IN 186.1 191.9 197.1 204.9 206.9 205.8 190.3 185.1 185.7	VEL 0UT 167.0 171.2 173.8 181.7 179.3 169.2 157.3 156.3	MERI. 1N 130.0 148.6 156.1 165.2 160.9 158.5 138.8 134.4	D VEL 0UT 166.7 171.1 173.8 181.7 179.3 169.1 157.0 155.4 149.6	TAN IN 133.2 121.5 120.3 121.2 130.0 131.4 130.1 127.2 127.8	G VEL OUT 10.1 6.2 2.8 3.0 3.8 4.6 -8.9 -16.8 -26.3	WHEEL IN 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.	SPEED OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9	ABS M IN 0.533 0.553 0.570 0.597 0.605 0.606 0.560 0.544	ACH NO OUT 0.477 0.491 0.500 0.526 0.521 0.492 0.458 0.455	REL M 1N 0.533 0.553 0.570 0.597 0.605 0.606 0.560 0.544	ACH NO OUT 0.477 0.491 0.500 0.526 0.521 0.492 0.458 0.455 0.442	MERID M IN 0.372 0.428 0.451 0.481 0.471 0.466 0.409 0.395 0.397	0.476 0.491 0.500 0.526 0.521			MERID VEL R 1.282 1.151 1.113 1.100 1.114 1.067 1.131 1.156 1.111	
RP 1 2 3 4 5 6 7 8 9	PERCENT SPAN 5.00 10.00 15.00 30.00 50.00 70.00 85.00 90.00 95.00	INCI MEAN 5.2 -1.3 -3.0 -4.7 -2.5 -2.2 0.9 1.1	DENCE	DEV 19.1 17.4 15.9 15.0 14.2 13.3 7.7 4.5	D-FACT 0.553 0.506 0.502 0.453 0.448 0.452 0.461 0.449	EFF 0. 0. 0. 0. 0. 0.	LOSS C TOT 0.133 0.117 0.116 0.083 0.061 0.090 0.004 -0.058 -0.010	PROF 0.133 0.117 0.116 0.083 0.061 0.090 0.004 -0.058	LOSS P TOT 0.091 0.075 0.075 0.050 0.032 0.041 0.001 -0.023	PROF 0.091 0.078 0.075 0.050 0.032 0.041 0.001

FOR STATOR 55

(s) 120 Percent design speed; reading 1696

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 25.230 25.298 24.547 24.671 23.876 24.049 21.847 22.222 19.164 19.827 16.502 17.465 14.519 15.682 13.858 15.070 15.200 14.448	ABS IN 42.0 34.4 32.9 33.7 35.7 37.4 40.6 40.9	BETAM OUT -0.9 -0.3 -0.5 -0.7 -0.7 1.0 -2.4 -4.2 -8.3	REL IN 42.0 34.4 32.9 33.7 35.7 40.6 40.6	BETAM OUT -0.9 -0.3 -0.5 -0.7 -0.7 1.0 -2.4 -4.2 -8.3	TOTA IN 319.6 316.6 313.7 310.9 308.0 305.4 304.5 303.5	L TEMP RATIO 0.993 0.995 0.995 0.997 0.996 0.998 1.000 1.002	TOTAL IN 12.89 13.45 13.40 13.10 12.77 12.12 11.90 11.85	PRESS RATIO 0.991 0.975 0.974 0.990 0.991 0.993 0.999 1.011
RP 1 2 3 4 5 6 7 8 9	ABS VEL IN OUT 190.4 173.4 204.7 178.8 212.5 183.8 218.1 193.3 216.2 189.4 213.7 181.0 200.3 171.1 193.2 168.1 191.3 160.6	204.7 212.5 218.1 216.2 213.7 200.3 193.2	VEL OUT 173.4 178.8 183.8 193.3 189.4 181.0 171.1 168.1 160.6	IN 141.5 168.9	178.8 183.8 193.3	TAN IN - 127.5 115.6 115.4 121.0 126.2 129.9 130.4 126.6 124.5	G VEL OUT -2.7 -0.9 -1.6 -2.5 -2.3 3.1 -7.1 -12.2 -23.1	WHEEL IN 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.	SPEED OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.547 0.497 0.593 0.515 0.639 0.562 0.636 0.553 0.631 0.529 0.591 0.500 0.570 0.469	0.593 0.618 0.639 0.636 0.631 0.591 0.570	CH NO OUT 0.497 0.515 0.531 0.562 0.553 0.529 0.500 0.492	MERID M 1N 0.406 0.489 0.519 0.532 0.516 0.501 0.449 0.431	ACH NO OUT 0.497 0.515 0.531 0.562 0.553 0.529 0.500 0.464			MERID VEL R 1.225 1.058 1.030 1.065 1.078 1.067 1.124 1.149	
RP 1 2 3 4 5 6 7 8 9	PERCENT INC SPAN MEAN 5.00 1.5 10.00 -6.2 15.00 -7.7 30.00 -7.3 50.00 -5.7 70.00 -4.4 85.00 -1.4 95.00 -1.8	IDENCE	DEV 14.8 15.0 14.5 13.3 12.3 12.7 8.6 6.5 2.2	D-FACT 0.555 0.503 0.490 0.448 0.432 0.417 0.416 0.400 0.439	EFF 0. 0. 0. 0. 0. 0.	LOSS C TOT 0.048 0.120 0.113 0.040 0.036 0.070 0.006 -0.053 -0.011	PROF 0.048 0.120 0.113 0.040 0.036 0.070 0.006 -0.053	LOSS PA TOT 0.033 0.080 0.073 0.024 0.019 0.032 0.003 -0.021 -0.004	PROF 0.033 0.080 0.073 0.024 0.019 0.032 0.003 -0.021

FOR STATOR 55

(t) 120 Percent design speed; reading 1697

RP 1 2 3 4 5 6 7 8 9	RADII IN OUT 24.729 24.71 24.026 24.02 23.322 23.34 21.173 21.28 18.321 18.54 15.540 15.79 13.541 13.74 12.906 13.05 12.289 12.37	IN 4 0.0 8 0.0 3 0.0 5 0.0 2 0.0 9 0.0 1 0.0 6 0.0	BETAM OUT 33.6 29.3 28.5 30.9 33.3 33.3 36.7 37.6	IN 49.8 48.8 48.3	BETAM 0UT 38.3 34.3 32.0 25.3 15.6 7.6 1.9 -0.2 0.3	TOTA IN 289.1 288.4 287.9 287.9 287.9 287.9 287.9	L TEMP RAT10 1.095 1.091 1.085 1.079 1.068 1.056 1.053 1.048	TOTAL IN 10.04 10.13 10.14 10.15 10.14 10.15	PRESS RAT10 1.238 1.271 1.284 1.285 1.276 1.245 1.166 1.147
RP 1 2 3 4 5 6 7 8 9	ABS VEL IN OUT 210.0 205. 210.9 222. 208.2 228. 209.3 232. 208.9 238. 208.8 240. 209.8 220. 207.1 215. 201.7 206.	IN 2 325.4 1 320.0 1 312.9 3 298.2 3 278.4 1 260.6 9 249.8 6 244.2	VEL 0UT 217.7 234.4 236.3 220.5 206.8 202.4 177.2 170.7 165.7	MERII IN 210.0 210.9 208.2 209.3 208.9 208.8 209.8 207.1 201.7	D VEL OUT 170.9 193.7 200.5 199.3 199.2 200.7 177.1 170.7 165.7	TAN IN 0.0 0.0 0.1 0.0 0.1 0.1	G VEL OUT 113.6 108.7 108.7 119.3 130.8 131.8 132.1 131.6 123.5	WHEEL IN 248.6 240.6 233.6 212.5 184.0 155.9 135.8 129.5 123.5	SPEED OUT 248.5 240.6 233.8 213.6 186.2 158.5 137.8 131.0 124.4
RP 1 2 3 4 5 6 7 8 9	ABS MACH N IN OUT 0.641 0.59 0.644 0.65 0.636 0.67 0.640 0.68 0.639 0.70 0.639 0.71 0.642 0.65 0.633 0.64	IN 6 0.993 0 0.977 0 0.955 6 0.912 8 0.851 8 0.797 9 0.764 3 0.746	ACH NO OUT 0.632 0.686 0.694 0.651 0.614 0.605 0.528 0.509 0.493	MERID M IN 0.641 0.636 0.640 0.639 0.639 0.642 0.633 0.615	ACH NO OUT 0.496 0.567 0.589 0.588 0.591 0.600 0.528 0.509 0.493			MERID VEL R 0.814 0.918 0.963 0.952 0.953 0.961 0.844 0.825 0.821	
RP 1 23 4 5 6 7 8 9	SPAN ME 5.00 -0 10.00 -1 15.00 -1 30.00 -2 50.00 -2 70.00 -2 85.00 -1 90.00 -0	.5 .4 .3 .6 .9	DEV 10.6 10.2 10.8 11.8 11.4 12.4 12.9 12.8 15.2	0.455	0.667 0.779 0.825 0.869 0.913 0.951 0.795 0.749 0.712	LOSS C TOT 0.208 0.138 0.112 0.086 0.059 0.032 0.118 0.142 0.155	OEFF PROF 0.208 0.138 0.112 0.086 0.059 0.032 0.118 0.142 0.155	LOSS P TOT 0.091 0.063 0.051 0.041 0.028 0.015 0.052 0.061 0.066	ARAM PROF 0.091 0.063 0.051 0.041 0.028 0.015 0.052 0.061

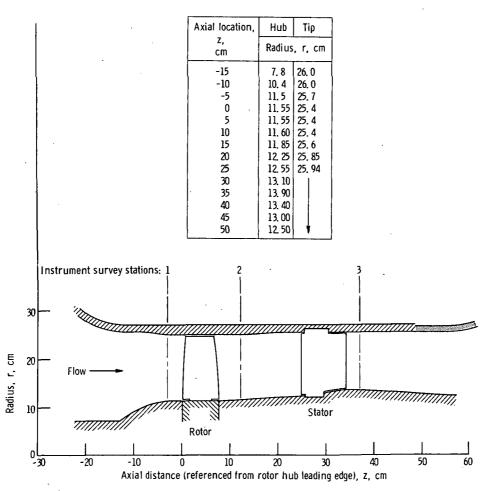


Figure 1. - Flow path of fan stage 55-55.

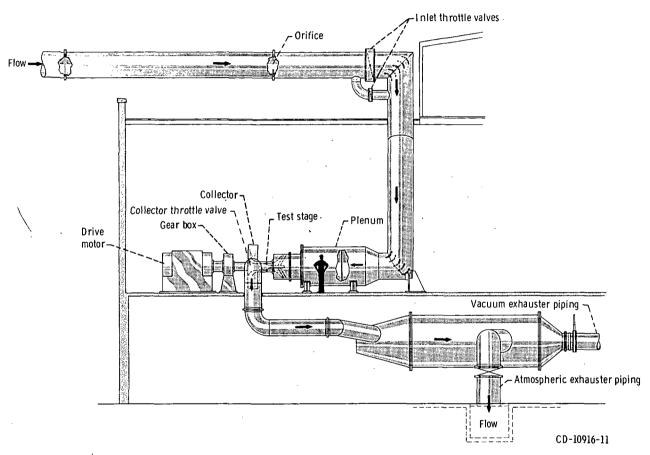


Figure 2. - Single-stage compressor facility.

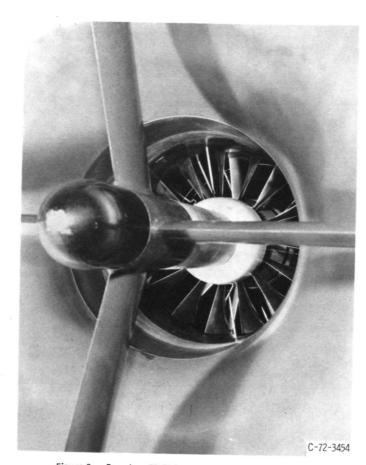
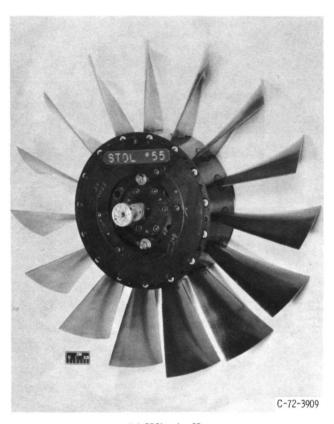
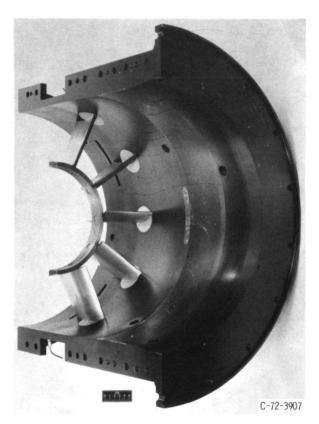


Figure 3. - Fan stage 55-55 in compressor research facility.

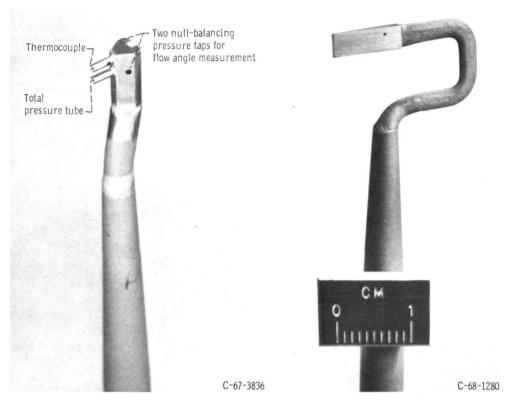




(a) STOL rotor 55.

Figure 4. - Fan stage 55-55.

(b) STOL stator 55.



(a) Combination total pressure, total temperature, and flow angle probe.

(b) Static pressure probe; $8^{\rm O}$ C-shaped wedge.

Figure 5. - Survey probes.

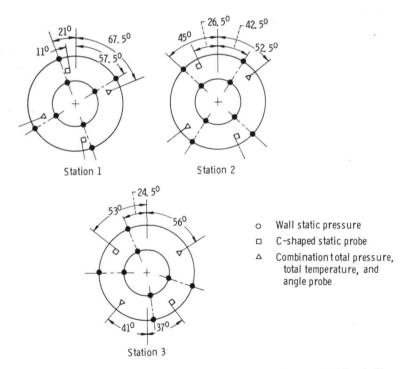
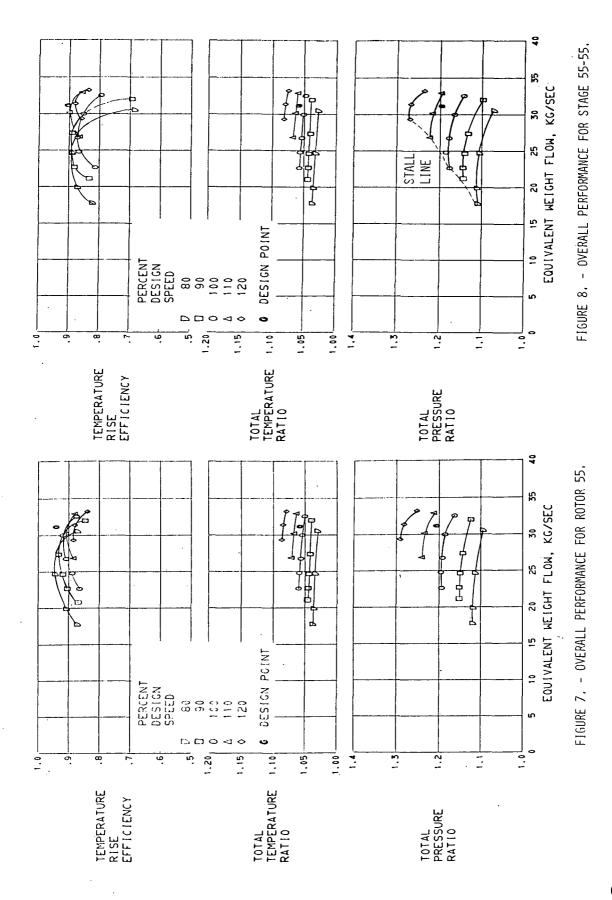
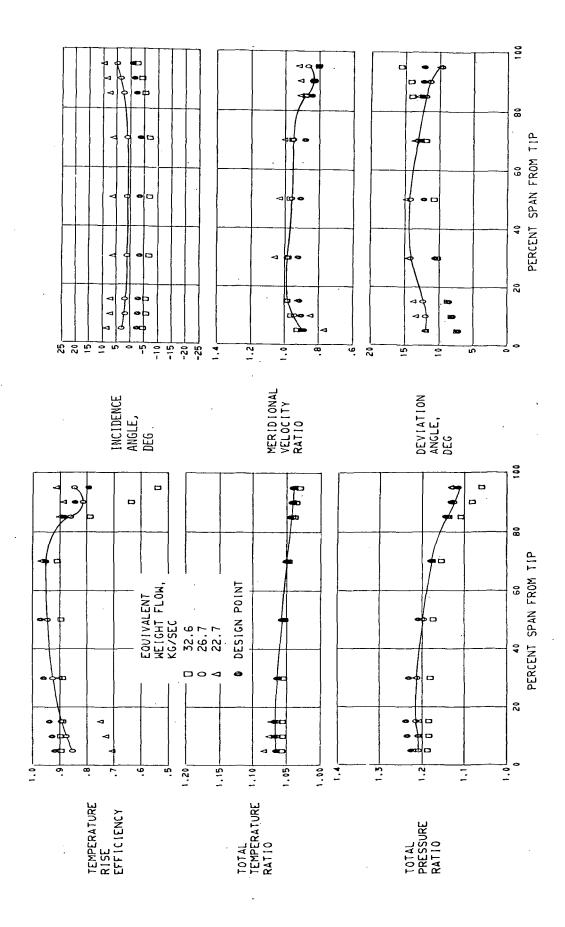


Figure 6. - Circumferential location of survey instrumentation at each station looking downstream.





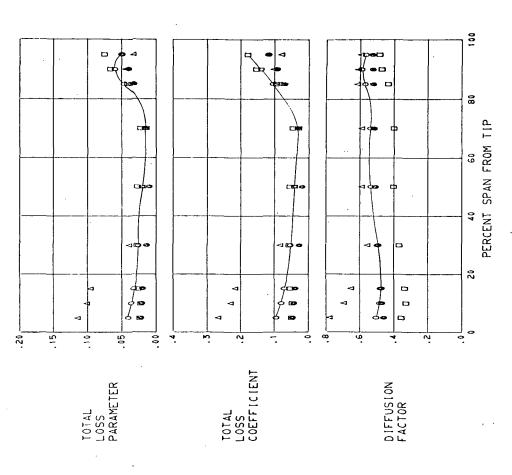


FIGURE 9. - RADIAL DISTRIBUTION OF PERFORMANCE FOR ROTOR 55. 100 PERCENT DESIGN SPEED.

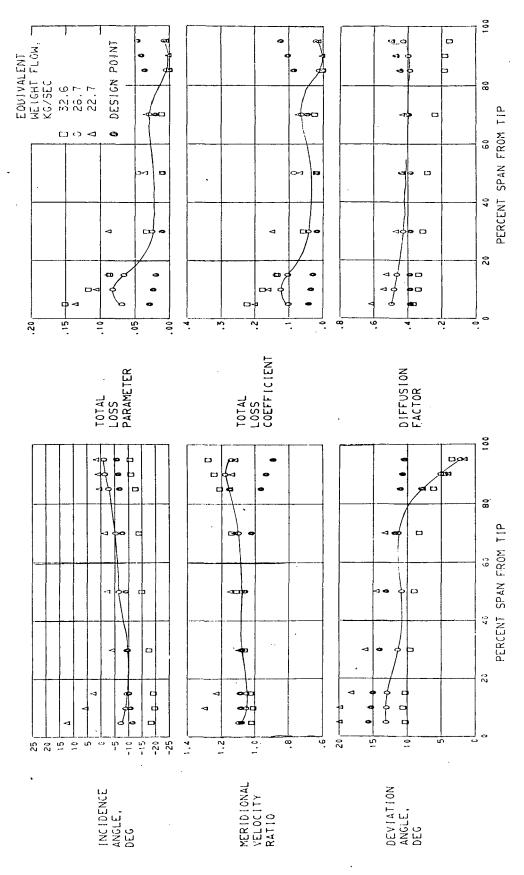
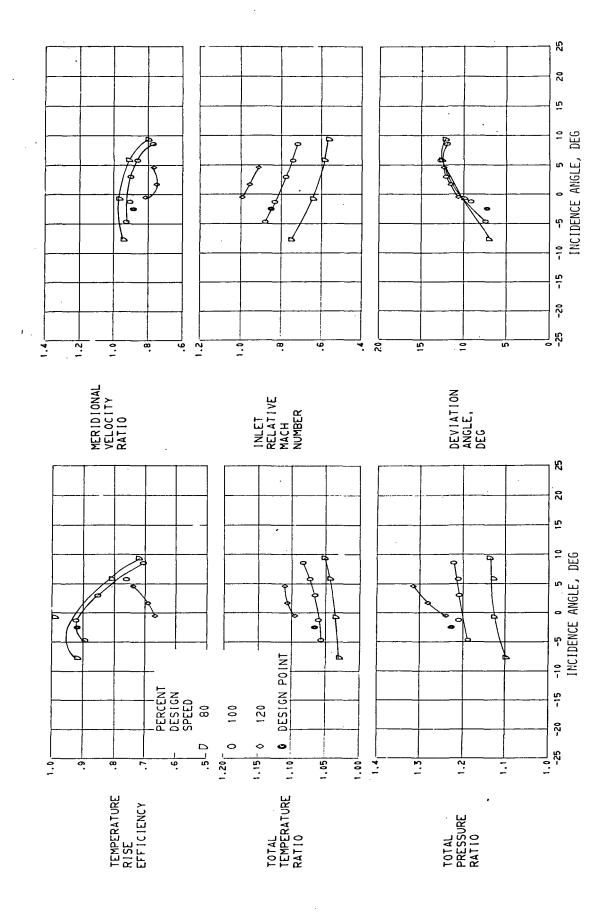


FIGURE 10, - RADIAL DISTRIBUTION OF PERFORMANCE FOR STATOR 55, 100 PERCENT DESIGN SPEED



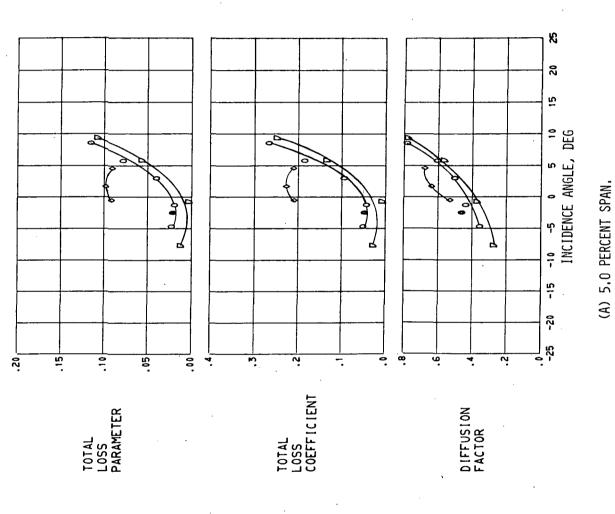
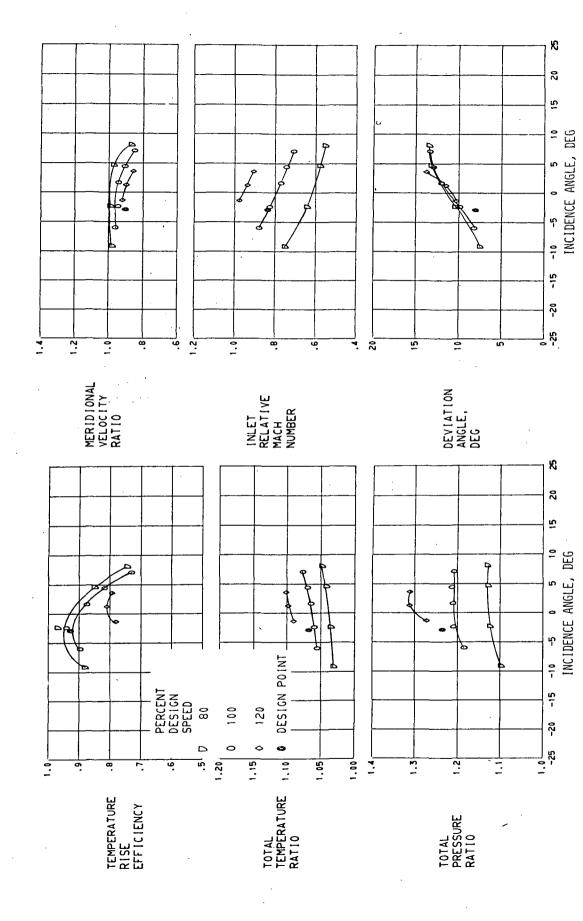


FIGURE 11, - BLADE-ELEMENT PERFORMANCE FOR ROTOR 55,



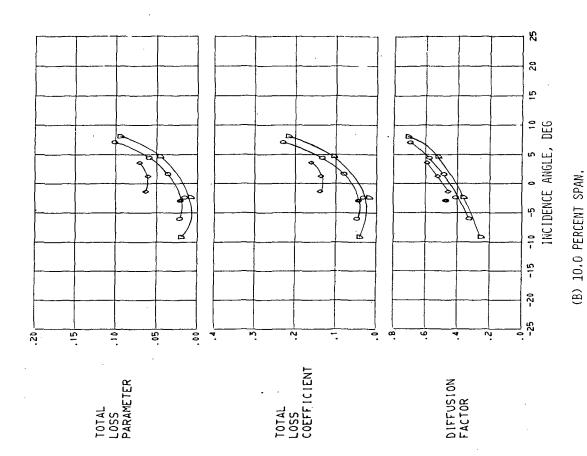
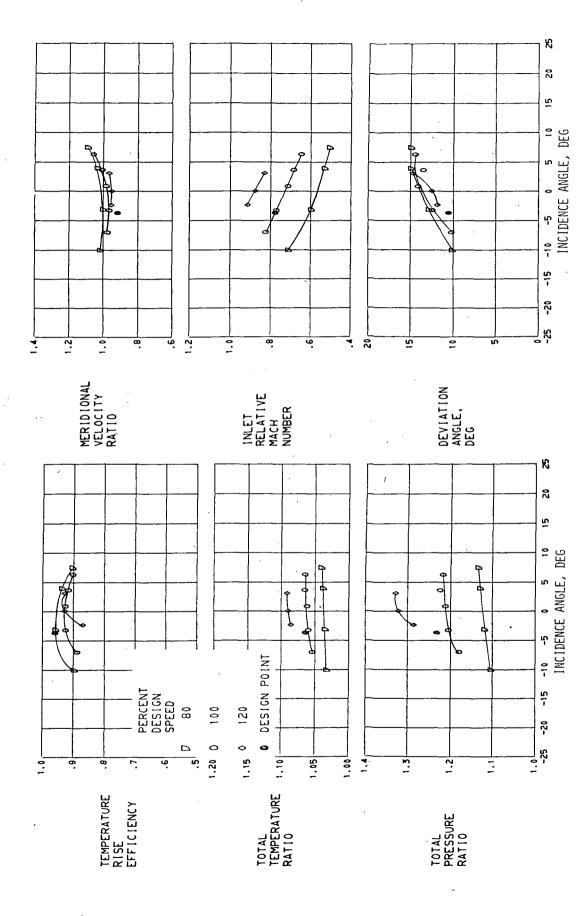


FIGURE 11. - CONTINUED. BLADE-ELEMENT PERFORMANCE FOR ROTOR 55.



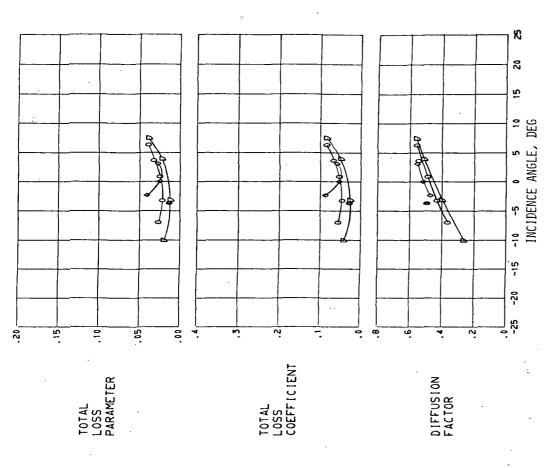
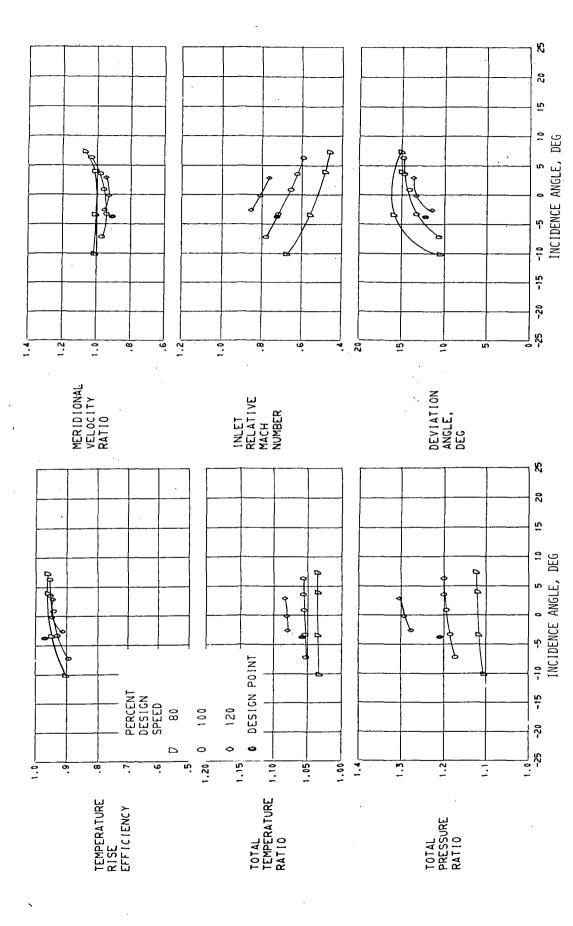


FIGURE 11. - CONTINUED. BLADE-ELEMENT PERFORMANCE FOR ROTOR 55,

(C) 30.0 PERCENT SPAN.



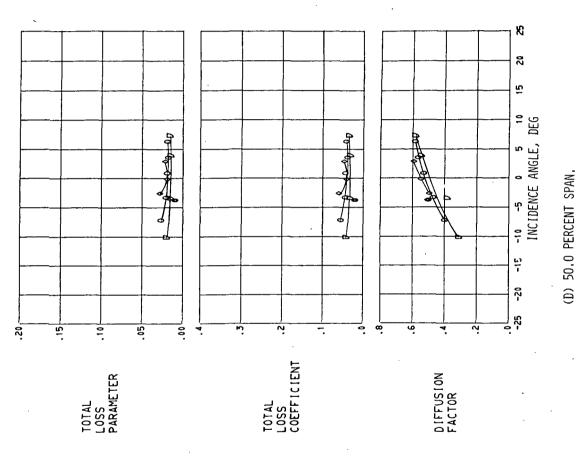
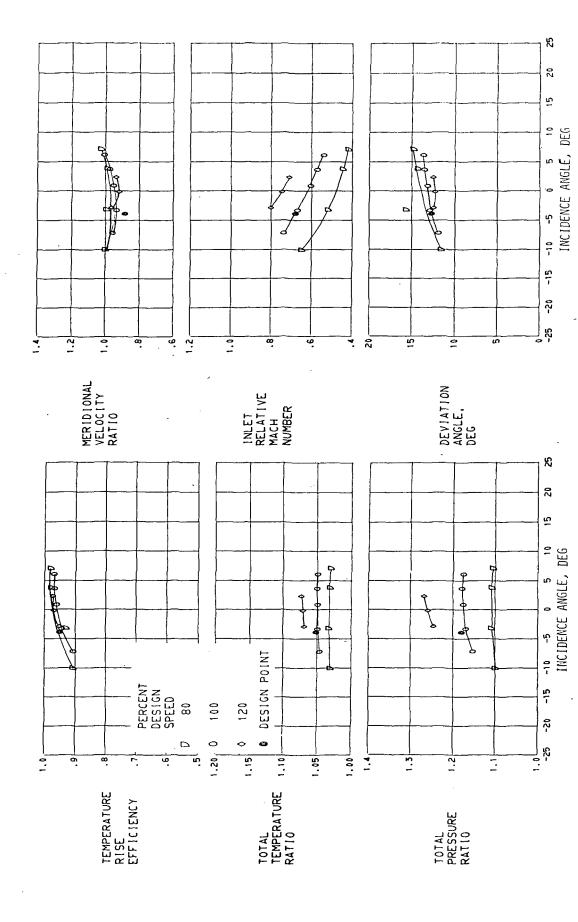


FIGURE 11. - CONTINUED. BLADE-ELEMENT PERFORMANCE FOR ROTOR 55.



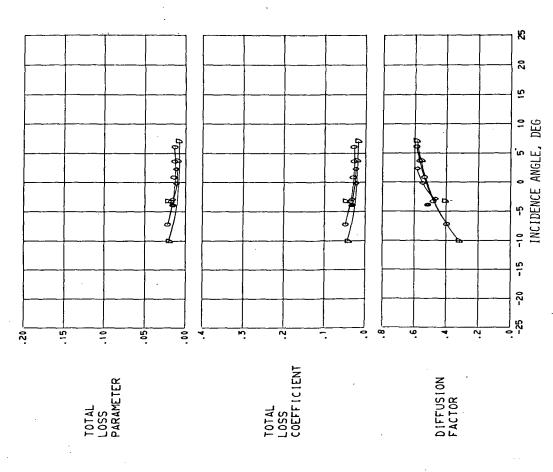
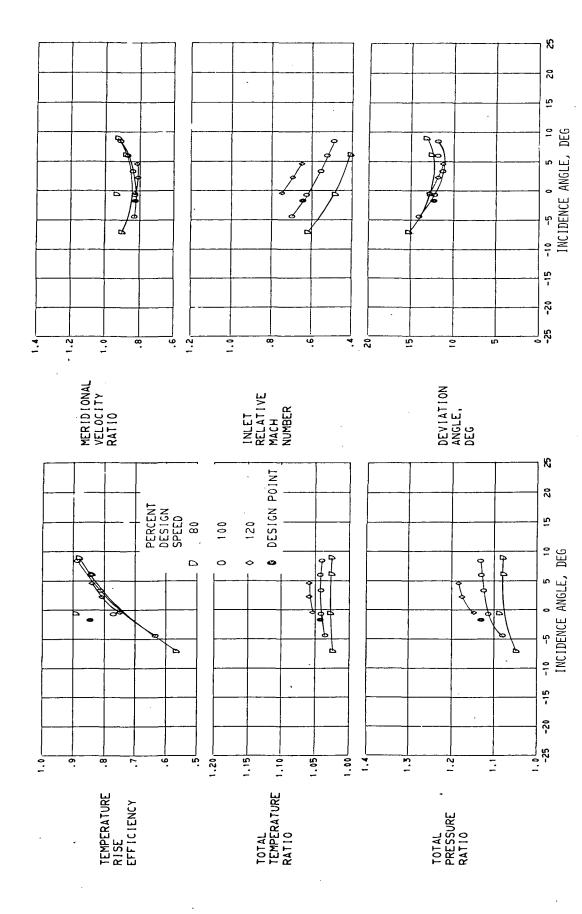


FIGURE 11. - CONTINUED, BLADE-ELEMENT PERFORMANCE FOR ROTOR 55.

(E) 70.0 PERCENT SPAN,



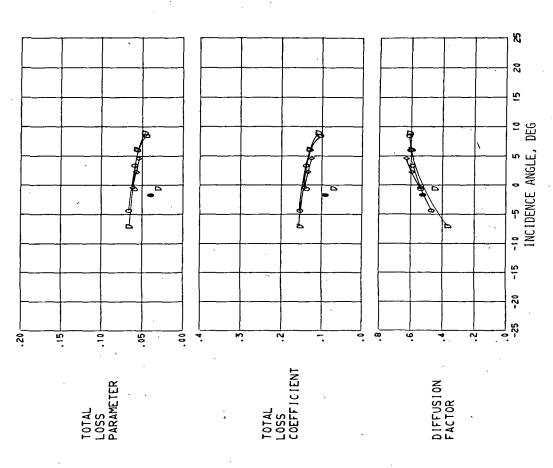
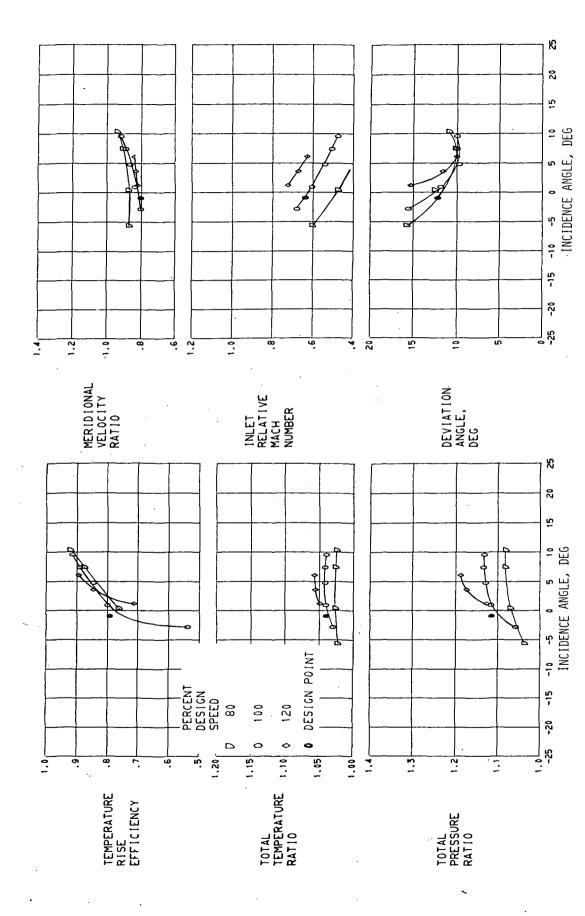


FIGURE 11. - CONTINUED. BLADE-ELEMENT PERFORMANCE FOR ROTOR 55.

(F) 90.0 PERCENT SPAN,



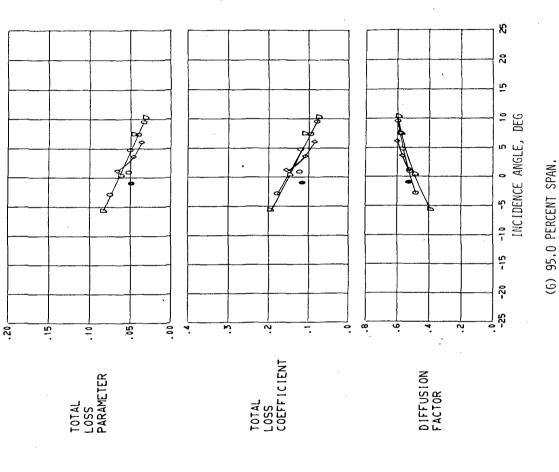


FIGURE 11. - CONCLUDED. BLADE-ELEMENT PERFORMANCE FOR ROTOR 55,

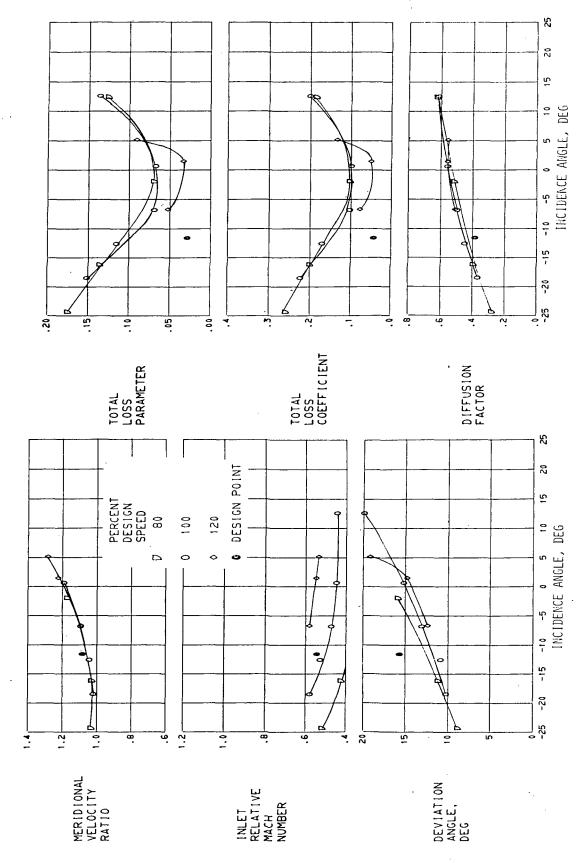


FIGURE 12. - BLADE-ELEMENT PERFORMANCE FOR STATOR 55.

(A) 5.C PERCENT SPAN.

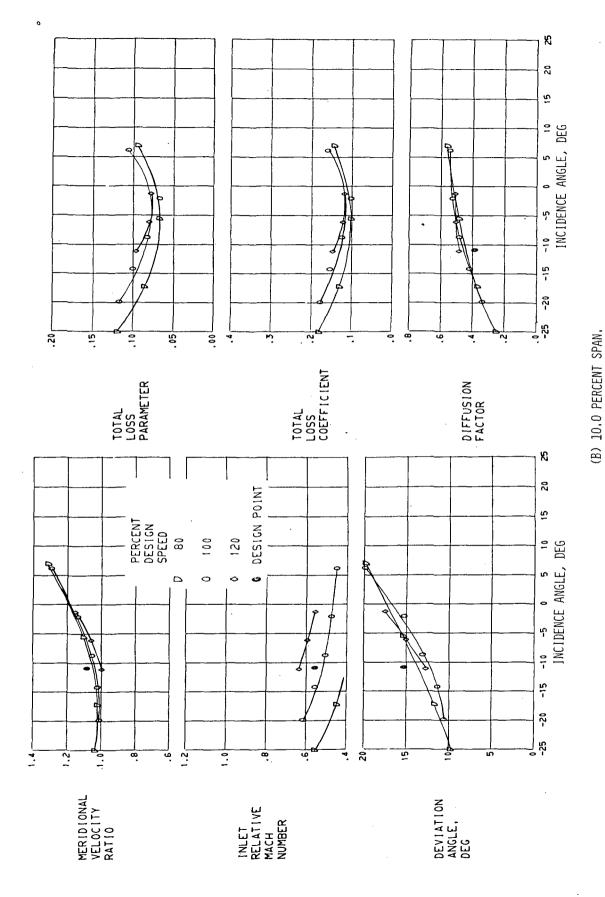


FIGURE 12. - CONTINUED. BLADE-ELEMENT PERFORMANCE FOR STATOR 55.

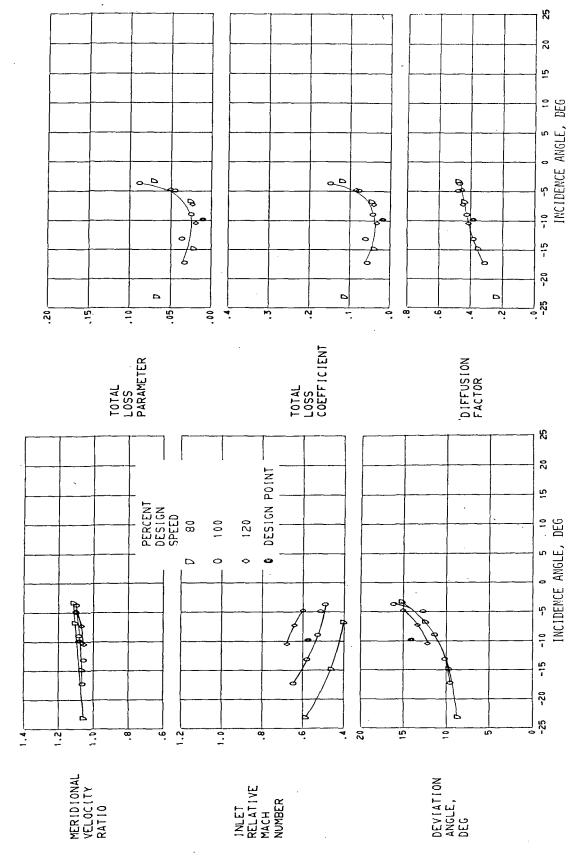


FIGURE 12. - CONTINUED. BLADE-ELEMENT PERFORMANCE FOR STATOR 55.

(C) 30,0 PERCENT SPAN,

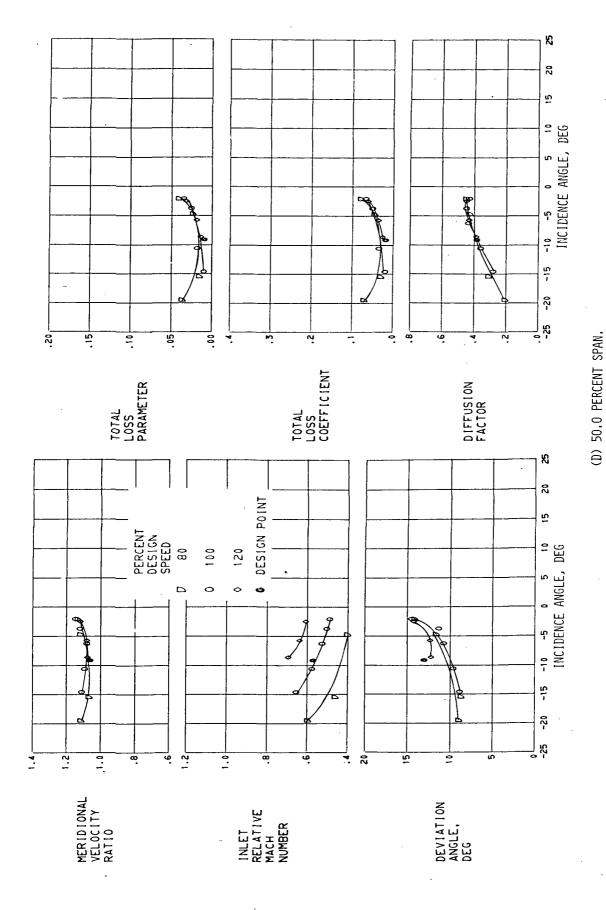


FIGURE 12. - CONTINUED. BLADE-ELEMENT PERFORMANCE FOR STATOR 55.

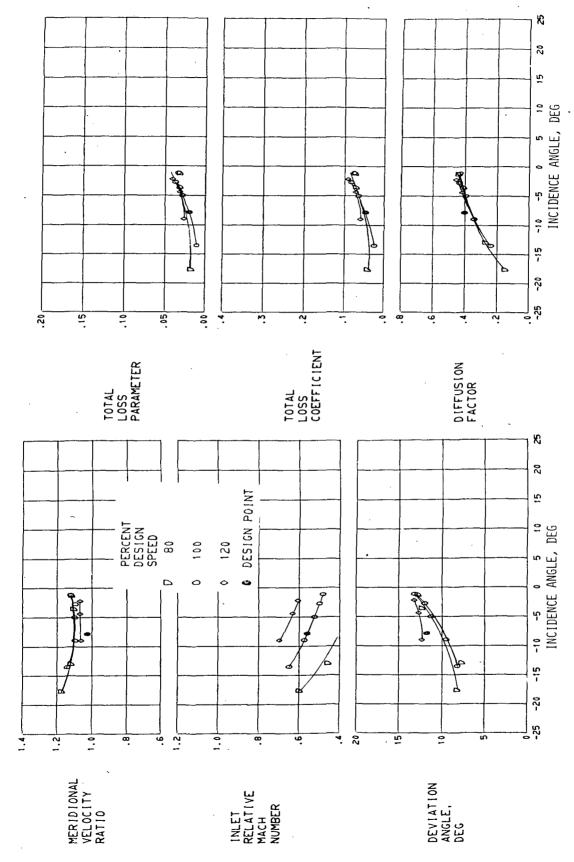


FIGURE 12. - CONTINUED. BLADE-ELEMENT PERFORMANCE FOR STATOR 55.

(E) 70,0 PERCENT SPAN.

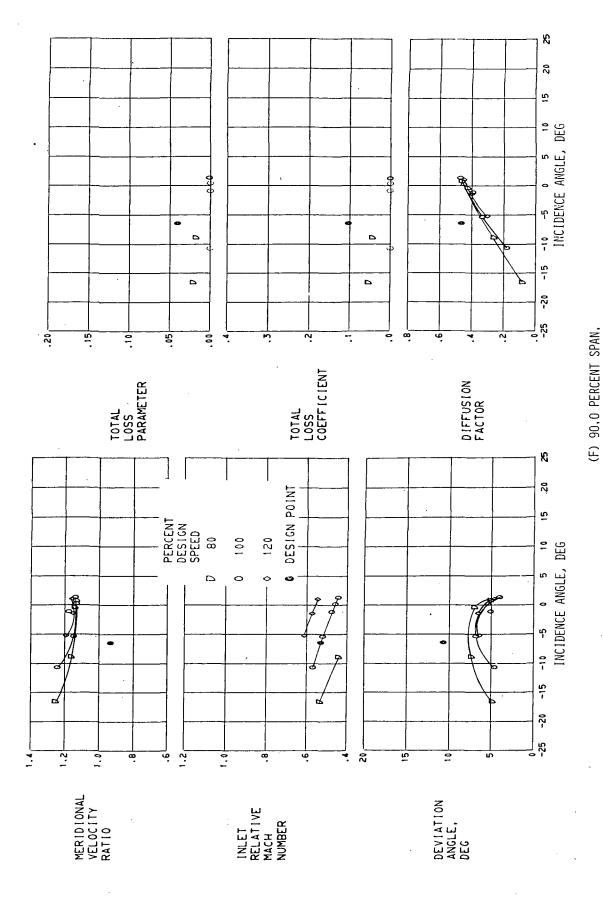


FIGURE 12, - CONTINUED. BLADE-ELEMENT PERFORMANCE FOR STATOR 55.

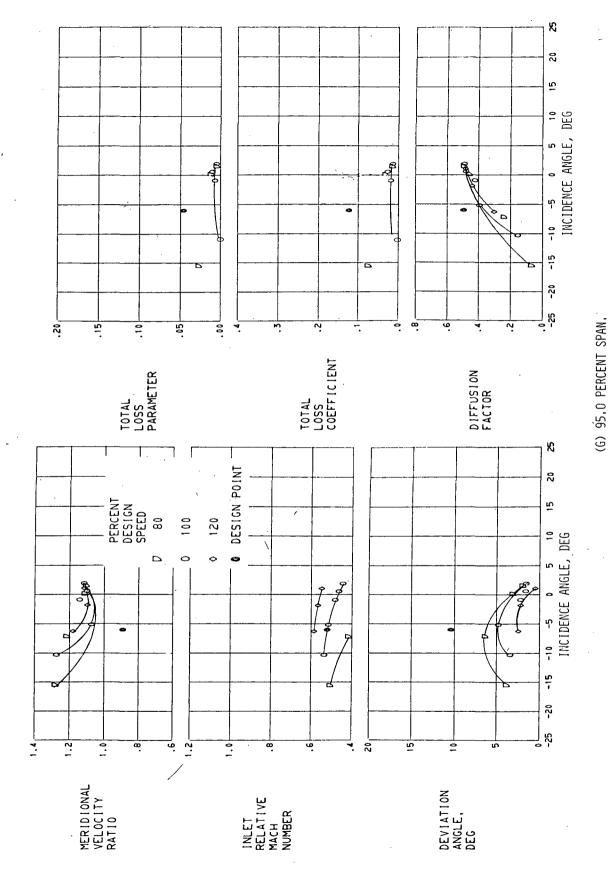


FIGURE 12. - CONCLUDED, BLADE-ELEMENT PERFORMANCE FOR STATOR 55.

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